## OPTIMUM DESIGN OF MULTISTEP SPUR GEARBOX

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### ABSTRACT

# OPTIMUM DESIGN OF MULTISTEP SPUR GEARBOX

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Optimum design of multistep gearbox, since many high-performance power transmission applications (e.g., automotive, space industry) require compact volume, has become an important interest area. This design application includes more complicated problems that are not taken into account while designing single stage gear drives. Design applications are generally made by trial and error methods depending on the experience and the intuition of the designer.

In this study, using Visual Basic 6.0, an interactive program is developed for designing multistep involute standard and nonstandard spur gearbox according to the American Gear Manufacturers Association (AGMA) Standards 218.01 and 2001-B88. All the equations for calculating the pitting resistance geometry factor I, and the bending strength geometry factor J, are valid for external spur gears that are generated by rack-type tools (rack cutters or hobs). The program is made for two-stage to six-stage gear drives, which are commonly used in the industry. Compactness of gear pairs and gearbox, and equality of factor of safety against bending failure is taken as the design objective. By considering the total required

gear ratio, the number of reduction stages is input by the user. Gear ratios of every stage is distributed to the stages according to the total gear ratio that satisfies the required precision (from  $\pm 0.1$  to  $\pm 0.00001$  on overall gear ratio) depending on the user selected constraints (unequal gear ratio for every stage, noninteger gear ratio e.g.). Dimensional design is determined by considering bending stress, pitting stress, and involute interference constraints. These steps are carried out iteratively until a desirable solution is acquired. The necessary parameters for configuration design such as number of teeth, module, addendum modification coefficient, are selected from previously determined gear pairs that satisfies the constraints by user interaction considering the performance criterion from the developed program. The positions of gears and shafts are determined automatically in order to keep the volume of gearbox as minimum while satisfying the nonlinear spatial constraints (center distance constraint for proper meshing of gear pairs, face distance constraint for proper assembly of pinion and gear having same shaft, gear interference constraint for preventing interferences between gears, shaft interference constraint for preventing interferences between gears and shafts) by using DLL (Dynamic Link Library) technology of Lingo 8.0 optimization software together with Visual Basic 6.0. If shaft interference constraint is removed then cantilevered mounting of gear pairs would also be possible, otherwise the gears should be mounted between bearings. Visual output of assembly is made by using Autodesk Inventor 7.0, automatically by the program.

Keywords: Gears, Spur Gears, Multistep Gearbox, Dimensional Design, Configuration Design

## ÖZ

# ÇOK KADEMELİ DÜZ DİŞLİ KUTUSUNUN OPTİMUM TASARIMI

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Çok kademeli dişli kutusunun optimum tasarımı, yüksek performanslı güç iletim uygulamalarının(otomotiv, uzay endüstrisi vb.) kompakt hacim gereksiniminden dolayı, önemli bir ilgi alanı haline gelmiştir. Tek kademeli dişli çarkların tasarımında göz önüne alınmayan, daha karmaşık problemleri içermektedir. Tasarım uygulamaları genellikle tasarımcının deneyimlerine ve kabullenmelerine bağlı olarak deneme ve yanılma yöntemiyle yapılmaktadır.

Bu çalışmada, Visual Basic 6.0 programı kullanılarak, çok kademeli evolvent profilli standart ve tashihli düz dişlilerin Amerikan Dişli Üreticileri Birliği 218.01 ve 2001-B88 Standartlarına göre tasarımını yapan interaktif bir program geliştirilmiştir. Yüzey basınç geometri faktörü I ve eğilme mukavemeti faktörü J'nin hesaplanması için gerekli olan bütün denklemler, kremayer tipindeki takımlar (kremayer takım veya freze) tarafından oluşturulan dış düz dişli çarklar için geçerlidir. Program sanayide çoğunlukla kullanılan iki kademeden, altı kademeye kadar olan çok kademeli dişliler için yapılmıştır. Dişli çiftlerinin ve dişli kutusunun kompakt olması ve eğilme dayanım güvenlik katsayılarının eşit olması, tasarım amacı olarak alınmıştır. İstenilen toplam cevrim oranına göre, kademe sayısı kullanıcı tarafından belirlenmektedir. Her kademenin çevrim oranı, istenilen toplam çevrim oranını istenilen hassasiyeti (±0.1'den ±0.00001'e kadar) sağlayacağı şekilde, kullanıcının belirlediği sınırlamalara (her kademe için eşit olmayan çevrim oranı, tamsayı olmayan cevrim oranı vb.) bağlı olarak dağıtılmaktadır. Dislilerin boyutları, eğilme mukavemeti, yüzey basıncı ve evolvent interferansı göz önüne alınarak belirlenmiştir. Bu basamaklar istenilen sonuç sağlanıncaya kadar iteratif bir şekilde yapılmıştır. Konfigürasyon tasarımı için gerekli olan diş sayıları, modül, modifikasyon katsayıları gibi parametreler, daha önce belirlenmiş gerekli şartları sağlayan dişli çiftlerinden, performans kriterlerinin değerlendirilmesiyle kullanıcı tarafından seçilmektedir. Dişlilerin ve şaftların konumları, lineer olmayan sınır şartlarını (dişli çiflerinin uygun eşleşmesi için merkezler arası uzaklık sınırlaması, aynı şafttaki pinyon ve dişlinin uygun montajı için yüzey uzaklık sınırlaması, dişliler arasındaki interferansı engellemek için dişli interferansı sınırlaması, dişlilerle şaftlar arasındaki interferansı engellemek için şaft interferansı sınırlaması) sağlayacak şekilde dişli kutusunun hacminin minimum olması için, Lingo 8.0 optimizasyon programının DLL (Dinamik Bağlantı Kütüphanesi) teknolojisinin Visual Basic 6.0 ile birlikte kullanılmasıyla, otomatik olarak yapılmaktadır. Eğer şaft interferansı sınırlaması kaldırılırsa, dişli çiftlerinin ankastre yataklanması da mümkün takdirde dislilerin rulmanlar olabilmektedir. aksi arasında vataklanması gerekmektedir. Montaj resminin görüntüsel cıktısı Autodesk Inventor 7.0 kullanılarak program tarafından otomatik olarak yapılmaktadır.

Anahtar Kelimeler: Dişliler, Düz Dişliler, Çok Kademeli Dişli Kutusu, Boyutsal Tasarım, Konfigürasyon Tasarımı

To My Parents

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## NOMENCLATURE

a	Addendum
a <sup>c</sup>	Rack cutter addendum
a <sub>c</sub>	Chordal addendum
b	Dedendum
b <sup>c</sup>	Rack cutter dedendum
BHN <sub>p</sub>	Brinell hardness number of pinion
BHNg	Brinell hardness number of gear
С	Standard Center distance
C <sub>m</sub>	Modified center distance
Carack	Rack cutter addendum factor
C <sub>b</sub>	Rack cutter dedendum factor
$C_{\mathrm{f}}$	Rack cutter tooth tip fillet radius factor
c	Bottom clearence
Ca	Application factor for pittting resistance
$C_{\mathrm{f}}$	Surface condition factor
C <sub>H</sub>	Hardness ratio factor
C <sub>L</sub>	Life factor for pitting resistance
C <sub>m</sub>	Load distribution factor for pitting resistance
C <sub>ma</sub>	Mesh alignment factor
C <sub>mc</sub>	Lead correction factor
$C_{mf}$	Face load distribution factor
C <sub>mt</sub>	Transverse load distribution factor
C <sub>e</sub>	Mesh alignment correction factor
$C_{pf}$	Pinion proportion factor
C <sub>pm</sub>	Pinion proportion modifier

C <sub>p</sub>	Elastic coefficient
C <sub>R</sub>	Reliability factor for pitting resistance
Cs	Size factor for pitting resistance
C <sub>T</sub>	Temperature factor for pitting resistance
$C_v$	Dynamic factor for pitting resistance
D	Gear pitch diameter
D <sub>HPSTC</sub>	Gear diameter at HPSTC
D <sub>LPSTC</sub>	Gear diameter at LPSTC
D <sub>b</sub>	Gear base diameter
Do	Gear outside diameter
D <sub>r</sub>	Gear root diameter
$D_x$	Diameter at which the load is acting to the gear tooth
d	Pinion pitch diameter
$d_{\text{HPSTC}}$	Pinion diameter at HPSTC
$d_{LPSTC}$	Pinion diameter at LPSTC
$d_b$	Pinion base diameter
$d_o$	Pinion outside diameter
$d_r$	Pinion root diameter
$d_x$	Diameter at which the load is acting to the pinion tooth
E <sub>G</sub>	Modulus of elasticity of gear
E <sub>P</sub>	Modulus of elasticity of pinion
et	Total lead mismatch between mating teeth
F	Face width
G	Tooth stiffness constant
H,L,M	Parameters used for calculation of $K_{\rm f}$
$h_{\mathrm{f}}$	Height of the Lewis parabola
HPSTC	Highest point of single-tooth contact
Ι	AGMA geometry factor for pitting resistance
J	AGMA geometry factor for bending strength
$\mathbf{K}_{\mathrm{f}}$	Stress correction factor
Ka	Application factor for bending strength
K <sub>L</sub>	Life factor for bending strength

K <sub>m</sub>	Load distribution factor for bending strength
K <sub>R</sub>	Reliability factor for bending strength
Ks	Size factor for bending strength
K <sub>T</sub>	Temperature factor for bending strength
$K_v$	Dynamic factor for bending strength
LPSTC	Lowest point of single-tooth contact
m	Module
$m_{\rm f}$	Slope of the tooth root fillet curve
m <sub>p</sub>	Contact ratio, Slope of the Lewis parabola
m <sub>g</sub>	Gear ratio
m <sub>a</sub>	Aspect ratio
Ν	Number of load cycles
n <sub>p</sub>	Pinion speed
n <sub>G</sub>	Gear speed
N <sub>p</sub>	Number of teeth for pinion
N <sub>G</sub>	Number of teeth for gear
n <sub>bp</sub>	Factor of safety for pinion against bending failure
n <sub>bG</sub>	Factor of safety for gear against bending failure
n <sub>cp</sub>	Factor of safety for pinion against surface failure
n <sub>cG</sub>	Factor of safety for gear against surface failure
p <sub>b</sub>	Base pitch
p <sup>c</sup>	Rack cutter pitch
Р	Transmitted power
$Q_v$	AGMA quality number
R <sub>i</sub>	Radius of a point on involute gear tooth profile
r	Radius
r <sub>L</sub>	Theoretical limit radius
r <sub>f</sub>	Rack cutter tip fillet radius
r <sub>i</sub>	Radius of a point on involute pinion tooth profile
r <sub>o</sub>	Outside radius
r <sub>ps</sub>	Standard pitch radius
r <sub>ti</sub>	Radius at which involute tooth profile starts

$\mathbf{s_{f}}$	Tooth thickness at critical section				
S <sub>c</sub>	Surface fatigue strength				
St	Bending strength				
S <sub>ac</sub>	Allowable contact stress				
<b>S</b> <sub>at</sub>	Allowable bending stress				
t	Circular tooth thickness at standard pitch radius				
Т	Transmitted torque				
t <sub>t</sub>	Circular tooth thickness at tooth tip				
t <sup>c</sup>	Rack cutter tooth thickness at pitch line				
X	Addendum modification coefficient (rack shift) coefficient				
X <sub>f</sub>	x-coordinate on the trochoid fillet profile				
Xi	x-coordinate on the involute fillet profile				
X <sub>t</sub>	x-coordinate on the point of tangency				
Vt	Pitch line velocity				
$V_{\text{tmax}}$	Pitch line velocity limit				
W <sub>t</sub>	Transmitted tangential load				
Y	Tooth form factor				
y <sub>L</sub>	y-coordinate of the vertex of the Lewis parabola				
y <sub>f</sub>	y-coordinate on the trochoid fillet profile				
yi	y-coordinate on the involute fillet profile				
y <sub>t</sub>	y-coordinate on the point of tangency				
Z	Length of line of action				
Za	Length of pinion addendum (gear dedendum) portion of the line of				
	action				
Z <sub>b</sub>	Length of pinion dedendum (gear addendum) portion of the line of				
	action				
Zc	Distance between the pitch point and HPSTC of gear (LPSTC of				
	pinion) measured along the line of action when the gear is not				
	undercut				
Ze	Distance between the pitch point and HPSTC of pinion (LPSTC of				
	gear) measured along the line of action when the pinion is not				
	undercut				

$\Delta$	Half width of the hob-tooth land				
8	Distance between the pitch point and the contact point measured				
	measured along the line of action				
ε <sub>x</sub>	x-component of $\varepsilon$				
$\epsilon_{y}$	x-component of $\varepsilon$				
ρ	Radius of curvature				
$ ho_{ m f}$	Minimum radius of curvature of the fillet curve				
φ	Pressure angle				
$\phi_{\rm L}$	Load angle				
$\boldsymbol{\varphi}_t$	Pressure angle at tooth tip				
$\phi^{c}$	Rack cutter pressure angle				
β,γ,θ	Construction angles				
$oldsymbol{S}_b$	Bending Stress				
$\boldsymbol{S}_{c}$	Contact Stress				
$\mu_p$	Poison's ratio for pinion				
$\mu_{G}$	Poison's ratio for gear				
Subscripts					
i	Number of stage				
G	Gear				
m	Modified				
Р	Pinion				
S	Shaft				

### **CHAPTER 1**

#### **INTRODUCTION**

Gears are used to transmit torque and angular velocity in a wide variety of applications. The spur gear is designed to operate on parallel shafts and having teeth parallel to the shaft axis.

Gears are standardized with respect to the tooth shape and size. The American Gear Manufacturers Association (AGMA) publishes standards for their design. Most of the gears manufactured today have involute profiles in order to provide constant angular velocity ratio during meshing. The design process has become a challenge because of the complicated shape and geometry of gears.

The first formula for computing bending stress in gear teeth has been explained by Wilfred Lewis in 1892 and it still remains the basis for most gear design today. He derived an equation for determining the stress in a gear tooth by treating the tooth as a cantilever beam. This equation considers only the bending of the tooth and neglects the compression due to radial component of the transmitted load. At that time, stress concentration factors are not also included in the equation. The tooth root stress is calculated by using the transmitted tangential load, face width, circular pitch and Lewis Form Factor (Y). This formula for bending stress assumed no load sharing between teeth. It was calculated by assuming that the greatest force is exerted at the tip of the tooth. If the contact ratio is 1, the load will be applied at the tip of the tooth, creating the largest possible bending moment. In this case, one tooth is leaving contact just as the next is beginning contact, this is undesirable because slight errors in the tooth spacing will cause oscillations in the velocity, vibration, and noise which are not desired to achieve a quality gear set. At larger contact ratios than 1, examination of run-in teeth show that the tip loading is not the worst because another pair of teeth will be in contact when this condition occurs. These will occur toward the center of the mesh region where the load is applied at a lower position on the tooth, rather than its tip. This point is called the highest point of single-tooth contact or HPSTC.

Then American Gear Manufacturers Association (AGMA) introduced new equations to calculate both the bending strength and pitting resistance of gears including the radial component of the transmitted load and load sharing in stress calculations based on the studies of Earl Buckingham, Dolan and Broghamer. Dolan and Broghamer conducted a photoelastic investigation of stress concentration which is used for the calculation of  $K_f$  (stress correction factor by AGMA). Buckingham introduced an equation to calculate the pitting resistance of gears. Besides, the geometry factor J was established in which the stress concentration is also taken into consideration.

Based on the work of Lewis [1], AGMA [2] has established an improved model for the elastic behaviour of gear teeth, which uses the Lewis parabola to determine the critical section by the tangency point of the parabola and the tooth profile.

Mitchiner and Mabbie [3] developed a method valid for rack type, standard spur gears, and Tables for the AGMA J factor for usual sets of tooth proportions were given but neither hob offsets nor backlash were considered.

Based on the work of Errichello [4], AGMA [2] developed an augmented method considering spur and helical gears, rack-type or pinion-type cutters, tool protuberance, tool offsets, and backlash. This method is general and provides extremely accurate results, but involves an iterative procedure very tedious to implement into a computer program.

Since the calculation of geometry factor J require iterative procedures, Pedrero, Fuentes and Estrems [5], developed an approximate method for the determination of the bending strength factor for external spur and helical gear teeth by analytical methods. This method is also valid for any set of tooth proportions, rack shift coefficients, backlash etc. but as the name implies the method is approximate.

The design of a gear pair usually requires some iteration. Not enough information typically exists in the problem statement to directly solve for the unknowns. The design is made by assuming some initial values and a trial solution is done.

Usually, the overall gear ratio and either the power and speed, or the torque and speed of one shaft are defined in the problem. The parameters to be defined are the number of teeth for the pinion and gear, the module, the face width, the safety factors. Some design decisions regarding the mesh-accuracy required, the number of cycles, the pressure angle, the tooth form (standard or non-standard), the gear manufacturing method, operating temperature range, and desired reliability must be made.

The studies on the design of gear drives have focused on the dimensional design of single-stage gear drives so far. In recent years, the need to achieve things like high gear ratio, high power capacity, and high efficiency in a compact unit increases the studies focused on applications of multi-stage gear drives.

Most of the studies have used various optimization techniques to overcome this complicated problem since there are a lot of conflicting parameters exist.

Optimum design of multi-stage gear drive, introduces a number of challenges, particularly if the design involves optimization of both its kinematic structure and gear geometries. The resulting optimization problem involves design variables which can be both integer-valued (for pinion and gear teeth), discrete-valued (for module), and real valued (for face width). The feasibility of a design solution depends on the satisfaction of a number of equality and inequality constraints involving strength and gear meshing restrictions.

Chong et al. proposed an automation algorithm for the design of two- and three-stage cylindrical gear drives. Two types of dimensional design processes are presented. The first design process uses the total volume of gears to determine gear ratio and uses K factor, unit load, and aspect ratio to determine gear ratios, the second one makes use of Niemann's formula and center distance to calculate gear ratio and gear dimensions. Then configuration design is done to determine the position of gears to minimize the geometrical volume of a gearbox by using simulated annealing algorithm [6].

Chong and Bae automated the design process by integrating the dimensional design and configuration design process. Number of reduction stages is assigned by the designer, overall gear ratio is split to stages using random search method, the dimensional design is made by generate and test method, and configuration design is done to minimize the geometrical volume of a gearbox [7].

Bae et al. presented a design algorithm to automate the preliminary design of multistage gear drives including both the dimensional design of gears and the configuration design of the gear train [8].

Hsu constructed an optimization model to select the reduction ratios that minimize the angular backlash of a gear train, under constraints on total gear ratio and available space [9].

Deb demonstrated the use of a multi-objective evolutionary algorithm, which is capable of solving the multi-speed gearbox design problem involving mixed discrete and real-valued parameters and more than one objectives, and is capable of finding multiple nondominated solutions in a single simulation run [10].

Rosic presented an analytical and computer aided procedure for the multicriteria design optimization of gear train transmission by applying the optimization methods in the field of gear train transmission including Monte Carlo Method [11].

Thompson et al. made a generalized optimal design formulation with multiple objectives applicable to a gear train of arbitrary complexity. Optimal value of diametral pitch is determined at which tooth bending fatigue failure and surface fatigue failure are equally likely [12].

Chong et al. demonstrated the analysis of the relation between the geometrical volume and the vibration of a gear pair satisfying strength and geometric constraints. The addendum modification coefficients are determined by equalizing the ratios of bending strengths and geometry factors between pinion and gear for optimal design case [13].

Dolen et al. investigates the optimal design of a four-stage gear train using genetic algorithms with five different encoding schemes [14].

All the studies are based on optimization techniques and use some heuristic search methods (simulated annealing methods, genetic algorithms) by simplifying the complicated formulation of the problem. Except Kubo, the addendum modification is not considered since they are mainly focused on preliminary design.

In this study, an interactive program is developed for designing multistep involute standard and non-standard spur gearbox according to the AGMA Standards. The program is made for two-stage to six stage gear drives. Compactness of gear pairs and gearbox, and equality of factor of safety against bending failure is taken as the design objective. By considering the total required gear ratio, the number of reduction stages is determined by the user. Gear ratios of every stage is distributed to the stages according to the total gear ratio that satisfies the required precision depending on the user selected constraints. Depending on the designer all alternative gear pairs or the selected ones are used for dimensional design. Dimensional design is determined by considering bending stress, pitting stress, and involute interference constraints. The necessary parameters for configuration design such as number of teeth, module, addendum modification coefficient, are selected from previously determined gear pairs that satisfies the constraints by user interaction considering the minimum volume criteria and the performance criterion from the developed program. The positions of gears and shafts are determined in order to keep the volume of gearbox as minimum while satisfying the spatial constraints by using Lingo 8.0 optimization software. Visual output of assembly is made automatically by using Autodesk Inventor 7.0.

In this thesis:

- In Chapter 2, spur gear geometry is explained.
- In Chapter 3, AGMA rating equations and factors are explained.
- In Chapter 4, addendum modification on gears is explained.
- In Chapter 5, multi-stage spur gear drive design is given.
- In Chapter 6, sample application problem is solved by the developed program.
- In Chapter 7, results of this study are explained and future works on this subject is given.
- In Appendix, user guide for the developed program and configuration design constraints for six stage spur gear drive are given.

## **CHAPTER 2**

### SPUR GEAR GEOMETRY

### 2.1 Spur Gears

Spur gears are used to transmit torque and angular velocity in a wide variety of applications. They are designed to operate on parallel shafts and having teeth parallel to the shaft axis.

An infinite number of curves can be used for gear-teeth profiles, which will produce conjugate action. The most used tooth profile form is the involute shape. Small changes in center distance do not affect tooth action.

The most common pressure angles are 14.5°, 20°, and 25°. The lower pressure angle has the advantage of smoother and quieter tooth action because of the larger profile contact ratio. Also lower loads are imposed on the support bearings because of a decreased radial component. The problem of undercutting associated with small numbers of pinion teeth is more severe with the lower pressure angle. The larger pressure angles have the advantages of better load carrying capacity.

#### 2.1.1 Spur Gear Nomenclature

Figure 2.1 shows a gear with the standard nomenclature and the terms explanations are as follows:



Figure 2.1 Gear Tooth Nomenclature [15]

Pinion and gear: The smaller of the two gears as the pinion and the other as the gear.

*Pitch circle:* It is an imaginary circle on a gear determined by dividing the number of teeth in the gear by the diametral pitch. Pitch circles of mating gears are tangent to each other.

Base circle: The circle from which the involute tooth profiles are generated.

Addendum circle: The circle passing from the tip of the tooth.

Dedendum circle: The imaginary circle tangent to the bottoms of the tooth spaces.

*Base pitch:* The distance along the line of action between successive involute tooth surfaces.

*Circular pitch:* The arc length along the pitch circle circumference measured from a point on one tooth to the same point on the next.

Face width: Length of the teeth in an axial plane.

*Module:* The ratio of the pitch diameter (mm) to the number of teeth.

Addendum: The radial distance between the outside diameter and the pitch circle.

Dedendum: The radial distance between the root diameter and the pitch circle.

*Clearance:* The amount between the gear dedendum and the addendum of the mating gear. The dedendum is slightly larger than the addendum to provide a small amount of clearance between the tip of one mating tooth and the bottom of the tooth space of the other.

*Backlash:* It is defined as the gap between mating teeth measured along the circumference of the pitch circle. Manufacturing tolerances precludes a zero backlash, as all teeth cannot be exactly the sane dimensions, and all must mesh without jamming.

*Pressure angle:* The angle between the tangent to the base circle and the line drawn normal to the line of centers at pitch point. It is shown in the Figure 2.3. The standard values are  $14.5^{\circ}$ ,  $20^{\circ}$ , and  $25^{\circ}$ . The most commonly used is  $20^{\circ}$  and the  $14.5^{\circ}$  now being obsolete.

*Involute:* The involute of a circle is a curve that can be generated by unwrapping a taut string from a cylinder, as shown in the Figure 2.2. The string is always tangent to the base circle. The center of curvature of the involute is always normal to the string with the base circle. A tangent to the involute is always normal to the string, which is the instantaneous radius of curvature of the involute curve.



Figure 2.2 Development of the involute of a circle [15]

## 2.1.2 Standard Modules

The module is a basic parameter, by which a gear tooth is specified and is equal to the ratio d/N, where d is the pitch diameter in millimetre and N is the number of teeth.

Modules are given with respect to the ISO 54 standard [16]. The modules are divided into two groups as the preferred series and the second choice series. In the Table 2.1 the values in the brackets are valid for the second choice series. Mostly the preferred series are used, only for some special applications the second choice series are used. In this study user can select the preferred series or both the preferred and the second choice modules in order to increase the alternatives.

0.12	0.6	2	(5.5)	(22)
(0.14)	(0.65)	(2.25)	(5.75)	25
0.16	0.7	2.5	6	(27)
(0.18)	(0.75)	(2.75)	(6.5)	(28)
0.2	0.8	3	(7)	(30)
(0.22)	(0.85)	(3.25)	8	32
0.25	0.9	(3.5)	(9)	(36)
(0.28)	(0.95)	(3.75)	10	(39)
0.3	1	4	(11)	40
(0.35)	(1.125)	(4.25)	12	(42)
0.4	1.25	(4.5)	(14)	(45)
(0.45)	(1.375)	(4.75)	16	50
0.5	1.5	5	(18)	(55)
(0.55)	(1.75)	(5.25)	20	60

Table 2.1 Standard Modules

#### 2.1.3 Mesh Geometry

Figure 2.3 shows a pair of involute pair of tooth forms in two positions, just beginning contact and about to leave contact.



Figure 2.3 Mesh Geometry [15]

The points of beginning and leaving contact define the mesh of the pinion and gear. The distance along the line of action between these points within the mesh is called the length of action Z, defined by the intersections of the respective addendum circles with the line of action. The distance along the pitch circle within the mesh is the arc of action, and the angles subtended by these points and the lines of centers are the angle of approach and angle of recess. The arc of action on both pinion and gear pitch circles must be the same length for zero slip between the theoretical rolling cylinders. The length of action Z can be calculated from the gear and pinion geometry.

#### 2.1.4 Undercutting

The involute tooth form is only defined outside of the base circle. In some cases, the dedendum will be large enough to extend below the base circle. If so, then the portion of tooth below the base circle will not be an involute and will interfere with the tip of the tooth on the mating gear, which is an involute. If the gear is cut with a standard gear shaper or a hob the cutting tool will also interfere with the portion of tooth below the base circle and will cut away the interfering material. This results in an undercut tooth.

Undercutting weakens the tooth by removing material at its root. The maximum moment and maximum shear from the tooth loaded as a cantilever beam both occur in this region. Severe undercutting will cause early tooth failure. Interference and its attendant undercutting can be prevented simply by avoiding gears with too few teeth.

### 2.1.5 Contact Ratio

It is the ratio of length of line of action divided by the base pitch and also contact ratio defines the average number of teeth in contact at any one time.

For contact ratios between 1 and 2, which are common for spur gears, there will still be times during the mesh when one pair of teeth will be taking the entire load.

However, these will occur toward the center of the mesh region where the load is applied at a lower position on the tooth, rather than its tip. This point is called the highest point of single tooth contact (HPSTC).

The minimum required contact ratio is 1.2 and the larger is better. Mostly the range is between 1.4 and 2.

### 2.1.6 Transmitted Tangential Load, W<sub>t</sub>

In most gear applications, the torque is not constant. Therefore, the transmitted tangential load will vary. To obtain values of the operating tangential load, the designer should use the values of power and speed at which the driven device will perform.  $W_t$  represents the tooth load due to the driven apparatus.

If the rating is calculated on the basis of uniform load, the transmitted tangential load is calculated by the following equations:

$$W_{t} = \frac{1000P}{v_{t}} = \frac{2000T}{d} = \frac{1.91 \times 10^{7} P}{n_{p} d}$$
(2.1)  
$$v_{t} = \frac{p.n_{p} d}{60000}$$
(2.2)

where,

T : transmitted pinion torque, (Nm)

P : transmitted power, (kW)

V<sub>t</sub> : pitch line velocity at operating pitch diameter, (m/s)

n<sub>p</sub> : pinion speed, (rpm)

### 2.1.7 Criteria for Tooth Capacity

#### 2.1.7.1 Relationship of Pitting Resistance and Bending Strength Ratings

There are two major differences between the pitting resistance and the bending strength ratings. Pitting is a function of the Hertzian contact (compressive) stresses between two cylinders and is proportional to the square root of the applied tooth load. Bending strength is measured in terms of the bending (tensile) stress in a cantilever plate and is directly proportional to this same load. The difference in nature of the stresses induced in the tooth surface areas and at the tooth root is reflected in a corresponding difference in allowable limits of contact and bending stress numbers for identical materials and load intensities.

The analysis of the load and stress modifying factors is similar in each case, so many of these factors have identical numerical values.

#### 2.1.7.2 Pitting Resistance

The pitting of gear teeth is considered to be a fatigue phenomenon. In most industrial practice, corrective and non-progressive initial pitting is not deemed serious. Initial pitting is characterized by small pits which do not extend over the entire face width or profile height of the affected teeth. The definition of acceptable initial pitting varies widely with gear application. Initial pitting occurs in localized, overstressed areas. It tends to redistribute the load by progressively removing high contact spots. Generally, when the load has been reduced or redistributed, the pitting stops.

The aim of the pitting resistance formula is to determine a load rating at which destructive pitting of the teeth does not occur during their design life. The ratings for pitting resistance are based on the formulas developed by Hertz for contact pressure between two curved surfaces, modified for the effect of load sharing between adjacent teeth.

## 2.1.7.3 Bending Strength

The bending strength of gear teeth is a fatigue phenomenon related to the resistance to cracking at the tooth root fillet in external gears and at the critical section in internal gears.

The basic theory employed in this analysis assumes the gear tooth to be rigidly fixed at its base, thus the critical stress occurs in the fillet. If the rim supporting the gear tooth is thin relative to the size of the tooth and the gear pitch diameter, another critical stress may occur not at the fillet but in the root area. The strength ratings are based on plate theory modified to consider:

- The compressive stress at tooth roots caused by the radial component of tooth loading.
- Non-uniform moment distribution resulting from the inclined angle of the load lines on the teeth.
- Stress concentrations at the tooth root fillets.
- The load sharing between adjacent teeth in contact.

The intent of the AGMA strength rating formula is to determine the load which can be transmitted for the design life of the gear drive without causing cracking or failure.

Occasionally, wear, surface fatigue, or plastic flow may limit bending strength due to stress concentrations around large, sharp cornered pits or wear steps on the tooth surface.

## 2.1.8 Fundamental Rating Formulas

Two fundamental equations are used according to the AGMA standards as defined in [17], one for pitting resistance and the other for bending strength.
### 2.1.8.1 Pitting Resistance

The fundamental pitting resistance formula for gear teeth is:

$$\boldsymbol{s}_{C} = \boldsymbol{C}_{P} \sqrt{\frac{W_{t} \boldsymbol{C}_{a}}{C_{v}} \frac{\boldsymbol{C}_{s}}{dF} \frac{\boldsymbol{C}_{m} \boldsymbol{C}_{f}}{I}}$$
(2.3)

where,

- $\boldsymbol{s}_{c}$ : Contact stress, MPa
- $C_p$  : Elastic coefficient, [MPa]<sup>0.5</sup>
- W<sub>t</sub> : Transmitted tangential load, N
- C<sub>a</sub> : Application factor for pitting resistance
- C<sub>s</sub> : Size factor for pitting resistance
- $C_m \;:\; Load \; distribution \; factor \; for \; pitting \; resistance$
- C<sub>f</sub> : Surface condition factor for pitting resistance
- C<sub>v</sub> : Dynamic factor for pitting resistance
- F : Net face width of narrowest member, mm
- I : Geometry factor for pitting resistance
- d : Operating pitch diameter of pinion, mm

## 2.1.8.2 Bending Stress

The fundamental bending stress formula for gear teeth is:

$$\boldsymbol{S}_{b} = \frac{W_{t}K_{a}}{K_{v}} \frac{1.0}{Fm} \frac{K_{s}K_{m}}{J}$$
(2.4)

where,

$$s_h$$
: Bending stress, MPa

- W<sub>t</sub> : Transmitted tangential load, N
- K<sub>a</sub> : Application factor for bending strength
- K<sub>s</sub> : Size factor for bending strength
- $K_m$ : Load distribution factor for bending strength
- $K_v$ : Dynamic factor for bending strength
- F : Net face width of narrowest member, mm
- m : Metric module in plane of rotation, mm
- J : Geometry factor for bending strength

## 2.1.8.3 Allowable Contact Stress Number

The relation of calculated contact stress number to allowable contact stress number is:

$$\boldsymbol{S}_{c} \leq \boldsymbol{S}_{c} \frac{\boldsymbol{C}_{L} \boldsymbol{C}_{H}}{\boldsymbol{C}_{T} \boldsymbol{C}_{R}}$$

$$(2.5)$$

where,

- $\boldsymbol{s}_{c}$ : Contact stress, MPa
- S<sub>c</sub> : AGMA surface fatigue strength, MPa
- $C_L$ : Life factor for pitting resistance
- $C_{H}\,$  : Hardness ratio factor for pitting resistance
- $C_T$ : Temperature factor for pitting resistance
- $C_R$ : Reliability factor for pitting resistance

## 2.1.8.4 Allowable Bending Stress Number

The relation of calculated contact stress number to allowable contact stress number is:

$$\boldsymbol{s}_{b} \leq \boldsymbol{S}_{t} \frac{\boldsymbol{K}_{L}}{\boldsymbol{K}_{T} \boldsymbol{K}_{R}} \tag{2.6}$$

where,

- $\boldsymbol{s}_b$ : Bending stress, MPa
- St : AGMA Bending Strength, MPa
- K<sub>L</sub> : Life factor for bending strength
- K<sub>T</sub> : Temperature factor for bending strength
- K<sub>R</sub> : Reliability factor for bending strength

The remaining equations and the factors will be explained in the following chapters including the developed program.

## **CHAPTER 3**

## AGMA RATING FACTORS

#### 3.1 Dynamic Factors, C<sub>v</sub> and K<sub>v</sub>

Dynamic Factors,  $C_v$  and  $K_v$ , account for internally generated gear tooth loads which are included by non-conjugate meshing action of the gear teeth. Even if the input torque and speed are constant, significant vibration of the gear masses, therefore dynamic tooth forces can exist. These forces result from the relative displacements between the gears as they vibrate in response to an excitation known as "transmission error".

When transmission error is unavailable, it is reasonable to use pitch (spacing) and profile accuracy.  $Q_v$  is the transmission accuracy level number.  $Q_v$  can be same as the quality number for the lowest quality member in the mesh from AGMA 2000-A88 when manufacturing techniques ensure equivalent transmission accuracy, or when the pitch and profile accuracy are the same as AGMA 2000-A88 tolerances.

Due to the approximate nature of the empirical curves and the lack of measured tolerance values at the design stage of the job, the dynamic factor curve should be selected based on experience with the manufacturing methods and operating considerations of the design.

Where gearing is manufactured using process controls which provide tooth accuracy's which correspond to  $Q_v > 12$  limits, or where the design and manufacturing techniques ensure a low transmission error which is equivalent to this accuracy, values of  $C_v$  and  $K_v$  between 0.90 and 0.98 may be used depending on the specifier's experience with similar applications and the degree of accuracy actually achieved. To use these values, the gearing

must be maintained in accurate alignment and adequately lubricated so that its accuracy is maintained under the operating conditions. Spur gears should have properly designed profile modification.

The empirical curves of Figure 3.1 are generated by the following equations for values of  $Q_v$ , such that  $6 < Q_v < 11$ . The curves may be extrapolated beyond the end points shown in Figure 3.1 based on experience and careful consideration of the factors influencing dynamic load. For the purposes of computer calculations, Equation 3.4 defines the end points of the curves on Figure 3.1.

$$C_{v} = K_{v} = \left(\frac{A}{A + \sqrt{200v_{t}}}\right)^{B}$$
(3.1)

where,

$$A = 50 + 56(1.0 - B) \tag{3.2}$$

$$B = \frac{(12 - Q_V)^{0.667}}{4} \tag{3.3}$$

Q<sub>v</sub>: transmission accuracy level number

$$v_{t \max} = \frac{[A + (Q_v - 3)]^2}{200}$$
(3.4)

where,

 $V_{tmax:}$  pitch line velocity maximum at operating pitch diameter (end point of  $C_v$  and  $K_v$  curves on Figure 3.1), m/s

For 
$$Qv = 5$$

$$C_{v} = K_{v} = \frac{50}{50 + \sqrt{200v_{t}}}$$
(3.5)

This equation should be used for gearing where the manufacturing process can be expected to give accuracies resulting in  $Q_v < 6$ . Equation 3.5 should not be used where V<sub>t</sub> exceeds 13 m/s.



Figure 3.1 Dynamic Factors, C<sub>v</sub> and K<sub>v</sub> [18]

#### 3.2 Application Factors, C<sub>a</sub> and K<sub>a</sub>

The loading model assumed that the transmitted load  $W_t$ , was uniform with time. The fluctuating moments on the teeth described in that section are due to the teeth coming into and out of mesh under a uniform or average load. If either the driving or driven machine has time-varying torques or forces, then these will increase the loading felt by the gear teeth above the average values.

In the absence of definitive information about the dynamic loads in the driving and driven machines, an application factor  $K_a$  can be applied to increase the tooth stress based on the shockiness of the machinery connected to the gear train. For example, if the gear train connects an electric motor to a centrifugal water pump (both of which are smooth-running devices) there is no need to increase the average loads and  $K_a = 1$ . But, if a single-cylinder, internal-combustion engine drives a rock crusher through a gear train, both the power source and the driven device deliver shock loads to the gear teeth and  $K_a > 1$ . Table 3.1 shows some AGMA-suggested values for  $K_a$  based on the assumed level of shock loading in driving and driven devices

**DRIVEN MACHINE** POWER SOURCE Uniform Moderate Shock Heavy Shock Uniform 1.00 1.25 1.75 or higher (Electric motor, turbine) Light shock 1.25 1.50 2 or higher (Multicylinder shock) Medium shock 1.50 1.75 2.25 or higher (Single-cylinder engine)

Table 3.1 AGMA application factors C<sub>a</sub> and K<sub>a</sub> [15]

## **3.3 Elastic Coefficient, Cp**

The elastic coefficient,  $C_p$ , is defined by the following equation:

$$C_{p} = \sqrt{\frac{10}{p \left[ \left( \frac{1.0 - m_{p}^{2}}{E_{p}} \right) + \left( \frac{1.0 - m_{G}^{2}}{E_{G}} \right) \right]}}$$
(3.6)

where,

$$\begin{array}{ll} C_P & : \mbox{Elastic coefficient, [Mpa]}^{0.5} \\ \mu_p \mbox{ and } \mu_G & : \mbox{Poisson's ratio for pinion and gear, respectively} \\ E_p \mbox{ and } E_G & : \mbox{Modulus of elasticity for pinion and gear, respectively, (Mpa)} \end{array}$$

The values of  $C_p$  for various combinations of gear and pinion materials are shown in Table 3.2.

Pinion Material	Modulus of Elasticity E <sub>p</sub> <sup>2</sup> lb/in <sup>2</sup> (Mpa)	Gear Material and Combined Elastic Coefficient, $C_p$ , [lb/in <sup>2</sup> ], ([Mpa] <sup>0.5</sup> )						
Wateria		Steel	Malleable Iron	Nodular Iron	Cast Iron	Aluminum Bronze	Tin Bronze	
Steel	$30 \ge 10^6$	2300	2180	2160	2100	1950	1900	
Steel	$(2 \times 10^5)$	(191)	(181)	(179)	(174)	(162)	(158)	
Malleable	25 x 10 <sup>6</sup>	2180	2090	2070	2020	1900	1850	
Iron	$(1.7 \times 10^5)$	(181)	(174)	(172)	(168)	(158)	(154)	
Nodular Iron	24 x 10 <sup>6</sup>	2160	2070	2050	2000	1880	1830	
	$(1.7 \times 10^5)$	(179)	(172)	(170)	(166)	(156)	(152)	
Cast Iron	22 x 10 <sup>6</sup>	2100	2020	2000	1960	1850	1800	
	$(1.5 \times 10^5)$	(174)	(168)	(166)	(163)	(154)	(149)	
Al. Bronze	17.5 x 10 <sup>6</sup>	1950	1900	1880	1850	1750	1700	
	$(1.2 \times 10^5)$	(162)	(158)	(156)	(154)	(145)	(141)	
Tin Bronze	16 x 10 <sup>6</sup>	1900	1850	1830	1800	1700	1650	
	$(1.1 \times 10^5)$	(158)	(154)	(152)	(149)	(141)	(137)	

Table 3.2 Elastic Coefficient, C<sub>p</sub> [18]

# 3.4 Surface Condition Factor, C<sub>f</sub>

The surface condition factor, C<sub>f</sub>, used only in the pitting resistance formula, depends on:

- Surface finish as affected by, but not limited to, cutting, shaving, lapping, grinding, shot peening
- Residual stress
- Plasticity effects (work hardening)

Standard surface condition factors for gear teeth have not yet been established for cases where there is a detrimental surface finish effect.

In such cases, some surface finish factor greater than unity should be used.

## 3.5 Size Factors, C<sub>s</sub> and K<sub>s</sub>

The size factor reflects non uniformity of material properties. It depends primarily on:

- Tooth size
- Diameter of parts
- Ratio of tooth size to diameter of part
- Face width
- Area of stress pattern
- Ratio of case depth to tooth size
- Hardenability and heat treatment of materials

Standard size factors for gear teeth have not yet been established for cases where there is a detrimental size effect. In such cases, some size factor greater than unity should be used.

The size factor may be taken as unity for most gears, provided a proper choice of steel is made for the size of the part and its heat treatment and hardening process.

### 3.6 Load Distribution Factors, C<sub>m</sub> and K<sub>m</sub>

Load distribution factor reflects the non-uniform distribution of the load along the lines of contact ( $C_m = K_m$ ). It depends on:

- Gear tooth manufacturing accuracy
- Alignment of the axes of rotation of the pitch cylinders of the mating gear elements.

- Elastic deflections of gear elements, shafts, bearings, housing and foundations which support the gear elements.
- Bearing clearances
- Hertzian contact and bending deformations at the tooth surface.
- Thermal expansion and distortion due to operating temperature
- Centrifugal deflections due to operating speed
- Tooth crowning and end brief

Values for C<sub>m</sub> and K<sub>m</sub> can be obtained from:

$$C_m = K_m = C_{mf} C_{mt} \tag{3.7}$$

where,

 $C_{mf}$ : face load distribution factor  $C_{mt}$ : transverse load distribution factor

Transverse load distribution factor accounts for the non-uniform distribution of load among the gear teeth, since standard procedures to evaluate the influence of  $C_{mt}$  have not been established, a value of unity may be used.

Face load distribution factor accounts for the non-uniform distribution of load across the gearing face width. There are two basic methods to evaluate it, empirical methods and analytical method.

Empirical method is used when:

- $F/d \leq 2$
- Gear elements are mounted between bearings (not overhung)
- Face width is up to 1016 millimetres
- Contact across full face width of narrowest member when loaded.

Following equation is used to calculate C<sub>mf</sub> by this method:

$$C_{mf} = 1 + C_{mc} \left( C_{pf} C_{pm} + C_{ma} C_{e} \right)$$
(3.8)

where,

- $C_{mc}$  : lead correction factor
- C<sub>pf</sub> : pinion proportion factor
- C<sub>ma</sub> : mesh alignment factor
- C<sub>e</sub> : mesh alignment correction factor

Above mentioned factors are found as follows:  $C_{mc}$  modifies peak load intensity when crowning or lead modification is applied.

 $C_{mc}$ : 1.0 for gear unmodified leads.  $C_{mc}$ : 0.8 for gear with leads properly modified crowning.  $C_{mc}$ : 0.8 for gear with lead correction.

 $C_{pf}$  accounts for deflections due to load (Figure 3.7)

when,

$$F \le 25.4 \text{ mm}$$

$$C_{pf} = \frac{F}{10.d} - 0.025 \tag{3.9}$$

 $25.4 < F \le 431.8 \text{ mm}$ 

$$C_{pf} = \frac{F}{10.d} - 0.00375 + 0.000492.F \tag{3.10}$$

 $431.8 < F \le 1016 \text{ mm}$ 

$$C_{pf} = \frac{F}{10.d} - 0.1109 + 0.000815F - 0.000000353F^2$$
(3.11)

For values  $\frac{F}{10d}$  less than 0.05, 0.05 is used for this value in the above equations.

The pinion proportion modifier,  $C_{pm}$ , alters  $C_{pf}$ , based on the location of the pinion relative to its bearing centerline.

$$C_{pm}$$
: 1.0 for straddle mounted pinions with  $(S_1/S) < 0.175$   
 $C_{pm}$ : 1.1 for straddle mounted pinions with  $(S_1/S) \ge 0.175$ 

where,

 $S_1$ : the offset of the pinion, i.e., the distance from the bearing span centerline to the pinion mid face, mm as shown in the Figure 3.2.

S : the bearing span, i.e., the distance between the bearing centerlines, mm



Figure 3.2 Evaluation of S and S<sub>1</sub>

The mesh alignment factor accounts for the misalignment of the axes of rotation of the pitch cylinders of the mating gear elements from all causes other then elastic deformations. The mesh alignment factor can be obtained from Figure 3.3.

The four curves of Figure 3.3 provide representative values for  $C_{ma}$  based on the accuracy of gearing and misalignment effects which can be expected for the four classes gearing shown.

The values for the four curves of Figure 3.3 are defined as follows:

$$C_{ma} = A + B(F) + C(F)^2$$
 (3.12)

where, empirical constants A, B, and C are given in Table 3.3

	А	В	С
Curve 1 Open Gearing	$2.47 \times 10^{-1}$	$0.657 \times 10^{-3}$	-1.186x10 <sup>-7</sup>
Curve 2 Commercial Enclosed Gear Units	$1.27 \times 10^{-1}$	$0.622 \times 10^{-3}$	-1.69x10 <sup>-7</sup>
Curve 3 Precision Enclosed Gear Units	$0.675 \times 10^{-1}$	$0.504 \times 10^{-3}$	-1.44x10 <sup>-7</sup>
Curve 4 Extra Precision Enclosed Gear Units	0.380x10 <sup>-1</sup>	$0.402 \times 10^{-3}$	$-1.27 \times 10^{-7}$

Table 3.3 Empirical Constants A, B, C



Figure 3.3 Mesh Alignment Factor, C<sub>ma</sub> [18]

During the program the gearing type is asked from the user that is to be selected from the list given below and then  $C_{ma}$  is calculated:

- Open gearing
- Commercial enclosed gear units
- Precision enclosed gear units
- Extra precision enclosed gear units

 $C_e$  is used to modify the mesh alignment when the manufacturing or assembly techniques improve the effective mesh alignment.

 $C_e = 0.8$  when gearing is adjusted at the assembly or when the compatibility of the gearing is improved by lapping.

 $C_e = 1.0$  for all other conditions.

Analytical method is used when:

- F/d > 2
- Applications with overhung gear element
- Application with long shafts subjects to large deflections
- Applications where contact does not extend across full face or narrowest member when loaded.

Two conditions are given to the user input:

• Contact across the entire face width under normal operating load (the most commonly encountered condition), the face load distribution factor can be evaluated with the following equation:

$$C_{mf} = 1 + \frac{G.e_t.F}{2W_t} \tag{3.13}$$

• Total tooth contact length under normal operating load is less than the face width, the face load distribution factor can be evaluated with the following equation:

$$C_{mf} = \sqrt{\frac{2.G.e_t.F}{W_t}} \tag{3.14}$$

where,

### G : tooth stiffness constant, MPa

The average mesh stiffness of a single pair of teeth in the normal direction. The usual range of this value that is compatible with this analysis, for steel gears is 1.0 to  $1.4 \times 10^4$  MPa. As advised by AGMA the most conservative value, the highest is taken for the developed program.

 $e_t$ : total lead mismatch between mating teeth, mm  $p_b$ : base pitch The amount of lead mismatch cannot be known at the design stage, then an average value is calculated as follows:

$$e_t = \frac{W_t}{1000.C_e} \tag{3.15}$$

where,

Ce : error coefficient

the usual range for  $C_e$  is between 0-15000, then 7500 is selected for evaluation of  $C_e$ 

## 3.7 Allowable Stress Numbers, Sac and Sat

The allowable stress numbers depend on:

- Material composition and cleanliness
- Mechanical properties
- Residual stress
- Hardness
- Type of heat treatment (surface or through hardened)

An allowable stress number for unity application factor, 10 million cycles of load application, 99 percent reliability and unidirectional loading, is determined or estimated from laboratory and field experience for each material and condition of that material. This stress number is designated  $S_{ac}$  or  $S_{at}$ . The allowable stress numbers are adjusted for design life load cycles by the use of life factors. The allowable stress numbers for gear materials vary with material composition, cleanliness, quality, heat treatment, and processing practices. For materials other than steel, a range is shown, and the lower values should be used for general design purposes. The allowable stress numbers are shown in Tables 3.4, 3.5 and Figures 3.4, 3.5.



Figure 3.4 Allowable Contact Stress Number for Steel Gears,  $S_{ac}$  [18]



Figure 3.5 Allowable Bending Stress Number for Steel Gears, Sat[18]

Material	AGMA	Commercial	Heat	Minimum Hardness at	S <sub>ac</sub>	
	Class	Designation	Treatment	Surface	Psi	MPa
				180 BHN	85-95000	(590-660)
	A-1		Through-	& less		
Steel	Through		hardened	240 BHN	105-115000	(720-790)
bieer	A-5	_	and	300 BHN	120-135000	(830-930)
	11.5		tempered	360 BHN	145-160000	(1000-1100)
				400 BHN	155-170000	(1100-1200)
			Flame- or	50 HRC	170-19000	(1200 - 1300)
			induction-	54 HRC	175-195000	(1200-1300)
			hardened			(
			Carburised	55 HRC	180-200000	(1250-1400)
			and case- hardened	60 HRC	200-225000	(1400-1550)
		A 161 4140	Nitrided	48 HRC	155-180000	(1100-1250)
		AISI 4140	Nitrided	46 HRC	150-175000	(1050-1200)
		AISI 4540 Nitrolloy 125 M	Nitrided	60 HRC	170-195000	(1170-1350)
		2 16 % abroma	Nitrided	54 HRC	155-172000	(1100-1200)
		2 72 % chiloine	Nitrided	60 HRC	192-216000	(1300-1500)
Cast	20		As cast	-	50-60000	(340-410)
Iron	30		As cast	175 BHN	65-75000	(450-520)
11011	40		As cast	200 BHN	75-85000	(520-590)
	A-7-a	60-14-18	Annealed	140 BHN	90-10	00% of
Nodular	A-7-c	80-55-06	Quenched	180 BHN	S <sub>c</sub> value	e of steel
(Ductile)	A-7-d	100-70-03	&	230 BHN	with	same
Iron	A-/-e	120-90-02	Tempered	270 BHN	hard	lness
	4.9					
Mallaabla	A-8-C	45007	-	165 BHN	72000	(500)
Iron	A-o-c	50005	-	180 BHN	78000	(540)
(Pearlitic)	A-0-1 A-8-i	53007	-	195 BHN	83000	(570)
(i caritic)	A-0-1	80002	-	240 BHN	94000	(650)
				Tensile		
	Decent		Sand and	strength		
Bronze	Bronze	AGMA	Sand-cast	minimum	30000	(205)
	2	20	Sand-cast	40000 lb/in <sup>2</sup>		
				275( MPa)		
				Tensile		
	Al/Br	ASTM	Heat-	strength		(150)
	3	B-148-52	Treated	minimum $00000 \text{ lb}/\text{km}^2$	65000	(450)
		Alloy 9C		50000 ID/III 620( MDa)		
				020( MPa)		

Table 3.4 Allowable Contact Stress Number,  $S_{ac}$  [17]

$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	Material	AGMA Class	Commercial Designation	Heat Treatment	Minimum Hardness at	CORE	S	at
Steel         A-1 Through A-5         A-1 Through A-5         Through A-5         A-1 Through A-5         Through A-5         A-1 A-1         Through A-5         A-1 A-5         Through A-5         A-1 A-5         Through A-5         A-1 A-5         Through A-5         A-1 A-5         Through A-5         A-1 A-5         Through A-4         A-1 A-5         Through A-5         A-5         Through A-5         A-5         Through A-5         A-5         Through A-5         A-5         Through A-5         A-5         Through A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a A-5         A-7-a B-0-14-18         Annealed Quenched & A-7-a A-5         A-7-a A-5         A-7-a B-0-12         A-7-a A-5         A-7-a B-0-12         A-7-a A-5         A-7-a B-0-12         A-7-a A-5		Chubb	Designation	Treutifient	Surface		Psi	MPa
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$					180 BHN	_	25-33000	(170-230)
Steel         Through A-5         hardened and ad         240 BHN 360 BHN 400 BHN         1         3141000 3647000         (210-280) (250-320)           A-5         -         -         and tempered         360 BHN 360 BHN         -         36-47000         (20-320)           A-5         -         Flame-or induction- hardened with type         Flame-or induction- hardened with type         -         45-55000         (310-380)           B         -         B pattern         -         22000         (150)           -         -         22000         (150)         (380-450)           -         -         22000         (380-450)           -         -         55-65000         (380-450)           -         -         22000         (250-321)           -         -         -         55-65000         (380-450)           -         -         -         55-65000         (380-450)           -         -         -         -         55-05000         (380-450)           -         -         -         -         55-05000         (380-450)           -         -         -         -         5000         (35)           -         - <td< td=""><td></td><td>Δ_1</td><td></td><td>Through-</td><td>&amp; less</td><td>_</td><td></td><td></td></td<>		Δ_1		Through-	& less	_		
Sile         Information of the second state of the strength o	Staal	Through		hardened	240 BHN	-	31-41000	(210-280)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	Sieer			and	300 BHN	-	36-47000	(250-320)
Image: constraint of the second sec		A-J	-	tempered	360 BHN	-	40-52000	(280-360)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$					400 BHN	-	42-56000	(290-390)
Image: second				Flame- or				
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				induction-				
Also         with type B pattern         spattern         -         22000         (150)           Image: Solution of the second state of the second stat				hardened	50-54 HRC	-	45-55000	(310-380)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				with type				· · · · ·
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				B pattern				
Induction-hardened with type         -         22000         (150)           2000         (150)         (150)         (150)           2000         (150)         (150)         (150)           2000         (150)         (150)         (150)           2010         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           2011         (150)         (150)         (150)           100         (160)         (10)         (10)         (10)           100<				Flame- or				
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				induction-				
With type B pattern         With type B pattern         Loso         (100)           Alsi 4140         Alsi 4140         Nitrided Alsi 4340         55 HRC 60 HRC         -         55-65000         (380-450)           Alsi 4340         Nitrided         48 HRC 0 HRC         300 BHN 30 BHN         34-45000         (250-32)           21/2         % chrome         Nitrided         46 HRC         300 BHN         34-45000         (250-32)           21/2         % chrome         Nitrided         54-60 HRC         300 BHN         38-48000         (260-330)           21/2         % chrome         As cast         -         -         5000         (380-450)           Cast         20         As cast         -         -         5000         (260-330)           Nodular         A-7-a         60-14-18         Annealed         140 BHN         -         8500         (69)           Nodular         A-7-a         60-14-18         Annealed         140 BHN         -         St value of steel           (Ductile)         A-7-a         80-55-06         Quenched         180 BHN         -         St value of steel           Iron         A-8-c         50005         -         165 BHN         -         10000 </td <td></td> <td></td> <td></td> <td>hardened</td> <td></td> <td>-</td> <td>22000</td> <td>(150)</td>				hardened		-	22000	(150)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				with type			22000	(150)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				B pattern				
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				Carburised				
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$				and case-	55 HRC	_	55-65000	(380-450)
$ \begin{array}{c c c c c c c c c c c c c c c c c c c $				hardened	60 HRC		55-70000	(380-480)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	-		AISI 4140	Nitrided	48 HRC	300 BHN	34-45000	(230-310)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$			AISI 4340	Nitrided	46 HRC	300 BHN	36-47000	(250-325)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$			Nitrallov 135 M	Nitrided	60 HRC	300 BHN	38-48000	(260-320)
$ \begin{array}{c c c c c c c c c c c c c c c c c c c $			$2\frac{1}{2}$ % chrome	Nitrided	54-60 HRC	300 BHN	55-65000	(200, 350) (380-450)
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$	-	20	272 % emonie	As cast	51 00 1110	JOO DIIIV	5000	(35)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	Cast	30		As cast	175 BHN	_	8500	(53)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	Iron	40		As cast	200 BHN	_	13000	(0)
$ \begin{array}{c ccccccccccccccccccccccccccccccccccc$		40		As cast	200 DIII		13000	()0)
$ \begin{array}{ c c c c c c c c c c c c c c c c c c c$	Nodular	A-7-c	60-14-18	Annealed	140 BHN	-	90-10	0% of
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	(Ductile)	A-7-C	80-55-06	Quenched	180 BHN	-	S <sub>t</sub> value	of steel
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	(Ductric)	A-7-0	100-70-03	&	230 BHN	-	with	same
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	non	A-7-C	120-90-02	Tempered	270 BHN	-	hard	ness
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$		A-8-c						
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	Malleable	A-8-e	45007	-	165 BHN	-	10000	(70)
$\begin{array}{c ccccccccccccccccccccccccccccccccccc$	Iron	Δ_8_f	50005	-	180 BHN	-	13000	(90)
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	(Pearlitic)	A-0-1 A 8 i	53007	-	195 BHN	-	16000	(110)
BronzeBronze 2AGMA 2CSand-cast Sand-castTensile strength minimum $40000 \text{ lb/in}^2$ $275(\text{ MPa})$ 5700(40)Al/Br 3ASTM B-148-52 Alloy 9CHeat- TreatedTensile strength minimum 90000 lb/in^2 $(200 \text{ MPa})$ 23600(160)	(i carnic)	A-0-1	80002	-	240 BHN	-	21000	(145)
BronzeBronze 2AGMA 2CSand-cast Sand-caststrength minimum $40000 \text{ lb/in}^2$ $275(\text{ MPa})$ 5700(40)Al/Br 3ASTM B-148-52 Alloy 9CHeat- TreatedTensile strength minimum $90000 \text{ lb/in}^2$ 23600(160)					Tensile			
BronzeBronzeAGMA 2CSand-cast Sand-castSubscript minimum 40000 lb/in² 275( MPa)5700(40)Al/Br 3ASTM B-148-52 Alloy 9CHeat- TreatedTensile strength minimum 90000 lb/in² (20( MPa))23600(160)					strength			
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$	Bronze	Bronze	AGMA	Sand-cast	minimum		5700	(40)
$\begin{array}{c c c c c c c c c c c c c c c c c c c $	DIONZC	2	2C	Sand-cast	$40000 \text{ lb/in}^2$		5700	(40)
$\begin{array}{c c c c c c c c c c c c c c c c c c c $					$275(MP_0)$			
Al/Br 3ASTM B-148-52 Alloy 9CHeat- Treatedstrength minimum 90000 lb/in223600(160)					ZIJ(MFa)			
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$			A STM		strongth			
$\begin{array}{ c c c c c c c c c c c c c c c c c c c$		Al/Br	ASIM D 149 52	Heat-	minimum		22600	(160)
Alloy 9C 90000 10/11		3	D-140-32	Treated	$00000 \text{ lb}/\text{im}^2$		23000	(100)
670 870 87			Alloy 9C		620( MDa)			

Table 3.5 Allowable Bending Stress Number,  $S_{at}$  [17]

### 3.8 Hardness Ratio Factor, C<sub>H</sub>

The pinion generally has a smaller number of teeth than the gear and consequently is subjected, to more cycles of contact stress. If both the pinion and the gear are through-hardened, then a uniform surface strength can be obtained by making the pinion harder than the gear. A similar effect can be obtained when a surface-hardened pinion is mated with a through-hardened gear. The hardness-ratio factor  $C_H$  is used only for the gear. Its purpose is to adjust the surface strengths for this effect. The values of  $C_H$  are obtained from the equation give below:

$$C_H = 1.0 + A(m_G - 1.0) \tag{3.16}$$

where,

$$A = 8.98.10^{-3} \left(\frac{H_{BP}}{H_{BG}}\right) - 8.29.10^{-3}$$
(3.17)

$$m_G = \frac{N_G}{N_P} \tag{3.18}$$

The terms  $H_{BP}$  and  $H_{BG}$  are the Brinell hardnesses (10-mm ball at 3000-kg load) of the pinion and gear, respectively. The term  $m_G$  is the speed ratio and is valid only when  $(H_{BP}/H_{BG}) \le 1.70$ .

#### 3.9 Life Factors, C<sub>L</sub> and K<sub>L</sub>

The life factors,  $C_L$  and  $K_L$ , adjust the allowable stress numbers for the required number of cycles of operation. For the purpose of this Standard, N, the number of load cycles is defined as the number of mesh contacts, under load, of the gear tooth being analyzed. AGMA allowable stress numbers are established for  $10^7$  tooth load cycles at 99 percent reliability. The life factor adjusts the allowable stress numbers for design lives other than  $10^7$  cycles.

The life factor accounts for the S/N characteristics of the gear material as well as for the gradual increased tooth stress which may occur from tooth wear, resulting in increased dynamic effects and from shifting load distribution which may occur during the design life of the gearing.

A  $C_L$  or  $K_L$  value of unity (1) may be used, where justified by experience, beyond  $10^7$  cycles.

### 3.9.1 Life Factors for Steel Gears

At the present time there is insufficient data to provide accurate life curves for all types of gears and gear applications. Experience, however, suggests life curves for pitting and strength of steel gears as shown in Figure 3.6 and Figure 3.7. These figures do not include data for nitrided gears. The upper portion of the shaded zones on the figures may be used for general commercial applications. The lower portions of the shaded zones are typically used for critical service applications where little pitting and tooth wear is permissible and where smoothness of operation and low vibration levels are required.



Figure 3.6 Bending Strength Life Factor, K<sub>L</sub> [18]



Figure 3.7 Pitting Resistance Life Factor, C<sub>L</sub>[18]

## 3.10 Reliability Factors, $C_R$ and $K_R$

The AGMA strengths presented are all based on a reliability R=0.99 corresponding to  $10^7$  cycles of life. Table 3.6 contains reliability factors which may be used to modify this allowable stresses to change that probability. These numbers are based on data developed for bending and pitting, failure by the U.S Navy. Other values may be used if specific data is available.

Table 3.6 Reliability Factors,  $C_R$  and  $K_R$  [15]

Requirements of application	C <sub>R</sub> ,K <sub>R</sub>
Fewer than one failure in 10000	1.5
Fewer than one failure in 1000	1.25
Fewer than one failure in 100	1
Fewer than one failure in 10	0.85

And also with Equation 3.19 C<sub>R</sub> and K<sub>R</sub> is calculated as:

$$C_R = \begin{cases} 0.7 - 0.15 \log(1 - R) & 0.9 \le R < 0.99 \\ 0.5 - 0.25 \log(1 - R) & 0.99 \le R < 0.9999 \end{cases}$$
(3.19)

## 3.11 Temperature Factors, C<sub>T</sub> and K<sub>T</sub>

The lubricant temperature is a reasonable measure of gear temperature. For steel materials in oil temperatures up to  $120^{\circ}$ C (250°F), K<sub>T</sub> can be set to 1. For higher temperatures, K<sub>T</sub> can be estimated from:

$$K_T = \frac{460 + T_F}{620} \tag{3.20}$$

where  $T_F$  is the oil temperature in °F and is valid only for steel gears.

### 3.12 AGMA Geometry Factor for Pitting Resistance I

This factor takes into account the radii of curvature of the gear teeth and the pressure angle shown in the Figure 3.8. AGMA defines Equation 3.21 for I:

$$I = \frac{\cos(f^c)}{d\left(\frac{1}{r_p} + \frac{1}{r_g}\right)}$$
(3.21)

where,

 $\boldsymbol{\varphi}^{c}$  : rack cutter pressure angle

 $\rho_p$ : radii of curvature of the pinion teeth

 $\rho_g$ : radii of curvature of the gear teeth

d : pitch diameter of pinion



Figure 3.8 Contact Point and Radii of Curvature [19]

 $\rho_p$  and  $\rho_g$  for any contact position can be found from the Figure 3.8 by the following equations.

$$\mathbf{x} = \frac{d}{2}\sin(\mathbf{f}^{c}) - \left[ (r_{i})^{2} - \frac{(d_{b})^{2}}{4} \right]^{0.5} = \left[ (R_{i})^{2} - \frac{(D_{b})^{2}}{4} \right]^{0.5} - \frac{d}{2}\sin(\mathbf{f}^{c}) \quad (3.22)$$

$$\boldsymbol{x}_{x} = \boldsymbol{x}\cos(\boldsymbol{f}^{c}) \tag{3.23}$$

$$\boldsymbol{x}_{y} = \boldsymbol{x}\cos(\boldsymbol{f}^{c}) \tag{3.24}$$

$$r_{i} = \left[ \left( \mathbf{x}_{x} \right)^{2} + \left( \frac{d}{2} - \mathbf{x}_{y} \right)^{2} \right]^{0.5}$$
(3.25)

$$R_{i} = \left[ \left( \mathbf{x}_{x} \right)^{2} + \left( \frac{D}{2} - \mathbf{x}_{y} \right)^{2} \right]^{0.5}$$
(3.26)

$$\boldsymbol{r}_{p} = \left[ \left( r_{i} \right)^{2} + \frac{\left( d_{b} \right)^{2}}{4} \right]^{0.5}$$
(3.27)

$$\boldsymbol{r}_{g} = \left[ \left( R_{i} \right)^{2} + \frac{\left( D_{b} \right)^{2}}{4} \right]^{0.5}$$
(3.28)

With respect to AGMA standard 218.01, LPSTC for the pinion is taken as the most critical point for calculation of AGMA geometry factor I, and all of the tables are prepared this rule.

"Although the rule is correct for most of the cases, there are cases for which it is not applicable. In x-zero gear pairs, a large positive pinion addendum modification coefficient results in a pinion having a very long addendum, and a very short dedendum; and a gear with a very long dedendum and a very short addendum. These tooth geometries shift the single tooth pair contact region of the pinion (region between HPSTC and LPSTC) above its pitch circle. A similar, but reverse shift is observed on the gear. Standard gears, or gears with small addendum modification coefficients have their pitch circles within this region; thus LPSTC is below, HPSTC is above the pitch circle. When the single tooth pair contact regions are shifted in this way, not LPSTC, but HPSTC becomes the most critical point" [20].

When running the developed program, the user is asked for radii of curvature calculation options, as defined by AGMA the critical, at LPSTC or both at LPSTC and at HPSTC.

## 3.13 Contact Ratio m<sub>p</sub>

Contact ratio  $m_p$  defines the average number of teeth in contact at any one time. It is calculated from Equation 3.29.

$$m_{p} = \frac{Z_{a} + Z_{b}}{p_{b}} = \frac{Z}{p_{b}}$$
(3.29)

where,

- Z<sub>a</sub>: length of the pinion addendum (gear dedendum) portion of the line of action
- Z<sub>b</sub>: length of the pinion dedendum (gear addendum) portion of the line of action
- $Z_c$ : distance between the pitch point and HPSTC of gear (LPSTC of pinion) measured along the line of action when the gear is not undercut
- $Z_e$ : distance between the pitch point and HPSTC of pinion (LPSTC of pinion) measured along the line of action when the pinion is not undercut.
- p<sub>b</sub>: base pitch

When the gears are not undercut, pinion dedendum and addendum portions of the length of action can be calculated by using the intersection points of the outside circles with the line of action. Contact ratio, and pinion dedendum and addendum portions of the line of action can be calculated by using the following equations [20].

where,

$$Z_a = \frac{\left(d_0^2 - d_b^2\right)^{0.5} - d\left(\sin f^c\right)}{2} \qquad r_{\rm tiG} \le r_{\rm LG} \tag{3.30}$$

$$Z_{b} = \frac{\left(D_{0}^{2} - D_{b}^{2}\right)^{0.5} - D\left(\sin f^{c}\right)}{2} \qquad r_{tiP} \le r_{LP}$$
(3.31)

$$Z_{a} = \frac{D \sin f^{c} - 2 \left[ r_{tiG} - \left(\frac{D_{b}}{2}\right)^{2} \right]^{\frac{1}{2}}}{2} \qquad r_{tiG} > r_{LG} \qquad (3.32)$$

$$Z_{b} = \frac{d \sin f^{c} - 2 \left[ r^{2}_{tiP} - \left(\frac{d_{b}}{2}\right)^{2} \right]^{\frac{1}{2}}}{2} \qquad r_{tiP} > r_{LP} \qquad (3.33)$$

In the above equations, condition  $r_{ti}>r_L$  indicates undercutting.

### 3.14 AGMA Geometry Factor for Bending Strength J

The geometry factor J is calculated from a complicated algorithm defined in AGMA standard 908-B89 which is based on methods given in [3]. The calculation of the geometry factor J is explained in the following sections after the definition of rack cutter geometry, spur gear geometry and their equations. The equation is simply as follows.

$$J = \frac{Y}{K_f} \tag{3.34}$$

#### 3.14.1 Rack Cutter Geometry

In Figure 3.9 rack cutter geometry and in the Table 3.7 its dimensions with respect to the AGMA 908-B89 are given.



Figure 3.9 Rack Cutter Geometry [19]

Rack cutter dimension are found from the following equations.

$$a^{c} = C_{ac} m \tag{3.35}$$

$$\mathbf{b}^{\mathrm{c}} = \mathbf{C}_{\mathrm{b}} \, \mathbf{m} \tag{3.36}$$

$$\mathbf{r}_{\mathrm{f}} = \mathbf{C}_{\mathrm{f}} \, \mathbf{m} \tag{3.37}$$

Standard	$\phi^{c}$	a <sup>c</sup>	b <sup>c</sup>	$r_{\rm f}$	t <sup>c</sup>
AGMA 908-B89	14.5 <sup>°</sup>	1.0 m	1.157 m	0.157 m	$\pi$ m/2
AGMA 908-B89	$20^{\circ}$	1.0 m	1.250 m	0.250 m	$\pi$ m/2
AGMA 908-B89	$25^{\circ}$	1.0 m	1.350 m	0.270 m	$\pi$ m/2

Table 3.7 Rack Cutter Dimensions

Amount of rack cutter offset used to cut gears with non-standard proportions is expressed by using the addendum modification (rack shift) coefficient. Positive coefficients are used for advances. Amount of rack cutter shift is found by multiplying the addendum modification coefficient by the module.

Pinion and gear dimensions are found from the following equations whether they are modified or not. When there is no rack shift then  $x_p$  and  $x_G$  are zero and the equations are valid for standard proportions.

$$d = m N_p \tag{3.38}$$

$$\mathbf{D} = \mathbf{m} \, \mathbf{N}_{\mathbf{G}} \tag{3.39}$$

$$r_{\rm p} = (m N_{\rm p}) / 2$$
 (3.40)

$$r_{\rm G} = (m N_{\rm G}) / 2$$
 (3.41)

$$\begin{split} & d_{o} = m \ N_{p} + 2 \ a^{c} + 2 \ m \ x_{p} & (3.42) \\ & D_{o} = m \ N_{G} + 2 \ a^{c} + 2 \ m \ x_{G} & (3.43) \\ & r_{op} = d_{o} / \ 2 & (3.44) \\ & r_{oG} = D_{o} / \ 2 & (3.45) \\ & d_{r} = m \ N_{p} - 2 \ b^{c} + 2 \ m \ x_{p} & (3.46) \\ & D_{r} = m \ N_{G} - 2 \ b^{c} + 2 \ m \ x_{G} & (3.47) \\ & d_{b} = m \ N_{p} \cos(\phi) & (3.48) \\ & D_{b} = m \ N_{G} \cos(\phi) & (3.49) \\ \end{split}$$

$$p_{b} = \pi \operatorname{m} \cos(\phi) = p^{c} \cos(\phi)$$
(3.50)

$$C = m \frac{N_p + N_G}{2} \tag{3.51}$$

$$c = C - \frac{D_o + D_r}{2}$$

$$(3.52)$$

### **3.14.2 Spur Gear Tooth Geometry**

In order for the fundamental law of gearing to be true, the gear tooth contours on mating teeth must be conjugates of one another. There is infinite number of possible conjugate pairs that could be used, but only a few curves have been practical application as gear teeth. The cycloid is still used as a tooth form in some watches and clocks, but most gears use the involute of a circle for their working portion and the trochoid as the fillet portion.

The involute function is calculated from Equation 3.53.

$$\operatorname{inv}(\phi) = \tan(\phi) - \phi \tag{3.53}$$

Coordinates of involute (working portion) of tooth profile is calculated by the following equations. The coordinate system and tooth profile is given in Figure 3.10.



Figure 3.10 Coordinate system and tooth profile [19]

$$x_i = r_i \sin(\gamma)$$
  $r_{ti} \le r_i \le r_o$  (3.54)

$$y_i = r_i \cos(\gamma)$$
  $r_{ti} \le r_i \le r_o$  (3.55)

where,

$$\gamma = \frac{t}{2r} + inv(f^{c}) - inv(arc(\cos(\frac{r\cos(f^{c})}{r_{i}}))) \qquad r_{ti} \le r_{i} \le r_{o} \qquad (3.56)$$

$$r_{ti} = (x_f^2 + y_f^2)^{0.5}$$
 at  $\theta = \theta_{max}$  (3.57)

Circular tooth thicknesses at pitch circles and tooth tips are calculated by the following equations:

$$t_p = m(\frac{p}{2} + 2x_p \tan(f^c) - backlash)$$
(3.58)

$$t_{G} = m(\frac{p}{2} + 2x_{G}\tan(f^{c}) - backlash)$$
(3.59)

$$t_{tp} = do(\frac{t_P}{d} + inv(f^c) - inv(f_{tP}))$$
(3.60)

$$t_{tG} = Do(\frac{t_G}{D} + inv(f^c) - inv(f_{tG}))$$
(3.61)

where,

$$\phi_{tP} = \operatorname{arc}(\cos(d_b/d_o)) \tag{3.62}$$

$$\phi_{tG} = \operatorname{arc}(\cos(D_b/D_o)) \tag{3.63}$$

According to the AGMA 908-B89, limiting value for the tooth thickness calculated at tooth tip is 0.3 module. During the developed program this is also taken as upper limit for modification. The limit of this approach occurs when the pinion tooth becomes pointed (excessively-thinned tooth tip).

Coordinates of trochoid (fillet portion) of tooth profile is calculated by the following equations [19].

$$x_{f} = \left(r - b^{c} + m \chi + rf\right) \cdot \sin(\beta + \theta) - r \cdot \theta \cdot \cos(\beta + \theta) \dots + \frac{-rf}{\left[\left(b^{c} - m \chi - r_{f}\right)^{2} + r^{2} \cdot \theta^{2}\right]^{0.5}} \cdot \left[\left(b^{c} - m \chi - r_{f}\right) \cdot \sin(\beta + \theta) + r \cdot \theta \cdot \cos(\beta + \theta)\right]$$
(3.64)

$$y_{f} = \left(r - b^{c} + m \chi + rf\right) \cdot \cos(\beta + \theta) + r \cdot \theta \cdot \sin(\beta + \theta) \dots + \frac{-rf}{\left[\left(b^{c} - m \chi - r_{f}\right)^{2} + r^{2} \cdot \theta^{2}\right]^{0.5}} \cdot \left[\left(b^{c} - m \chi - r_{f}\right) \cdot \cos(\beta + \theta) + r \cdot \theta \cdot \sin(\beta + \theta)\right]$$
(3.65)

For both of the above equations  $\theta_{min} \leq \theta \leq \theta_{max}$  .

where,

$$\beta = \frac{p}{N} - \frac{\Delta}{r} \tag{3.66}$$

$$\Delta = \frac{pm}{4} - (b^c - r_f) \tan(f^c) - \frac{r_f}{\cos(f_c)}$$
(3.67)

$$\theta_{\min} = 0 \tag{3.68}$$

$$\theta_{\max} = \frac{b^c - mx - r_f}{r \tan(f_c)}$$
(3.69)

## 3.15 Calculation of Geometry factor J

The AGMA factor J employs a modified value of the Lewis form factor. Y is the tooth form factor obtained from a generated layout of the tooth profile and  $K_f$  is the stress correction factor based on a formula deduced from a photoelastic investigation of stress concentration in gear teeth over 40 years ago.

$$J = \frac{Y}{K_f} \tag{3.70}$$

$$Y = \frac{1}{\frac{\cos(f_{L})}{\cos(f^{c})}(\frac{6h_{f}}{{s_{f}}^{2}} - \frac{\tan(f_{L})}{s_{f}})}$$
(3.71)

$$K_{f} = H + \left(\frac{s_{f}}{r_{f}}\right)^{L} + \left(\frac{s_{f}}{r_{f}}\right)^{M}$$
(3.72)

where,

 $H=0.331 - 0.436 \phi^{c}$ (3.73)

$$L=0.324 - 0.492 \phi^{c}$$
(3.74)

$$M = 0.261 + 0.545 \phi^{c} \tag{3.75}$$

While calculating the geometry factor J, the problem is to determine the critical section parameters ( $s_f$ ,  $h_f$ ,  $\rho_f$ ) from the condition of tangency of the Lewis parabola and the root trochoid (fillet portion), and numerical techniques with iterative procedures are required. This procedure is done by the methods defined in [19].

Height of the Lewis parabola  $h_f$ , and corresponding tooth thickness at critical section  $s_f$  is calculated by finding the coordinates of tangency point of the Lewis parabola and the root.

It is seen from Figure 3.11 that y-coordinate of the vertex of the parabola  $y_L$  can be found by making use of the equation of the line whose slope is  $tan(\phi_L)$  and passing through a point on the involute tooth profile ( $x_i$ ,  $y_i$ ). Values of them are found from Equation 3.76.

$$y_L = y_i - x_i \tan(\phi_L) \tag{3.76}$$

The load angles are found from the following equations according to the diameters at which the load is acting (HPSTC or Tip loading). During the calculation of AGMA

bending stress factor J, HPSTC is considered the most critical point and use HPSTC diameters at the following equations, when the tip loading is the case then outside diameters are used.

$$\phi_{\rm LP} = \frac{(d_x^2 - d_b^2)^{0.5}}{d_b} - (\frac{t_P}{d} + inv(f^c))$$
(3.77)

$$\phi_{\rm LG} = \frac{(D_x^2 - D_b^2)^{0.5}}{D_b} - (\frac{t_G}{D} + inv(f^c))$$
(3.78)

where,

when  $r_{tiP} \leq r_{LP}$ ;

$$d_{\text{HPSTC}} = 2 \left[ \left( \frac{d_b}{2} \right)^2 + \left( \frac{d \sin(f^c)}{2} + Z_e \right)^2 \right]^{0.5}$$
(3.79)

$$Z_{\rm e} = p_b - \frac{1}{2} \left[ D_o^2 - D_b^2 \right]^{0.5} - \left[ D^2 - D_b^2 \right]^{0.5}$$
(3.80)

when  $r_{tiG} \leq r_{LG}$ ;

$$D_{\text{HPSTC}} = 2 \left[ \left( \frac{D_b}{2} \right)^2 + \left( \frac{D \sin(f^c)}{2} + Z_c \right)^2 \right]^{0.5}$$
(3.81)

$$Z_{c} = p_{b} - \frac{1}{2} \left[ d_{o}^{2} - d_{b}^{2} \right]^{0.5} - \left[ d^{2} - d_{b}^{2} \right]^{0.5}$$
(3.82)

when  $r_{tiP} > r_{LP}$ ;

$$d_{\text{HPSTC}} = 2\left\{ \left[ \frac{d_b}{2} \right]^2 + \left[ \left( r^2_{tiP} - \left( \frac{d_b}{2} \right)^2 \right)^{\frac{1}{2}} + P_b \right]^2 \right\}^{\frac{1}{2}}$$
(3.83)

when  $r_{tiG} > r_{LG}$ ;

$$D_{\text{HPSTC}} = 2\left\{ \left(\frac{D_b}{2}\right)^2 + \left[ \left(r_{tiG}^2 - \left(\frac{D_b}{2}\right)^2\right)^{\frac{1}{2}} + P_b \right]^2 \right\}^{\frac{1}{2}}$$
(3.84)



Figure 3.11 Lewis Parabola [19]

An opening down parabola, having its vertex at  $(y_L, 0)$  can be expressed by using Equation 3.85.

$$\mathbf{y} = -\mathbf{c}\mathbf{x}^2 + \mathbf{y}_{\mathrm{L}} \tag{3.85}$$

By taking derivative with respect to the x then, the slope of the parabola is found as follows:

$$m_{\rm p} = \frac{d_y}{d_x} = -2cx \tag{3.86}$$

Slope of the tooth root fillet curve, m<sub>f</sub> is found from Equation 3.87:

$$\mathbf{m}_{\mathbf{f}} = \frac{\frac{d_{y_f}}{dq}}{\frac{d_{x_f}}{dq}}$$
(3.87)

By equating two slopes at the point of tangency  $(x_t, y_t)$ , between the Lewis parabola and the fillet curve and also using the opening down parabola equation,  $\theta$ , c and then critical section parameters are found by using Newton Raphson method [21].

$$y_{f}(\theta) = -(c x_{f}(\theta))^{2} + y_{L}$$
(3.88)

and

$$-2 \operatorname{c} \operatorname{x}_{\mathrm{f}}(\theta) = \frac{\frac{d_{y_f}}{dq}}{\frac{d_{x_f}}{dq}}$$
(3.89)

are two simultaneous nonlinear equations with two unknowns,  $\theta$  and c. They can be expressed in the form of

$$U(\theta,c) = y_{f}(\theta) + (c x_{f}(\theta))^{2} - y_{L} = 0$$
(3.90)
$$V(\theta,c) = -2 c x_{f}(\theta) - \frac{\frac{d_{y_{f}}}{dq}}{\frac{d_{x_{f}}}{dq}} = 0$$
(3.91)

Thus the solution would be the values of c and  $\theta$  that make the functions U( $\theta$ ,c) and V( $\theta$ ,c) equal to zero. Newton-Raphson method used the derivative (slope) of a function to estimate the root. This estimate was based on a first-order Taylor series expansion,

$$\mathbf{x}_{i+1} = \mathbf{x}_i - \frac{f(x_i)}{f'(x_i)}$$
(3.92)

which is the single-equation form for the Newton-Raphson method. A similar multiequation form is used for this study. The denominators of each of these equations are formally referred to as the determinant of the Jacobian of the system. It can be employed iteratively until an acceptable solution is obtained.

In this study initial guesses for c = 0 and  $\theta = \pi/32$ ,

$$\theta_{i+1} = \theta_i - \frac{(u(q,c)\frac{\partial}{\partial c}v(q,c)) - (v(q,c)\frac{\partial}{\partial c}u(q,c))}{(\frac{\partial}{\partial q}u(q,c)\frac{\partial}{\partial c}v(q,c)) - (\frac{\partial}{\partial c}u(q,c)\frac{\partial}{\partial q}v(q,c))}$$
(3.93)

$$c_{i+1} = c_i - \frac{(v(q,c)\frac{\partial}{\partial c}u(q,c)) - (u(q,c)\frac{\partial}{\partial c}v(q,c))}{(\frac{\partial}{\partial q}u(q,c)\frac{\partial}{\partial c}v(q,c)) - (\frac{\partial}{\partial c}u(q,c)\frac{\partial}{\partial q}v(q,c))}$$
(3.94)

After calculation of Equation 3.93 and 3.94, values are used for the Equation 3.90 and 3.91 and this process is done iteratively. Then c and  $\theta$  are ready to be used for the critical section parameters.

$$\mathbf{h}_{\mathrm{f}} = \mathbf{y}_{\mathrm{L}} - \mathbf{y}_{\mathrm{t}} \tag{3.95}$$

$$s_f = 2 x_t$$
 (3.96)

$$\rho = \frac{\left(\left(\frac{\partial x_f}{\partial q}\right)^2 + \left(\frac{\partial y_f}{\partial q}\right)^2\right)^{3/2}}{\left|\frac{\partial x_f}{\partial q}\frac{\partial^2 y_f}{\partial q^2} - \frac{\partial y_f}{\partial q}\frac{\partial^2 x_f}{\partial q^2}\right|}$$
(3.97)

 $\rho_f$  is the minimum radius of curvature occurs at the tangency point of fillet curve and the root circle, so it is found by using  $\theta = 0$  in the Equation 3.97.

# **CHAPTER 4**

### ADDENDUM MODIFICATION FOR GEARS

The most extensively used modification method is the addendum modification for gear manufacturing and it can be applied with standard tools and machines.

Modification of the addendum of the pinion, and in most cases the gear member, is recommended for gears serving the following applications:

- Meshes in which either or both members have small numbers of teeth.
- Meshes operating on non-standard center distances because of limitations on ratio or center distances.
- Meshes of speed-increasing drives.
- Meshes designed to carry maximum power for the given weight allowance. (This type of gearing is usually designed to achieve the best balance in strength, wear, specific sliding, pitting, or scoring.)
- Meshes in which an absolute minimum of energy loss through friction is to be achieved.

Only base circles of gears affect the kinematics of a gear pair which have gears with involute tooth profiles. Addendum modification is based on this property of involute gears. By changing the diameters of pitch circles of the pinion and the gear, they can be run at any center distance. If one of the gears is thought as a cutter with infinite number of teeth, i.e. the rack cutter, the mating gear can be run with this gear at different pitch diameters properly. Also gears manufactured by the same cutter can be run with each other. If the pitch diameter of the gear to be manufactured coincides with the datum line of this cutter, a standard gear is obtained which is called as O-Gear (Gear with no modification). If the cutter is withdrawn from the gear by (x/m), a gear which is called as plus V-Gear is obtained. If withdrawal is made in the opposite direction by (-x/m) i.e., the cutter is advanced towards the gear, a minus V-Gear is obtained. Here, x is called as the addendum modification coefficient. It is positive for withdrawal, negative for advance of cutter. Upper limit of addendum modification is formed by point tooth and lower limit of that is formed by cutter interference.

#### 4.1 Types of Corrected Gear Mechanisms

Based on the correction aspect, gearing can be classified into the following types: O-Gearing, V-Gearing, and V-O-Gearing

- *O-Gearing:* In standard gearing, the gear teeth are generated with standard proportions in the normal manner without any profile correction or modification. Two 0-Gears are used. (Standard gear drive).
- *V-Gearing:* In V-Gearing the sum of the profile corrections of the two mating gears is not equal to zero. However, the sum is positive in almost all cases in order to take advantage of the beneficial effects of positive correction. Usually, the sum is so divided that the pinion gets the bigger share for the positive correction. Sometimes a V-Pinion may mate with a normal, uncorrected gear. It all depends on how the situation warrants it.
- V-O-Gearing: In V-O-Gearing, the mating pair of gears receive numerically equal but opposite values for correction factors. Normally, the smaller gear (pinion) is provided with positive correction and the larger gear with negative correction. Hence in such gears (x<sub>p</sub> + x<sub>G</sub> =0). In these type of gearing, one of the gears is a plus V-Gear and the other is a minus V-Gear with the same amount of addendum modification coefficient. This is also known as "Long and Short Addendum Gearing" in AGMA standards and "V-Null Gearing" in German technical literature.

# 4.1.1 O-Gearing

It is a standard gearing which consists of two O-Gears. These type of gearing is used when the number of teeth is equal to or greater than the minimum number of teeth not to have undercutting and when there are not any special requirements for the center distance and tooth strength. Generating pitch diameters and operating pitch diameters are the same for these gears.

The mechanism is defined as

$$\mathbf{x}_{\mathbf{P}} = \mathbf{x}_{\mathbf{G}} = \mathbf{0} \tag{4.1}$$

$$x_{\rm T} = x_{\rm P} + x_{\rm G} = 0 \tag{4.2}$$

$$\phi = \phi^{c} \tag{4.3}$$

Then the dimensions are as follows:

$$d = m N_p \tag{4.4}$$

$$\mathbf{D} = \mathbf{m} \, \mathbf{N}_{\mathbf{G}} \tag{4.5}$$

$$d_o = m N_p + 2 a^c \tag{4.6}$$

$$D_o = m N_G + 2 a^c \tag{4.7}$$

$$d_r = m N_p - 2 b^c$$
 (4.8)

$$D_r = m N_G - 2 b^c \tag{4.9}$$

 $d_{\rm b} = m N_{\rm p} \cos(\phi) \tag{4.10}$ 

$$\mathbf{D}_{\mathrm{b}} = \mathbf{m} \, \mathbf{N}_{\mathrm{G}} \cos(\phi) \tag{4.11}$$

$$c = C - \frac{D_o + D_r}{2} \tag{4.12}$$

$$t_{\rm p} = m(\frac{p}{2}) \tag{4.13}$$

$$t_{\rm G} = m(\frac{p}{2}) \tag{4.14}$$

$$t_{tp} = do(\frac{t_P}{d} + inv(f) - inv(f_{tP}))$$

$$(4.15)$$

$$t_{tG} = Do(\frac{t_G}{D} + inv(f) - inv(f_{tG}))$$

$$(4.16)$$

$$\phi_{tP} = \operatorname{arc}(\cos(d_b/d_o)) \tag{4.17}$$

$$\phi_{tG} = \operatorname{arc}(\cos(D_b/D_o)) \tag{4.18}$$

$$C = m \frac{N_p + N_G}{2} \tag{4.19}$$

where,

- a<sup>c</sup> : Rack cutter addendum
- b<sup>c</sup> : Rack cutter dedendum
- C : Operating Center distance
- c : Bottom clearance
- D : Gear pitch diameter
- $D_b$  : Gear base diameter
- $D_o \quad : \ Gear \ outside \ diameter$
- $D_r$  : Gear root diameter
- d : Pinion pitch diameter
- $d_b \quad : \ Pinion \ base \ diameter$

- d<sub>o</sub> : Pinion outside diameter
- d<sub>r</sub> : Pinion root diameter
- m : Module
- $N_p$  : Number of teeth for pinion
- $N_G$  : Number of teeth for gear
- t : Circular tooth thickness at standard pitch radius
- $t_t$  : Circular tooth thickness at tooth tip
- x : Addendum modification coefficient (rack shift) coefficient
- $\phi$  : Pressure angle
- $\phi_t$ : Pressure angle at tooth tip
- $\phi^c$  : Rack cutter pressure angle

### **Subscripts**

- P : Pinion
- G : Gear
- m : Modified

#### 4.1.2 V-O-Gearing

In V-O-Gearing, because of the fact that the amount of correction is equal and opposite, their effect is to nullify each other as far as certain dimensions are concerned, so that the two pitch circles contact each other at the point P and the working pressure angle remains as the standard pressure angle as in the case of uncorrected gear. Also, the centre distance remains unaltered and is equal to the sum of the pitch circle radii. However, unlike uncorrected gearing system, the reference line of the reference profile (the basic rack) does not pass through the normal pitch point P in this case. It is shifted away by an amount equal to the numerical value of xm (mm). In case of the pinion, which normally receives the positive correction, the cutter is moved away from the gear blank centre by an extra amount of xm (mm) while cutting, so that an enlarge pinion with its tip diameter increased by an amount of 2xm (mm) is produced. In case of the gear, the cutter is moved towards the gear centre by the same amount, so that its diameter is smaller by an amount of 2xm (mm) than in the case of uncorrected gears.

Since topping is not necessary for V-O-Gearing, the gear blanks are simply made bigger or smaller, as the case may be, by the amount indicated before feeding them to the gear-cutting machine.

The V-O-Gearing is normally meant where the reduction ratio is large. Thicker pinion teeth are ensured and the gear teeth also do not become significantly weak. However, V-O-Gearing is not recommended for small reduction ratios as it tends to weaken the teeth of the gear. The V-O-Gearing is also sometimes recommended where for certain specific reasons the normal tooth-thickness of the gear pair or the specific sliding velocities between the meshing teeth flanks are to be changed. Besides, since normally the pinion teeth are weaker than gear teeth when both are made of the same material, they are more vulnerable to breakage and wear. The, V-O-Gearing system tends to equalise the tooth strength and thereby reduces the susceptibility to such damage.

This type of gearing is recommended mainly for the following cases.

- If the number of teeth is smaller than the minimum number of teeth not to have under cutting, a plus V-Gearing is used in order to avoid interference. In contrast, if center distance is not required to be changed and a number of teeth of the gear is large enough, a minus V-Gearing is used. Thus, the sum of addendum modification coefficients is zero. V-O-Gearing have the same center distances as O-Gearing. Generating pitch diameters of the gears are tangent to each other but not coincident with the datum line of cutter.
- If pinion and gear strengths are required to be balanced this type of modification can also be applied.

The mechanism is defined as:

$$\mathbf{x} = \mathbf{x}_{\mathbf{P}} = -\mathbf{x}_{\mathbf{G}} \tag{4.20}$$

$$x_{\rm T} = x_{\rm P} + x_{\rm G} = 0 \tag{4.21}$$

$$\phi = \phi^{c} \tag{4.22}$$

Then dimensions are as follows:

$$d = m N_p \tag{4.23}$$

$$\mathbf{D} = \mathbf{m} \, \mathbf{N}_{\mathbf{G}} \tag{4.24}$$

$$d_{o} = m N_{p} + 2 a^{c} + 2 m x_{p}$$
(4.25)

$$D_{o} = m N_{G} + 2 a^{c} + 2 m x_{G}$$
(4.26)

$$d_{\rm r} = m N_{\rm p} - 2 b^{\rm c} + 2 m x_{\rm p} \tag{4.27}$$

$$D_{\rm r} = m N_{\rm G} - 2 b^{\rm c} + 2 m x_{\rm G}$$
(4.28)

$$d_{\rm b} = m \, N_{\rm p} \cos(\phi) \tag{4.29}$$

$$D_{b} = m N_{G} \cos(\phi) \tag{4.30}$$

$$c = C - \frac{D_o + D_r}{2}$$
 (4.31)

$$t_p = m(\frac{p}{2} + 2x_P \tan(f))$$
 (4.32)

$$t_{\rm G} = m(\frac{p}{2} + 2x_{\rm G}\tan(f)) \tag{4.33}$$

$$t_{tp} = do(\frac{t_P}{d} + inv(f) - inv(f_{tP}))$$

$$(4.34)$$

$$t_{tG} = Do(\frac{t_G}{D} + inv(f) - inv(f_{tG}))$$
(4.35)

$$\phi_{tP} = \operatorname{arc}(\cos(d_b/d_o)) \tag{4.36}$$

$$\phi_{tG} = \operatorname{arc}(\cos(D_b/D_o)) \tag{4.37}$$

$$C = m \frac{N_p + N_G}{2} \tag{4.38}$$

### 4.1.3 V-Gearing

On this gearing, generating pitch diameters of the gears are not tangent to each other while running. So, center distances of these mechanisms are different form V-O-gearing and O-gearing. If a center distance other than the standard center distance is required, this type of modification can be used.

The mechanism is defined as:

$$x_P \neq -x_G \tag{4.39}$$

$$\mathbf{x}_{\mathrm{T}} = \mathbf{x}_{\mathrm{P}} + \mathbf{x}_{\mathrm{G}} \neq \mathbf{0} \tag{4.40}$$

$$\phi \neq \phi^{c} \tag{4.41}$$

For this case the generating pitch diameters are not tangent to each other. A nonzero backlash term  $\beta$  should be used in the equations when it is desired to have a circumferential backlash B during meshing. The modified pressure angle ( $\phi^m$ ), backlash term ( $\beta$ ), and total addendum modification coefficient ( $x_T$ ) is found as follows

$$f^{m} = \arccos(\frac{C^{s} \cos(f^{c})}{C^{m}}$$
(4.42)

$$\beta = \frac{B}{m(N_p + N_G)} \frac{\cos(f^c)}{\cos(f^m)}$$
(4.43)

$$x_T = (inv(f^m) - inv(f^c) - b)\frac{N_p + N_G}{2}$$
(4.44)

Allocation of the sum of the total addendum modification coefficients on pinion and gear teeth is will be made for the desired performance criteria and this case is beyond the scope of this study.

#### **4.2 Applications of Addendum Modification**

In this study, addendum modification is applied for three main reasons, namely to have compact gears, to have gears with small numbers of teeth without undercutting, and to have gears with better balance of strength between the pinion and the gear. Increasing that addendum modification of teeth makes the gear teeth stronger. Decrease in the addendum modification of teeth has the reverse on strength of teeth.

#### **4.3 Interference and Undercutting**

The involute tooth form is only defined outside of the base circle. In some cases the dedendum will be large enough to extend below the base circle. If so, then the portion of tooth below the base circle will not be an involute and will interfere with the tip of the tooth on the mating gear, which is an involute. If the gear is cut with a standard gear shaper or a hob the cutting tool will also interfere with the portion of tooth below base circle and will cut away the interfering material. This results in an undercut gear. Undercutting weakens the tooth by removing material at its root. The maximum moment and maximum shear from the tooth loaded as a cantilever beam both occur in this region. Severe undercutting will cause early tooth failure. To prevent undercutting,  $x_{gmin}$ , must be equal to or greater than the expression in Equation 4.45, derived from the Figure 4.1;

$$x_{g\min} = h_{ao} - r_{ao} (1 - \sin f) - \frac{n}{2} \sin^2 f$$
(4.45)

where,

- $x_{gmin}$ : the generating rack shift coefficient
- hao: nominal tool addendum
- $\rho_{ao}$ : tool tip radius
- n : pinion or gear number of teeth



Figure 4.1 Undercutting Criteria [2]

Interference and its attendant undercutting can be prevented simply by avoiding gears with too few teeth. If a pinion has a large number of teeth, they will be small compared to its diameter. As the number of teeth is reduced for a fixed diameter pinion, the teeth must become larger. At some point, the dedendum will exceed the radial distance between the base circle and the pitch circle, and interference will occur. If the theoretical limit radius of a gear, which is the radius at which involute profile on a gear should start in order to make use of the full length of the involute profile of the mating gear, is smaller than the radius at which involute tooth profile starts, involute interference occurs. As mentioned before when gear teeth are produced by a generating process using a rack cutter or hobbing, interference is automatically eliminated since the cutting tool removes the interfering portion. This results in a reduced contact ratio. Theoretical limit radii for the pinion and the gear, which are calculated by using gear kinematics, are given below [22].

$$r_{LP} = \frac{1}{2} \left[ \left( d_b \right)^2 + \left[ \left( d + D \right) \sin f - \left( D_o^2 - D_b^2 \right)^{\frac{1}{2}} \right]^2 \right]^{\frac{1}{2}}$$
(4.46)

$$r_{LG} = \frac{1}{2} \left[ \left[ D_b \right]^2 + \left[ \left( d + D \right) \sin f - \left( d_o^2 - d_b^2 \right)^{\frac{1}{2}} \right]^2 \right]^{\frac{1}{2}}$$
(4.47)

# **CHAPTER 5**

## **MULTISTEP GEAR BOX DESIGN**

#### 5.1 Overview

A gear train is any collection of two or more meshing gears. Single stage gear drive (a pair of gears) is the simplest form of gear train and limited to a ratio from 1:1 to 8:1. Ratios of 10:1 are also possible for ordinary spur gear design practices [23]. To get a train ratio greater than 10:1, it is necessary to compound the train.

Until now, the studies on the design of gear drives have been focused on single stage gear drives. Nowadays, with increasing the applications of the gear drives in high speed and small space, the need for designing multi-stage gear drives become an important interest subject. A number of complicated problems should be considered in the design of multi-stage gear drives which don't exist in the single-stage gear drives. These problems are summarized as follows:

- Determination of the number of reduction stages,
- Determination of the gear ratios of each stage,
- Determination of dimension of gears,
- Configuration design of the gear drive elements.

#### **5.2 Determination of the Number of Reduction Stages**

The number of reduction stage is input from the designer considering the overall gear ratio, available space, and other design specifications. Some simple recommendations are proposed the number of reduction stages. As explained before, it is recommended to handle gear ratios from 1:1 to 8:1 (max. 10:1) in a single

reduction for spur gear design practices [23]. It is recommended by AGMA to add another stage if the gear ratio of stage is greater than 5:1 [24]. Then the number of reduction stages is determined by the intuition of the designer. Where larger ratios are needed, the choice of gear arrangement is determined by finding the arrangement with the fewest number of parts that will do the job adequately for the compactness case.

In this study the number of reduction stage is an input from the user. The chance is given to the user to change the range of gear ratios of each stage by changing the number of teeth of pinion and gear. Thus a given overall gear ratio will be allocated between diverse number of stages. And also the upper and lower limits of each stage will be defined by the user.

### 5.3 Determination of the Gear Ratios of Each Stage

## 5.3.1 Determination of Number of Teeth in Each Gears

The gear ratio of a gear train can be easily found if the number of teeth in each gear is known.

$$m_{\rm G} = \frac{N_{G_1}}{N_{P_1}} \frac{N_{G_2}}{N_{P_2}} \frac{N_{G_3}}{N_{P_3}} \frac{N_{G_4}}{N_{P_4}} \dots$$
(5.1)

where,

 $m_G$ : Overall gear ratio  $N_P$ : Number of teeth of pinion  $N_G$ : Number of teeth of gear

To explain the situation easily, a two-stage gear drive is shown in Figure 5.1. The overall gear ratio of the two-stage gear train is found from Equation 5.2 easily.

$$m_{\rm G} = \frac{N_{G_1}}{N_{P_1}} \frac{N_{G_2}}{N_{P_2}}$$
(5.2)



Figure 5.1 Two-stage Gear Train

The inverse problem that of finding the number of teeth of each gear that satisfies the overall gear ratio is much more difficult. This problem is defined in literature as an optimization problem [25]. The objective of the gear train design is to find the number of teeth in each of the gears to minimize the error between the obtained gear ratio and the required gear ratio. Since the number of teeth must be integers, all variables are strictly integers. The optimization problem formulation for two-stage gear drive is as follows:

minimize 
$$f(N) = m_G - \frac{N_{G_1}}{N_{P_1}} \frac{N_{G_2}}{N_{P_2}} \dots \frac{N_{G_i}}{N_{P_i}}$$
 (5.3)

subject to:

$$N_{p\min} \le N_{p_i} \le N_{p\max} \tag{5.4}$$

$$N_{G\min} \le N_{G_i} \le N_{G\max} \tag{5.5}$$

all N<sub>i</sub> (number of teeth) values must be integer.

Continued or chain fractions can be used to determine the number of teeth for the pinion and gear in order to have a required gear ratio with a permissible error [26]. This method is not applicable to code manually in a program to the multi-stage gear drives because of increasing number of variables.

A numerical method is given to determine the number of teeth for the pinion and gear for two-stage gear drive satisfying the required ratio [27]. In this study, this algorithm is modified and expanded up to six stages satisfying the gear ratio with the required precision.

 $m_G$  = overall gear ratio eps = ± required precision on overall gear ratio

 $m_{Gmin} = m_G - eps \tag{5.6}$ 

$$m_{Gmax} = m_G + eps \tag{5.7}$$

$$\mathbf{P} = \text{Int}((\mathbf{N}_{p1}\mathbf{N}_{p2}...,\mathbf{N}_{pi}) \mathbf{m}_{\text{Gmax}})$$
(5.8)

$$Q = Int((N_{p1}N_{p2}....N_{pi}) m_{Gmin}) + 1$$
(5.9)

where,

i : Number of stages  $m_{Gmin}$  = Allowable minimum overall gear ratio  $m_{Gmax}$  = Allowable maximum overall gear ratio

A flowchart of the algorithm for three-stage gear drive is shown in Figure 5.2.



Figure 5.2 Flowchart to split overall gear ratio for three stage gear drive

### 5.3.2 Gear Ratio Constraints

No definite rule has been proposed to determine the gear ratios. The method recommended by Niemann is a practical one, in which gear ratios are based on the Hertz contact stress formula. This application is limited to the design of two-stage and three-stage gear drives [28]. Also with this method, to allocate the overall gear ratio one must previously determine the number of teeth and module [29].

In this study allocation of overall gear ratio to the required number of reduction stages (maximum 6-stage) is done with respect to the following premises:

- Each gear ratio must be greater than the successive gear ratio. In the developed program the option is given to the user to select any gear ratio also to be equal to each other to increase the alternatives,
- The gear ratios must be in such an order that decreases from the input of the gear drive to the output of the gear drive,
- Each gear ratio must not be an integer to avoid the same teeth coming in contact periodically. In the developed program the option is given to the user to select any gear ratio to be an integer to increase the alternatives.

#### **5.4 Determination of Design Parameters**

The design of gears usually requires some iteration. Not enough information exists in the problem to directly solve for the unknowns. The values of some parameters must be assumed and a trial solution done.

Usually, the power and speed are defined at the beginning of the problem. The parameters to be determined are the dimensions of the gear arrangements such as module, diameter and face width. Then the objective is to select the minimum volume gear pairs that satisfy the required gear ratio considering the performance criterion. Total gear volume is found from the following equation:

$$V = \sum_{i=1}^{n} F_i C_i^2 \frac{N_{pi}^2 + N_{Gi}^2}{(N_{pi} + N_{Gi})^2}$$
(5.10)

In order not to check all possible modules, a preliminary design is made to find a provisional value for module [30].

$$m = \left(\frac{2TK_a K_s K_m}{N_p^2 K_v J_p m_a S_{at}}\right)^{1/3}$$
(5.11)

where,

m : Module, (mm)

T : Transmitted torque,(Nmm)

m<sub>a</sub>: Aspect ratio

Since the transmitted power and input speed are known from the beginning then transmitted torque is found from the below equation:

$$T = \frac{30000.P}{p.n_p}$$
(5.12)

where,

P : Transmitted power, (W) n<sub>p</sub> : Pinion speed, (rpm)

Since the number of teeth of pinion and gear for each stage is known from the previous steps, aspect ratio is found from Equation 5.13 as defined in [31].

$$m_a = \frac{mg}{mg+1} \tag{5.13}$$

where,

$$\mathbf{m}_{\mathrm{G}} = \mathbf{N}_{\mathrm{G}} / \mathbf{N}_{\mathrm{P}} \tag{5.14}$$

Provisional face width is found from the Equation 5.15 as defined in [31]:

$$F = \frac{m_g}{m_g + 1} m N_p \tag{5.15}$$

AGMA rating factors in the preliminary module equation are known except the  $K_v$  value.

A provisional value for  $K_v = 0.7$  is selected for initial value as recommend by AGMA [21].

Allowable bending stress number is found from the Equation 5.16:

$$S_{at} = \frac{S_t K_L}{K_T K_R} \tag{5.16}$$

AGMA geometry factor J is found from a complicated algorithm explained in the previous chapters. The addendum modification coefficients of pinion and gear are determined automatically by equalizing the ratios of bending strengths and geometry factors between pinion and gear.

$$\frac{J_P}{J_G} = \frac{S_{atG}}{S_{atP}}$$
(5.17)

The preliminary module found by the above equation is rounded to a standard module according to ISO standard [16].

For all stages the above equations are applied. Than with the preliminary module value, all pinion and gear dimensions are found. With this values AGMA contact stress check is made for all stages with the following equation

$$C_{P} \left(\frac{W_{t}C_{a}}{C_{V}} \frac{C_{S}}{Fd} \frac{C_{m}C_{f}}{I}\right)^{1/2} \leq \frac{S_{C}C_{L}C_{H}}{C_{T}C_{R}}$$
(5.18)

This control is made until the equation above is satisfied with changing the module value to a successive module value. Since the module is changed all related variables are recalculated and then contact stress check is made.

The contact stress check is made for all the stages, and then with this module value, bending stress check is made for all the pinion and gear from the Equation 5.19:

$$\frac{W_t K_a}{K_v} \frac{1.0}{Fm} \frac{K_s K_m}{J} \le \frac{S_t K_L}{K_T K_R}$$
(5.19)

After ensuring all the loading cases, the safety factors are calculated. The safety factors against bending failure are found by comparing the allowable bending stress number to the calculated bending stress for the pinion and gear for all stages:

$$\mathbf{n}_{\mathrm{b}} = \frac{S_{at}}{S_{b}} \tag{5.20}$$

The safety factors against surface failure are found by comparing the allowable contact stress number to the calculated contact stress for the pinion and gear for all stages:

$$\mathbf{n}_{c} = \frac{S_{ac}}{S_{c}} \tag{5.21}$$

Then if the resulting safety factors are either too large or too small, the assumed values are adjusted and the calculation repeated until it converges to an acceptable solution in order to keep package size small.

In this study because the case is multi-stage, for all possible gear pairs satisfying the loading conditions and kinematic analysis, calculations are repeated to find the optimum values (having desired proportions for the lengths of the dedendum and addendum portions of the line of action, having sufficient bottom clearance, having minimum contact stresses, having minimum bending stresses, having balanced pinion and gear bending strength, etc.) with the user interaction.

The shafts are required to transmit the torque, ignoring the bending stress in this study for simplification. Shafts are assumed to be made of alloy steel. There is no axial loading on the shaft as the gears are spur gears. The allowable shear stress for alloy steel is selected as 150 MPa. The radius of the shaft is calculated using the torsion only by Equation 5.22:

$$\mathbf{r}_{s} = \sqrt[3]{\frac{2T}{pt_{all}}}$$
(5.22)

where,

T : Transmitted torque, (Nmm)

 $\tau_{all}$  : Allowable shear stress for shaft, (MPa)

r<sub>s</sub>: Radius of shaft, (mm)

#### **5.5** Configuration Design of the Gear Drive Elements

#### 5.5.1 Definition of an Optimization Model

In general, an optimization model consists of the following three items:

- Objective Function The objective function is a formula that expresses exactly what the function is to be optimized (minimization or maximization). In this study objective function is selected in order to minimize the volume of gearbox.
- Variables Variables are the quantities that are to be taken under control. The decision is made by the designer to choose the variables. For this reason, variables are sometimes also called decision variables. The goal of optimization is to find the values of a model's variables that generate the best value for the objective function, subject to any limiting conditions placed on the variables. In this study the variables are selected as the coordinates of gear arrangements in the three-dimension (x, y, z).
- Constraints Almost without exception, there must be some limit on the values of the variables in a model. These limits are expressed in terms of formulas that are a function of the model's variables. These formulas are referred to as constraints because they constrain the values the variables can take.

The configuration design is made to minimize the volume of a gearbox by using the model that defined in [32, 8]. Since pitch diameter, outside diameter, and face width is determined from the previous design steps by the developed program, the configuration design is considered as a problem of packing gears of fixed size in three-dimensional space. The formulation of the problem is defined as:

$$f = \sum_{i=2}^{k} CD_{(pinion_1 - gear_i)}$$
(5.23)

where,

- f : Objective function
- k : Number of stages
- CD: Center Distance

### **5.5.2 Center Distance Constraints**

The center distance constraints are used to provide proper meshing of gear pair (pinion and gear) of a stage. Since they are equal to zero then they are represented as equality constraint. It is shown in Figure 5.3.



Figure 5.3 Center distance constraint [32]

$$\sqrt{\left(x_{(2i-1)} - x_{(2i)}\right)^{2} + \left(y_{(2i-1)} - y_{(2i)}\right)^{2}} - \frac{d_{(2i-1)} + d_{(2i)}}{2} = 0$$
(5.24)  
$$z_{(2i-1)} - z_{(2i)} = 0$$
  
$$i=1,2... \text{ to n}$$

where,

n : total number of stages

d : pitch diameter of pinion and gear

(x, y, z): the center position of pinion and gear

the odd subscripts are valid for pinion and the even subscripts are for gear.

### **5.5.3 Gear Distance Constraints**

The gear distance constraints are used for mating pinion and gear having same shaft. Since they are equal to zero then they are represented as equality constraint. It is shown in Figure 5.4.



Figure 5.4 Gear distance constraint [32]

$$\left|z_{(2i)} - z_{(2i+1)}\right| - \frac{b_{(2i)} + b_{(2i+1)}}{2} = 0$$
(5.25)

$$x_{(2i)} - x_{(2i+1)} = 0 (5.26)$$

$$y_{(2i)} - y_{(2i+1)} = 0 (5.27)$$

where, b : face width of pinion and gear

### **5.5.4 Gear Interference Constraints**

The gear interference constraints are included to avoid interferences between gears. Since they are not equal to zero then they are represented as inequality constraint. It is shown in Figure 5.5.



Figure 5.5 Gear interference constraint [32]

$$\sqrt{\left(x_{(2i-1)} - x_{(2j-1)}\right)^{2} + \left(y_{(2i-1)} - y_{(2j-1)}\right)^{2}} - \frac{d_{O(2i-1)} + d_{O(2j-1)}}{2} > 0 \quad (5.28)$$

$$\sqrt{\left(x_{(2i-1)} - x_{(2j)}\right)^{2} + \left(y_{(2i-1)} - y_{(2j)}\right)^{2}} - \frac{d_{O(2i-1)} + d_{O(2j)}}{2} > 0$$
(5.29)

$$\sqrt{\left(x_{(2i)} - x_{(2j-1)}\right)^{2} + \left(y_{(2i)} - y_{(2j-1)}\right)^{2}} - \frac{d_{O(2i)} + d_{O(2j)}}{2} > 0$$
(5.30)

$$\sqrt{\left(x_{(2i)} - x_{(2j)}\right)^{2} + \left(y_{(2i)} - y_{(2j)}\right)^{2}} - \frac{d_{O(2i)} + d_{O(2j)}}{2} > 0$$
(5.31)

where,  $d_{\mbox{\scriptsize o}}$  : outside diameter of pinion and gear

# **5.5.5 Shaft Interference Constraints**

The shaft interference constraints are included to avoid interferences between gears and shafts. Since they are equal not equal to zero then they are represented as inequality constraint. It is defined as follows:

$$\sqrt{\left(x_{s(i)} - x_{(2j)}\right)^{2} + \left(y_{(i)} - y_{(2j)}\right)^{2}} - \frac{d_{s(i)} + d_{o(2j)}}{2} > 0$$
(5.32)

i = 1, 2...to ns-2j = i+1, i+2...to n

$$\sqrt{\left(x_{s(i)} - x_{(2j)}\right)^{2} + \left(y_{s(i)} - y_{(2j)}\right)^{2}} - \frac{d_{s(i)} + d_{o(2j)}}{2} > 0$$
(5.33)

$$\sqrt{\left(x_{s(i)} - x_{(1)}\right)^{2} + \left(y_{s(i)} - y_{(1)}\right)^{2}} - \frac{d_{s(i)} + d_{O(1)}}{2} > 0$$
(5.34)

i = 3, 4...to ns

where,

 $d_s$  : pinion and gear shaft diameter ( $x_s$ ,  $y_s$ ,  $z_s$ ) : center position of shaft ns : number of shafts

Since the number of constraints increases considerably with a larger number of stages, it is inefficient and error-prone to code them manually in the program. For example there are 92 constraints existing in a six-stage gear drive. Thus, this nonlinear constrained optimization problem is solved using a software package Lingo 8.0 with the DLL (Dynamic Link Library) interface together with the Visual Basic program automatically.

There are two options in the developed program for gearbox configuration.

• *Compact Gearbox Design:* The objective function together with all the constraints described above are necessary for this configuration design.

• *In-line Gearbox Design:* All the constraints are also necessary for this type of configuration but there is no need to construct an objective function. Extra slope constraints exist in this design to arrange the gear pairs in line.

In the developed program, 3mm clearance is intentionally given for the inequality constraints to provide gap between gears.

#### 5.6 Lingo 8.0 Optimization Software

LINGO is a simple tool for utilizing the power of linear and nonlinear optimization to formulate large problems coincisely, solve them, and analyze the solution. In this study, the objective function is minimized for the total volume of gearbox.

Optimization problems are often classified as linear or nonlinear, depending on whether the relationships in the problem are linear with respect to the variables. In this study since most of the constraints are nonlinear, with the usage of nonlinear solver of Lingo 8.0, the equations are solved simultaneously satisfying the constraints.

If the model does not have any nonlinear constraints, then any locally optimal solution will also be a global optimum. Thus, all optimized linear models will terminate in the global optimum state. If, on the other hand, as in this study, the model has one or more nonlinear constraints, then any locally optimal solution may not be the best solution to the model. There may be another "extremum" that is better than the current one.

That is to say, there is no guarantee to reach a global minimum unless the problem is continuous and has only one minimum. Starting the optimization from a number of different starting points can help to locate the global minimum or verify that there is only one minimum.

Lingo uses SLP (Successive Linear Programming) directions and GRG (Generalized Reduced Gradient) strategy programming with its default settings and there are four other options exist in the software as shown in Figure 5.6.

INGO Options Interface General Nonlinear Solver Integer Pre-Solv	Solver   Linear Solver ver   Integer Solver   Global Solver
Initial Nonl Feasibility Tol:	Einal Nonl Feasibility Tol:
Nonlinear Optimality Tol:	Slow Progress Iteration Limit
Derivatives:     Numerical     Analytical	Strategies: Crash Initial Solution Quadratic Recognition Selective Constraint Eval SLP Directions Steepest Edge
Help	Save Uygula <b>QK</b>

Figure 5.6 Lingo options

• Crash Initial Solution

If the Crash Initial Solution box is checked, Lingo's nonlinear solver will invoke a heuristic for generating a "good" starting point when you solve a model. If this initial point is relatively good, subsequent solver iterations should be reduced along with overall runtimes.

# • Quadratic Recognition

If the Quadratic Recognition box is checked, LINGO will use algebraic preprocessing to determine if an arbitrary nonlinear model is actually a quadratic programming (QP) model. If a model is found to be a QP model, then it can be passed to the faster quadratic solver. Note that the QP solver is not included with the standard, basic version of LINGO, but comes as part of the barrier option.

# • Selective Constraint Evaluation

If the Selective Constraint Eval box is checked, Lingo's nonlinear solver will only evaluate constraints on an as needed basis. Thus, not every constraint will be evaluated during each iteration. This generally leads to faster solution times, but can also lead to problems in models that have functions that are undefined in certain regions.

#### SLP Directions

If the SLP Directions box is checked, Lingo's nonlinear solver will use successive linear programming to compute new search directions. This technique uses a linear approximation in search computations in order to speed iteration times. In general, however, the number of total iterations will tend to rise when SLP Directions are used.

#### • Steepest Edge

If the Steepest Edge box is checked, Lingo's nonlinear solver will use the steepest-edge strategy when selecting variables to iterate on. When LINGO is not in steepest-edge mode, the nonlinear solver will tend to select variables that offer the highest absolute rate of improvement to the objective, regardless of how far other variables may have to move per unit of movement in the newly introduced variable. The problem with this strategy is that other variables may quickly hit a bound, resulting in little gain to the objective.

With the steepest-edge option, the nonlinear solver spends a little more time in selecting variables by looking at the rate that the objective will improve relative to movements in the other nonzero variables. Thus, on average, each iteration will lead to larger gains in the objective. In general, the steepest-edge option will result in less iteration. However, each iteration will take longer.

There is also a Global Solver option available in the Lingo 8.0. If the Global Solver box is checked, LINGO will invoke the global solver to solve a model. Many nonlinear models are non-convex and/or non-smooth. Nonlinear solvers that rely on local search procedures will tend to do poorly on these types of models. Typically, they will converge to a local, sub-optimal point that may be quite distant from the true, global optimal point. Global solvers overcome this weakness through methods of range bounding (e.g., interval analysis and convex analysis) and range reduction techniques (e.g., linear programming and constraint propagation) within a branchand-bound framework to find the global solutions to non-convex models. The only drawback to the global solver is that it runs considerably slower than the default local solver. Therefore, the preferred option is to always try and write smooth, convex nonlinear models, so the faster, default local solver can successfully solve them.

### 5.7 Gearbox Volume Calculation

Gearbox volume is calculated after the configuration design by the following methods.

### 5.7.1 Enclosed Gearbox Volume Calculation

Enclosed volume is calculated by finding the intersection points between the y lines and the circles (meshing gears). It is made by the following equation:

$$V_{enclosedvolume} = (z_{\max} - z_{\min}) \int_{y=y_{\min}}^{y=y_{\max}} (x_{\max} - x_{\min}) dy$$
(5.35)



Figure 5.7 Enclosed Gearbox Volume Calculation

# 5.7.2 Prismatic Gearbox Volume Calculation

Prismatic volume is calculated simply by multiplying the width, height and the depth of the gearbox. It is made by the following equation:



$$V_{prismaticvolume} = (x_{max} - x_{min})(y_{max} - y_{min})(z_{max} - z_{min})$$
(5.36)

Figure 5.8 Prismatic Gearbox Volume Calculation

# CHAPTER 6

# A SAMPLE APPLICATION WITH DEVELOPED SOFTWARE

In this chapter, a sample application is solved with developed software, which is originally introduced by Chong et al. [32]. Table 6.1 shows the design results and Figure 6.1 shows two and three-dimensional representation of the gears according to that study.

Stage	1	2	3	4
Module (mm)	1.25	2.0	2.5	4.0
Number of teeth in pinion	19	18	24	24
Number of teeth in gear	97	74	92	88
Gear ratio	5.105	4.111	3.833	3.667
Pitch diameter of pinion (mm)	23.75	36.0	60.0	96.0
Pitch diameter of gear (mm)	121.25	148.0	230.0	352.0
Face width factor	14	15	14	15
Face width (mm)	18.75	28	37.5	56.0
Volume of gearbox (mm <sup>3</sup> )	22.69 x 10 <sup>6</sup>			

# Table 6.1 Design result, four-stage gear drive [32]



Figure 6.1 Configuration design of the gear drive, front and isometric view [32]

#### CASE:

Design a four-stage spur gear drive with transmitted power of 8 kW at an input speed of 6000 rpm and total gear ratio of 300( $\pm$  0.0001). 55 HRC carburized and case hardened steel gears are used. Face width should be between 4 to 15 module. To account dynamic loads AGMA Quality of number 11 should be used. Straddle mounted with unmodified leads, (S/S<sub>1</sub>) < 0.175, commercial enclosed gear units and gearing adjusted at assembly selections. Pinions and gears are cut with a rack cutter or a hob that has a tool edge radius of 0.25 m. For all stages select minimum number of teeth for pinion as 14 and maximum number of teeth for pinion as 25 and minimum number of teeth for gear as 70 and maximum number of teeth for gear as 85. Tooth thinning for backlash is 0.024 module both for pinion and gear as advised by AGMA. Pinion should run minimum 10<sup>7</sup> cycles on this commercial application. Both preferred and second choice modules are to be used to increase the alternatives. Power source is a multi-cylinder engine and the driven machine is the textile machinery. Overall reliability is selected as 0.99. Lubricant temperature is 120° C. Power loss through the gear train can be assumed negligible.

Transmitted Power (kW)	8
Total Gear Ratio (± 0.0001)	300
Input Speed (rpm)	6000
Gear Type	External Spur
Pressure Angle (°)	20
Modification	Exist
Material	Steel
Heat Treatment	Carburised and Case Hardened
Material Hardness	55.0 HRC
AGMA Quality Number	11
Load Cycles	$10^{7}$
Tool Edge Radius (module)	0.25

Table 6.2 Specifications for design example

The problem is input to the program by the user as follows:

	RBOX DESIGN		
File Input Output Help			
Power Requirement and Tooth,Cutter Geometry	Material	Factors	
Power			
Transmitted Power(kW):	8		
Input Speed (rpm):	6000		
Overall Gear Ratio:	300		
Required Precision( ± ):	0.0001 💌		
Number of Stages:	four-stage 💌		
Power Source:	Light Shock(Multicyl	inder engine) 📃 💌	
Driven Machine:	Moderate shock(Cor	ncrete mixers,textile machiner,	y) 💌
AGMA Quality Number(Qv):	11 💌	Single Pair Reliability:	0,99
Number of Load Cycles:	10000000	Overall Reliability:	0,9606
Cutter Geometry		Standard Modules	
Rack Cutter or Hob		C Preferred	
Rack Cutter Pressure Angle:	20 🗸	Preferred and Second	Choice
Tool Edge Radius (* m):	0,25	Gearbox Design Type Compact Gearbox 💌 🔤	Options OK
2		In-line Gearbox	

Figure 6.2 Input form, power requirement, tooth, and cutter geometry
Options are given to the user for design parameters are as follows:

🖻 MU	LTISTEP SPUR GEARBOX DESIGN	×
File In	put Output Help	
Po <del>w</del> e and	_ Number of Teeth	
Pow	Pinion1: 14 25	Gear1: 70 85
Tran Inpul	Pinion2: 14 25	Gear2: 70 85
Over	Pinion3: 14 25	Gear3: 70 85
Requ	Pinion4: 14 25	Gear4: 70 85
Powe	Pinion5: 14 20	Gear5: 30 70
Drive	Pinion6: 14 20	Gear6: 30 70
AGM Num	Face Width Limits Lower Limit(module): 4	Backlash (module): 0.024
	Upper Limit(module): 15	Factor of safety: 1
Rack	C Standard © Nonstandard	Default Cancel OK
Tool		

Figure 6.3 Options form

Gear and Pinion materials are selected from material page as follows, option is given to the user to select different materials for each pinion and gear, and also hardness values are changeable when necessary. And the material is selected as follows:

HULTISTEP SPUR GE	ARBOX DESIGN		X
Power Requirement and Tooth,Cutter Geometry	Material	Factors	
Stage-4	Stage-5 Stage-1	Stage-6	Stage-3
Gear Material: Heat Treatment:		Steel	
Min.Hardness at suface Grade:	e: [i	55 HRC	
AGMA bending strengt AGMA surface fatigue	h(MPa): [: strength(MPa): [ <sup>-</sup>	380	
Young Modulus of Elas Poisson's Ratio:	sticity,Sat(MPa): [	200000	
			Help
55		2	

Figure 6.4 Input form, selected material

Mounting type is selected as straddle mounting with its options follows:

File       Input       Output       Help         Power Requirement and Tooth,Cutter Geometry       Material       Factors         S       STRADDLE MOUNTING       Image: Construction Factor(Cmc)       Printon Proportion Modifier(Cpm)         S       Image: Construction Factor(Cmc)       Printon Proportion Modifier(Cpm)       Straddle mounted pinions width         S       Image: Construction Factor(Cma)       Straddle mounted pinions width       Image: Construction Factor(Ce)         S       Image: Construction Factor(Cma)       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         D       Open Gearing       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         P       Precision Enclosed Gear Units       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         P       Precision Enclosed Gear Units       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         P       Precision Enclosed Gear Units       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         P       Precision Enclosed Gear Units       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         Image: Construction Enclosed Gear Units       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         Image: Constructin Enclosed Gear Units       Image: Construction Factor(Ce	MULTISTEP SPUR	GEARBOX DESIGN		×
Power Requirement and Tooth,Cutter Geometry       Material       Factors         STRADDLE MOUNTING       Image: Construction Factor(Cmc)       Pinion Proportion Modifier(Cpm)         Staddle mounted pinions width       Image: Construction Factor(Cma)       Staddle mounted pinions width         Image: Construction Factor(Cma)       Image: Construction Factor(Cma)       Staddle mounted pinions width         Image: Construction Factor(Cma)       Image: Construction Factor(Cma)       Image: Construction Factor(Cma)         Image: Construction Factor(Cma)       Image: Construction Factor(Cma)       Image: Construction Factor(Ce)         Image: Construction Factor(Cma)       Image: Construction Factor(Cma)       Image: Construction Factor(Ce)         Image: Construction Factor(Cma)       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         Image: Construction Factor(Cma)       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         Image: Construction Factor(Cma)       Image: Construction Factor(Ce)       Image: Construction Factor(Ce)         Image: Construction Factor Fac	File Input Output He	Þ		
STRADDLE MOUNTING       Image: Strandbalance of the strategy of the st	Power Requirement and Tooth,Cutter Geometry	Material	Factors	
	STRADDLE MOUNTI     A     Lead Correction Fa     Correction     Modified Lead:     correction     Mesh Alignment Fa     C Open Gearing     C Open Gearing     C Open Gearing     C Open Gearing     C Commercial En     C     C Extra Precision     Lubricant Temperature     Temperature up to 1     Temperature greater	NG actor(Cmc) : by crowning or lead actor(Cma) closed Gear Units seed Gear Units Enclosed Gear Units 20 degrees centigrade than 120 degrees centigrade	Pinion Proportion Modifie Straddle mounted pinions (° (S1/S) < 0.175 (° (S1/S) > 0.175 Mesh Alignment Correcti (° Gearing adjusted at a (° Gearing improved by (° All other conditions Help de	<pre>#(Cpm) swidth S, S1 on Factor(Ce) sssembly lapping Ok UK Ot </pre>

Figure 6.5 Input form, straddle mounting

And the other factors are selected as defined in the problem to the AGMA standard recommendations are as follows:

MULTISTEP SPUR	GEARBOX DESIGN		$\mathbf{X}$
File Input Output Hel	p		
Power Requirement and Tooth,Cutter Geometry	Material	Facto	ทร
Application Factor (Ca,K, Surface Condition Factor Reliability Factor (Cr,Kr): Size Factor(Cs,Ks): Temparature Factor (Ct,K Dynamic Factor (Cv,Kv):	a): 1.5 (Cf,Kf): 1 1 1 1 (4): 1 0,7	Mounting Type Traddle M C Overhang Application Ty C Commercia C Critical Ser	e ounting Mounting pe al Application vice
Pritting Stress Geometry f     Radii of Curvature ar     Radii of Curvature ar	-actor I e calculated at LPSTC e calculated both at HPSTI	C and LPSTC	Fending Stress Geometry Factor J     Tip Loading     HPSTC Loading
Lubricant Temperature     Temperature up to 1:     Temperature greater     Temperature greater	20 degrees centigrade than 120 degrees centigrad	le	OK Output
			2

Figure 6.6 Input form, AGMA rating factors

After all the initial values are input to the developed program then overall gear ratio is split to the stages with the desired accuracy and the restrictions. In this problem there are 1222 alternative gear pairs for the required gear ratio. Since all the alternatives provide required precision, all of them are valid values for design. User can select all the gear pairs or the ones that are applicable for the application.

F	our-Stage (	Gear Trai	n																×
						rial Nr	1 Na1	ma1	Np2	No2	ma2	Np3	No3	ma3	Np4	No4	ma4	Overall Bat	io 🔊
	Overall Gea	ar Ratio:		300	2	69 1	4 75	5,357	17	84	4,941	24	85	3,542	25	80	3,2	300	
	Number of a	near trains:	1	222	2	70 14	4 80	5,714	17	75	4,412	24	84	3,5	25	85	3,4	300	-
			1 1		2	71 14	4 80	5,714	17	75	4,412	24	85	3,542	25	84	3,36	300	
	Allow gear	ratio to be i	integer		2	72 14	4 85	6,071	17	75	4,412	24	84	3,5	25	80	3,2	300	
	Allow each	ratio to be	equal	Г	2	73 14	4 75	5,357	17	84	4,941	25	85	3,4	21	70	3,333	300	
				3,36	21	70	3,333	300											
	Solver State	B:		UNE	2	75 14	4 75	5,357	17	84	4,941	25	85	3,4	24	80	3,333	300	_
	Start	Interrru	pt Se	elect All		75 14	4 80	5,/14	17	84	4,941	25	85	3,4	24	/5	3,125	300	_
					Ľ		* 85	6,071		/5	4,412	25	84	3,36	24	180	3,333	300	
	Selected	Zb4/pb4	mp4	Jp4	Jq4	14	Snap4	Snag4	Vtotalo	jear	Vprismati	icgearb	ox Ve	nclosedge	earbox	Fea	sibility	Grade	~
	401	0,651	1,653	0,445	0,384	0,125	0,582	0,817	62908	385	1458	1544		116574	15	Fea	asible	99,01111	
	382	0,646	1,615	0,427	0,368	0,123	0,544	0,815	63658	312	1481	4346		118198	71	Fea	asible	97,85344	
	435	0,656	1,627	0,429	0,372	0,121	0,56	0,814	62806	689	1484	6697		122290	19	Fea	asible	97,41475	
	341	0,657	1,661	0,447	0,387	0,124	0,592	0,814	62860	)78	1531	6876		1299608	33	Fea	asible	95,16987	
	756 0,631 1,612 0,432 0,368 0,125 0,534 0,817 6428124 16347285 12623479 Feasible 93,88164						93,88164												
	396	0,641	1,633	0,44	0,376	0,126	0,56	0,816	63334	400	1641	5121		1325429	93	Fea	asible	92,98546	
	380	0,635	1,603	0,425	0,364	0,124	0,526	0,815	64429	902	1645	6039		1353970	06	Fea	asible	91,96901	
	875	0,635	1,603	0,425	0,364	0,124	0,526	0,816	64590	186	1664	7737		1353069	37	Fea	asible	91,79792	
	267	0,656	1,661	0,448	0,387	0,124	0,592	0,815	62271	36	1728	3894		1361018	08	Fea	asible	91,82185	
	422	0,657	1,653	0,443	0,383	0,123	0,585	0,814	6245/	/98 ))))	1745	0507		127925.	36	Fea	asible	92,85327	
	203	0,003	1,003	0,444	0,307	0,122	0,050	0,013	62230	145	1751	9307	_	1005000	00 10	Fea Fea	asible	91,33437 90 52600	
	397	0,635	1,624	0,435	0,375	0,123	0,545	0,010	63160	158	1756	3982		139362		Fee Fee	asible	90,53000	
	2	0,040	.,		0,010	0,124	0,004	0,010	00100		1100			.0002	~			00,0,004	>
																			<u> </u>
	Solve			Grading														Filterir	ig
	Find	Ascendin	ng Z	a / (Za+Z	2b) mj	0	1	Jp	J	g	Kv,Cv	Gea	rVol.	Prism.Vol	. Enc	.Vol.			
	Save	Descendi	ng	5 🔻	5	• 5	•	5 🗸	5	•	5 💌	20	•	25 💌	25	•	DONE	SELE	СТ
	Solver state																	Ne	xt

Figure 6.7 Input form, alternative gear pairs

In this application all alternatives are selected and solved as follows. The minimum volume value is selected considering the performance variables. Also a grading option exists in the developed program in order to make a decision from the various performance variables easily.

After selection is made with respect to the performance variables and total volume of gearbox then the output is taken shown in Figure 6.8. This output is containing all stages together in a single page, for the designer to analyse the solutions easily. Option is given to the user to consider all stages separately if necessary.

Overall gear ratio = (80/14) \* (70/18) \* (81/21) \* (84/24) = 300. Redesigned gearbox satisfies the overall gear ratio exactly while there is 1.66% error on overall gear ratio in the original problem.

F	our-Stage Output								
9	iraphics								
ſ	STAGE-1 STAGE-2	2	STAGE-3	T T	STAGE-4	GE	AR TRAIN	CONFI	GURATION
F								,L	
1	Stage		1		2		3		1
	Gear	Pinion1	Gear1	Pinion2	- Gear2	Pinion3	- Gear3	Pinion4	Gear4
	Number of teeth	14	80	18	70	21	81	24	84
	Gear ratio	5,7	143	3,8	889	3,8	571	3,	.5
	Module,(mm)	1,	5	1	2	2,	75	3,1	75
	Addendum modification coefficient	0,31	-0,31	0,33	-0,33	0,327	-0,327	0,304	-0,304
	Pitch diameter,(mm)	21	120	36	140	57,75	222,75	90	315
	Outside diameter,(mm)	24,93	122,07	41,32	142,68	65,0485	226,4515	99,78	320,22
	Face width,(mm)	17,8	372	28,2	336	35,0	591	49,1	033
	Contact stress,(MPa)	1245,	6525	1351	,0406	1454	.5427	1567,	.1831
	Permissible contact stress,(MPa)	1250,0242	1382,622	1378,1386	1490,006	1487,0417	1606,976	1603,8141	1723,338
	Bending stress,(MPa)	188,1014	205,3921	265,6063	312,9567	353,2012	414,751	454,0965	526,2317
	Permissible bending stress,(MPa)	386,7045	421,296	421,296	495,33	495,33	581,805	581,805	675,5078
	Bending resistance geometry factor,(J)	0,392	0,359	0,423	0,359	0,438	0,373	0,445	0,384
	Contact resistance geometry factor,(I)	0,12	238	0,12	247	0,1	27	0,12	251
	Addendum portion of the line of action	0,9092	0,6445	0,9666	0,6222	0,9925	0,6298	1,0012	0,6514
	Contact ratio	1,5	0.0045	1,5	888	1,6	223	1,6	526
	Lircular tooth thickness at tooth tip,[1/m]	0,4476	0,8015	0,5053	0,8008	0,5438	0,8028	0,5819	0,8165
l	Shaft diameter,(mm)	8		4	4	.2		5	53
								s	ave to Excel

Figure 6.8 Output form, four-stage gear drive

With the graphics option of the program, performance variables, gear ratio split and module, pitch diameter and face width, dynamic factors, volume of gears with respect to the stages are generated in a graphical interface automatically. They are shown through Figure 6.10 to 6.14.

The contact ratios for all stages are greater than 1.4 as recommended by standards for a smooth operation.

Circular tooth thickness at tooth tips for both pinion and gear are greater than 0.3 module as recommended by AGMA, avoiding excessively thinned at tooth tip automatically. Since smaller numbers of teeth are used for small packaging purposes, default addendum modification coefficient is given to the pinion and the same but opposite value for the gear, in order to prevent undercutting. In smaller number of teeth, the correction required to avoid undercut on gears operated on standard center distances is excessive, in many cases, in respect to equal tooth strength as in this

study. Therefore in case of undercutting the defined equations are used for finding contact ratio and geometry factor J for both pinion and gear.

With respect to AGMA standard 218.01, LPSTC for the pinion is taken as the most critical point for calculation of AGMA geometry factor I, and all of the tables are prepared this rule. It is true for unmodified gears, or gears with small addendum modification coefficients. When there is a large positive pinion addendum modification on a pinion then there is a very long addendum on pinion, and a very short dedendum; and a gear with a very long dedendum and a very short addendum in x-zero gear pairs. These tooth geometries shift the single tooth pair contact region of the pinion above its pitch circle. A similar, but reverse shift is observed on the gear. When the single tooth pair contact regions are shifted in this way, not LPSTC, but HPSTC becomes the most critical point [19]. Therefore the program is given options to the user for calculation of radii of curvature, as defined by AGMA the critical, at LPSTC or both at LPSTC and at HPSTC.

It is clearly seen from the following Figure 6.11 that with increasing torque, from the first stage to the last stage, the module increases. The overall gear ratio is split to stages in a descending order intentionally to compensate for the increasing module (compact volume criteria), because as the module increases from the first stage to the last stage, both face widths and diameters are increasing also seen from Figure 6.13.

Dynamic factors are related to pitch line velocity and AGMA quality number, and in this study since the AGMA quality numbers are selected same for all gears then, the  $K_v$  and  $C_v$  values are depending only to pitch line velocity. The pitch line velocity decreases from the input to the output so the dynamic factors are increasing also seen from the Figure 6.12.

As seen in the Figure 6.14 the total volume of gears is increasing from the first stage to the last stage as expected because of increasing torque.

The safety factors against bending failure are equal because of design intention (optimum design). The safety factors for bending failure are higher than those for surface failure. It is also beneficial because, the bending failure is sudden and catastrophic, resulting in tooth breakage and a disabling of the machine. Also the resulting safety factors are reasonable values (small) for the design intention.

Since the number of stages is assigned by the user, the overall gear ratio is distributed to the stages, and the dimensional design is made to minimize the total volume of gears satisfying the rating practices then all the parameters are ready for the configuration design.



Figure 6.9 Graphics form, performance variables



Figure 6.10 Graphics form, module and gear ratio



Figure 6.11 Graphics form, diameters and face widths



Figure 6.12 Graphics form, dynamic factors



Figure 6.13 Graphics form, volume

The configuration design is made by using nonlinear solver of Lingo 8.0 to find the position of the gear drive elements by constructing an objective function containing the center distance constraints. The objective is selected to minimize the constructed function satisfying all the equality and inequality constraints simultaneously as defined in the previous chapters. The output of the Lingo 8.0 determines the location of the gear drive elements (x, y, z) that minimizes the volume of the gearbox.



Figure 6.14 Configuration form of compact gearbox

The positions of the gear drive elements and all the dimensions are taken automatically from Lingo 8.0 in the previous step then the 3-dimensional drawing of gear train is made by the help of Autodesk Inventor 7.0 automatically. The corresponding result of the configuration design is shown in the Figure 6.15, 6.16 and solid modelling of a four-stage gear drive is seen in Figure 6.17. The gears reveal a tendency to minimize the total volume of the gearbox, while pinions and gears are meshing properly while gears and shafts are avoiding interferences between them.



Figure 6.15 Configuration design of compact gearbox, front and side view



Figure 6.16 Configuration design of compact gearbox, front and isometric view

Prismatic gearbox volume is calculated after the configuration design, simply by multiplying the width, height and the depth of the gear train. Net gearbox volume =  $502.05 \times 342.42 \times 84.72 = 14.6 \times 10^6 \text{ mm}^3$ . The redesigned gearbox reduced its net volume considerably (36%) compared with the original case, and the dimensional design shows reasonable result with using smaller number of teeth. The configuration is different from the original case as expected. The gears also reveal a tendency to minimize the total volume of the gearbox, while pinions and gears are meshing properly while gears and shafts are avoiding interferences between them.



Figure 6.17 Configuration design of compact gearbox, solid model

As mentioned before, the Lingo 8.0 carries out a configuration design only for the gears satisfying the AGMA strength and durability rating practices and the performance variables. Thus the final gear designs are safe enough by considering strength and durability criteria.

The same problem is solved for in-line gearbox design type from the gearbox design type options. Corresponding result of the configuration is shown in the Figure 6.18 and 6.19. The pairs are arranged in line, while pinions and gears are meshing properly while gears and shafts are avoiding interferences between them. In this configuration, gearbox volume is greater than the compact gearbox design as expected.

Four-Stage Output Graphics							×
STAGE-1	STAGE-2	STAGE-3	STA	GE-4	GEAR TR	AIN CONF	GURATION
				Configuratio	on Results		×
				Pinion1:	× 0,25	у 0	z 0,25
				Gear1:	70,75	0	0,25
			~	Pinion2:	70,75	0	23,3543520
	<i>.</i>		$\sim$	Gear2:	158,75	0	23,3543520
	e et	YY Y	•	Pinion3:	158,75	0	55,3681220
				Gear3:	299	0	55,3681220
				Pinion4:	299	0	13,0058671
				Gear4:	501,5	0	13,0058671
657,81 -360,1	Configuration Results	Save	Gearbox				

Figure 6.18 Configuration form of in-line gearbox



Figure 6.19 Configuration design of in-line gearbox, solid model

### **CHAPTER 7**

#### CONCLUSIONS AND RECOMMENDATIONS FOR FUTURE WORK

## 7.1 Conclusions

In this study, a program is developed for the optimum design of multistep involute standard and non-standard spur gear drive. AGMA standards are followed while rating the gear performance. Since the main design objective is the minimization of the total gearbox volume and to obtain equal strength of pinion and gear tooth for all stages, then instead of standard tooth form, a long addendum (short dedendum) form is used for pinions, and a short addendum (long dedendum) is used for gears. It is seen that by using non-standard gears, some important benefits are obtained. The most important ones are, having gears with smaller number of teeth without undercutting, reducing the total volume of gears and obtaining balanced strength between the pinion and the gear for all stages.

In the case studies, it is shown that all alternative gear pairs satisfying the required overall gear ratio with the desired accuracy is analyzed to select the minimum volume considering the performance criteria with the user interaction. Using smaller number of teeth in a normal design necessitate modification for gears only to prevent undercutting. In this study among minimum volume objective, also it is desired to equalize the factor of safeties for bending failure so sometimes the modification coefficients are not enough to prevent undercutting condition. Some of the gears are undercut and thus reduce the contact ratio. Thus a compromise must be maintained between conflicting objectives. By considering the performance criteria of all the

alternatives by the help of the developed program, sufficient ones are selected for final configuration design.

The configuration design was considered as a problem of packing gears in threedimensional space. It is problematic to conventional gradient-based optimization methods due to discontinuities and severe nonlinearities in its objective function and constraints. In this study in order to overcome these difficulties the configuration design is carried out using Lingo 8.0 optimization software with the necessary data (diameters, face widths) taken from the developed program by the use of DLL (Dynamic Link Library) interface automatically which was fixed in the previous design steps. It is made to minimize the volume of gearbox, while satisfying spatial constraints, such as gear meshing and interference. It is clearly seen from the application problem results that the minimum volume gear pairs are positioned in the gearbox in a compact position while satisfying all spatial nonlinear constraints. The major benefits of this study can be defined as follows;

- The ratio splitting algorithm, construct the overall gear ratio exactly with the desired precision from ±0.1 to ±0.00001. This algorithm is very useful not only for the single module for the design of multistep gearbox up to six stages, but also as an independent program to split overall gear ratio for any practical design problem.
- The design results are reasonable values considering the performance variables. The gear pairs are reliable for strength and durability.
- The objective function formulation and the constraints for configuration design give satisfactory results.
- The total design process will be used for design of multistep spur gear drives according to the AGMA standards, including detailed design parameters effectively.

# 7.2 Recommendations for Future Work

- The program is developed for multistep spur gearbox. Additional modules can be developed for other types of gears including helical, bevel, worm gearing and other types.
- Different objective functions can be used for other design intentions.
- More effective optimization software can be used for better configuration design.
- Lubrication of gears and gear dynamics can be included in the new studies.

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# **APPENDIX** A

## **USER'S GUIDE**

The developed program has the following sub menus:

- File
- Input
- Output
- Help

Under the File menu command, options can be selected to constraint the design limitations such as face width limits, number of teeth for pinion and gear, backlash, tooth type, factor of safety. When "exit" is selected the program is closed. The program is developed for optimum design of multistep spur gearbox. The design intention is selected as minimization of total gear volume and total gearbox volume. Gear type is standard and non-standard spur gear according to the input values, in order to satisfy the required rating practices, performance variables and gearbox volume minimization. The unit system used in the program is metric. After selecting these items, the main input form will be opened.

The main input form has three tab dialog boxes. The items of the tabs are as follows

- Power Requirement and Tooth, Cutter geometry
- Material
- Factors

Under the Power Requirement and Tooth, Cutter geometry tab, the following items are presented:

- *Power:* The user is expected to enter the transmitted power in kW.
- *Input Speed:* The user is expected to enter input speed in rpm.
- Overall Gear Ratio: The user is expected to enter overall gear ratio.
- *Required Precision:* The user is expected to enter required precision on overall gear ratio in a range of  $\pm 0.1$  to  $\pm 0.00001$ .
- *Number of Stages:* The user is expected to enter the number of stages considering the overall gear ratio from two to six stages which are common for conventional external spur gear drives.
- *AGMA Quality Number (Qv):* The user is expected to select AGMA quality number from the prepared combo box.
- *Reliability:* The user is expected to enter required reliability.
- *Number of Load Cycles:* The user is expected to enter load cycles.
- *Rack Cutter Pressure Angle:* The user is expected to select the pressure angle from the standard angles (14.5°, 20°, 25°)
- *Tool Edge Radius:* It is automatically shown on the textbox after selection of the rack cutter pressure angle. The user can change the tool edge radius if necessary.
- *Standard Modules:* Both preferred and second choice modules are input to the program and the user is asked to select the desired option.
- *Gearbox Design Type:* Compact gearbox and In-line gearbox design options are given to the user.

Under the Material tab, following items are presented.

Type of Material: It can be selected from material types given in the AGMA standard. The following materials are found in the standard:

• Steel (Through Hardened)

- Steel (Flame or Induction Hardened)
- Steel ( Carburized and Case Hardened )
- Steel (Nitrided)
- Cast Iron
- Nodular ( Ductile) Iron
- Malleable Iron
- Bronze

Once a material type with its hardness and grade is selected, then program automatically finds its allowable contact stress number, allowable bending stress number, Young Modulus of Elasticity (MPa), and Poisson's ratio. Upon request, they can be modified.

Under factors tab, the following items are listed:

- Application Factor (C<sub>a</sub>, K<sub>a</sub>)
- Surface Factor (C<sub>f</sub>, K<sub>f</sub>)
- Reliability Factor (Cr, Kr)
- Size Factor  $(C_s, K_s)$
- Temperature Factor  $(C_t, K_t)$
- Dynamic Factor ( C<sub>v</sub>, K<sub>v</sub>)
- Life Adjustment (Commercial or Critical Service)
- Pinion Mounting Type (Straddle or Overhung)

If the straddle mounting is selected another sub form is opened for the detailed information about straddle mounting which includes:

- Lead Correction Factor (C<sub>mc</sub>)
- Mesh Alignment Factor (C<sub>ma</sub>)
- Pinion Proportion Modifier (C<sub>pm</sub>)
- Mesh Alignment Correction Factor (C<sub>e</sub>)

If the Overhang mounting is selected than user is required to enter:

- Operating Condition (Contact across the full face)
- Tooth Stiffness Constant (MPa)
- Total lead mismatch between mating teeth (mm)

The designer can select any form by using the tab buttons which are at the top of the forms.

After finishing the data input part, "OK" command button is clicked to start the main algorithm. Then a form comes into the screen including all the gear alternatives that satisfies the overall gear ratio depending on the constraints that are specified by the user.

User can select all the alternatives in order to try all possible solutions or the ones that are sufficient according to the designer. Then selected gear pairs are input to the main algorithm and run to reach to the solutions.

The solutions are analyzed by the user considering the gearbox volume, and performance variables. A grading frame is existing in the program to give importance to the desired performance variables and the other criterion. Also some restrictions are given an option to filter the gear pairs. The form contains:

- Alternative Gear Pairs
- Number of Teeth of Pinions and Gears
- All Performance Variables
- Sorting Option for Any Desired Criteria
- Options for Splitting Overall Gear Ratio
- Filtering Options

After execution of the developed program, the output menu will be opened. The program developed gives the results in a similar form as in the input part using tabs. The results are presented in six different sub tabs for all stages depending on the number of stages and a graphics menu. These tabs are:

# 1. Gear Drive Properties-1:

- Number of teeth
- Normal Pressure Angle
- Pitch Diameter
- Operating Pitch Diameter
- Base Diameter
- Outside Diameter
- Root Diameter
- Face Width
- Circular Tooth Thickness at Tooth Tip
- Addendum Modification Coefficient
- 2. Gear Drive Properties-2:
  - Module
  - Operating Pressure Angle
  - Total Addendum Modification Coefficient
  - Standard Center Distance
  - Operating Center Distance
  - Pinion Addendum Length of Action
  - Pinion Dedendum Length of Action
  - Total Length of Action
  - Bottom Clearance
  - Contact Ratio
  - Total Gear Volume
- 3. Factors :
  - Application Factor (C<sub>a</sub>, K<sub>a</sub>)
  - Surface Factor (C<sub>f</sub>, K<sub>f</sub>)
  - Reliability Factor ( $C_r$ ,  $K_r$ )
  - Size Factor  $(C_s, K_s)$

- Temperature Factor  $(C_t, K_t)$
- Dynamic Factor (C<sub>v</sub>, K<sub>v</sub>)
- Load Distribution Factor (C<sub>m</sub>, K<sub>m</sub>)
- Life Factor for Pitting Resistance (C<sub>L</sub>)
- Life Factor for Bending Resistance (K<sub>L</sub>)
- Hardness Ratio Factor (C<sub>H</sub>)
- Geometry Factor for Pitting Resistance (I)
- Geometry Factor for Bending Resistance (J)
- 4. Strength and Durability Rating:
  - Bending Stress
  - Permissible Bending Stress
  - Contact Stress
  - Permissible Contact Stress
  - Tangential Load
  - Torque
  - Pitch Line Velocity
  - Load Cycles
- 5. Pinion Tooth Profile
- 6. Gear Tooth Profile

And the graphics pull down menu is including the following items:

- Performance Variables
- Gear Ratio Split and Module
- Pitch Diameter and Module
- Dynamic Factors
- Volume

A gear train tab exists in the program to analyze the important parameters for all stages in the gear drive together in a single form, which contains:

- Number of teeth
- Gear Ratio
- Module
- Addendum Modification Coefficient
- Pitch Diameter
- Outside Diameter
- Face Width
- Contact Stress
- Permissible Contact Stress
- Bending Stress
- Permissible Bending Stress
- Bending Resistance Geometry Factor
- Contact Resistance Geometry Factor
- Addendum Portion of the Line of Action
- Contact Ratio
- Circular Tooth Thickness at Tooth Tip
- Shaft Diameter

Also a configuration tab exists in the program to view the allocation of the gears schematically while they are meshing.

If the results are satisfactory than the user can save the design data to an excel file, if not may go to back and continue with the program in order to reach a more reasonable solution. The program specifies the positions of pinions and gears automatically by using DLL of Lingo 8.0 optimization software to minimize the volume of designed gear train while satisfying the spatial constraints. The three dimensional output is made automatically using Autodesk Inventor 7.0 with the results obtained from the developed program and the Lingo 8.0 optimization software. The following section includes the screen shots of the developed program in the visual basic form.



Figure A. 1 Start form

MULTISTEP SPUR GE	ARBOX DESIGN		
File Input Output Help			
Power Requirement and Tooth,Cutter Geometry	Material	Factors	
Power Transmitted Power(kW):	1		
Input Speed (rpm):	2500		
Overall Gear Ratio:	222,222		
Required Precision( ± ):	0.001 💌		
Number of Stages:	five-stage 👻		
Power Source:	Light Shock(Multicyli	nder engine) 📃 🚽	
Driven Machine:	Moderate shock(Con	crete mixers,textile machiner;	y) 🔻
AGMA Quality Number(Qv):	11 💌	Single Pair Reliability:	0,99
Number of Load Cycles:	1000000	Overall Reliability:	0,951
Cutter Geometry		Standard Modules	
Rack Cutter or Hob		C Preferred	
Rack Cutter Pressure Angle:	20 💌	Preferrred and Second	Choice
Tool Edge Radius (* m):	0,25	Gearbox Design Type Compact Gearbox 💌 🔤	Options OK
S	•		

Figure A. 2 Input form, power requirement, tooth, and cutter geometry

File Input Output Help         Power Requirement and Tooth,Cutter Geometry         Material       Factors         Stage-4       Stage-5       Stage-6         Same for all stages       Stage-1       Stage-2       Stage-3         Gear Material:       Image: Stage-1       Stage-2       Stage-3         Heat Treatment:       Carburised and Case-hardened         Min.Hardness at suface:       55 HRC         Grade:       1         AGMA bending strength(MPa):       380         AGMA surface fatigue strength(MPa):       1250         Young Modulus of Elasticity,Sat(MPa):       200000         Poisson's Ratio:       0.3	🖻 MULT	ISTEP SPUR G	EARBOX DESIG	N				×
Power Requirement and Tooth,Cutter Geometry       Material       Factors         Stage-4       Stage-5       Stage-6         Same for all stages       Stage-1       Stage-2         Gear Material:       Steel       Image: Carburised and Case-hardened         Heat Treatment:       Carburised and Case-hardened         Min.Hardness at suface:       55 HRC         Grade:       1         AGMA bending strength(MPa):       380         AGMA surface fatigue strength(MPa):       1250         Young Modulus of Elasticity,Sat(MPa):       200000         Poisson's Ratio:       0.3	File Input	: Output Help						
Stage-4       Stage-5       Stage-6         Same for all stages       Stage-1       Stage-2       Stage-3         Gear Material:       Image: Stage and Case-hardened       Image: Image: Stage and Case-hardened         Heat Treatment:       Carburised and Case-hardened       Image: Image: Image: Stage and Case-hardened         Min.Hardness at suface:       55 HRC       Image:	Power Red Tooth,Cut	quirement and ter Geometry	Material		Factors			
Same for all stages       Stage-1       Stage-2       Stage-3         Gear Material:       Image:		Stage-4	Stage-5		Stage-6	)		
Gear Material:       Steel         Heat Treatment:       Carburised and Case-hardened         Min.Hardness at suface:       55 HRC         Grade:       1         AGMA bending strength(MPa):       380         AGMA surface fatigue strength(MPa):       1250         Young Modulus of Elasticity.Sat(MPa):       200000         Poisson's Ratio:       0.3	Same	for all stages	Stage-1	ľ	Stage-2		Stage-3	
Min.Hardness at suface:       55 HRC         Grade:       1         AGMA bending strength(MPa):       380         AGMA surface fatigue strength(MPa):       1250         Young Modulus of Elasticity,Sat(MPa):       200000         Poisson's Ratio:       0.3	Gea	ar Material: at Treatment:		Stee	I urised and Case-hardened	•		
Grade:     1       AGMA bending strength(MPa):     380       AGMA surface fatigue strength(MPa):     1250       Young Modulus of Elasticity,Sat(MPa):     200000       Poisson's Ratio:     0.3	Min.	Hardness at sufac	ce:	55 H	RC			
AGMA bending strength(MPa):       380         AGMA surface fatigue strength(MPa):       1250         Young Modulus of Elasticity,Sat(MPa):       200000         Poisson's Ratio:       0.3	Gra	de:		1				
AGMA surface fatigue strength(MPa): 1250 Young Modulus of Elasticity,Sat(MPa): 200000 Poisson's Ratio: 0.3	AGN	MA bending streng	th(MPa):	380		-		
Young Modulus of Elasticity, Sat(MPa):     200000       Poisson's Ratio:     0.3	AGN	MA surface fatigue	strength(MPa):	1250	1			
Poisson's Ratio: 0.3	You	ing Modulus of Ela	isticity,Sat(MPa):	2000	00	-		
	Pois	sson's Ratio:		0.3				
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Figure A. 3 Input form, same material selection for all stages

MULTISTEP SPUR GE	ARBOX DESIGN				×
File Input Output Help					
Power Requirement and Tooth,Cutter Geometry	Material	F	actors		
Stage-4	Stage-5	Y s	itage-6	ך ר	
Same for all stages	Stage-1	Sta	ige-2	Stage-3	Π
Material	PINION-1		E	GEAR-1	
Material:	Steel	-	Steel		
Heat Treatment:	Through-hardened and	tempered	Through-ha	rdened and tempered	
Min.Hardness at suface:	400 BHN		360 BHN		
Grade:	1		1		
AGMA bending strength(MPa):	290		280		
AGMA surface fatigue strength(MPa):	1100		1000		
Young Modulus of Elasticity,Sat(MPa):	200000		200000		
Poisson's Ratio:	0.3		0.3		μ
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Figure A. 4 Input form, different material selection for all stages

STEEL GE	AR SPECIFICAT	IONS						×
					AGMA Bending S	trength,psi(MPa)	AGMA Surface Fatig	ue Strength,psi(MPa)
Material	AGMA Class	Commercial Designation	Heat Treatment	Min.Hardness at Surface	Grade1	Grade2	Grade1	Grade2
Steel	A-1 through A-5		Through-hardened & tempered	180 BHN	25000 (170)	33000 (230)	85000 (590)	95000 (660)
Steel			Through-hardened & tempered	240 BHN	31000 (210)	41000 (280)	105000 (720)	115000 (790)
Steel			Through-hardened & tempered	300 BHN	36000 (250)	47000 (320)	12000 (830)	135000 (930)
Steel			Through-hardened & tempered	360 BHN	40000 (280)	52000 (360)	145000 (1000)	160000 (1100)
Steel			Through-hardened & tempered	400 BHN	42000 (290)	56000 (390)	155000 (1100)	170000 (1200)
Steel			Flame or Induction-hardened	50 HRC	45000 (310)	55000 (380)	170000 (1200)	190000 (1300)
Steel			Flame or Induction- hardened	54 HRC	22000 (150)	22000 (150)	175000 (1200)	195000 (1300)
Steel			Carburised and Case-hardened	55 HRC	55000 (380)	65000 (450)	180000 (1250)	200000 (1400)
Steel			Carburised and Case-hardened	60 HRC	55000 (380)	70000 (480)	200000 (1400)	225000 (1550)
Steel		AISI 4140	Nitrided	48 HRC	34000 (230)	45000 (310)	155000 (1100)	180000 (1250)
Steel		AISI 4340	Nitrided	46 HRC	36000 (250)	47000 (325)	150000 (1050)	1750000 (1200)
Steel		Nitralloy 135M	Nitrided	60 HRC	38000 (260)	48000 (330)	170000 (1170)	195000 (1350)
Steel		2.5% Chrome	Nitrided	54 HRC	55000 (380)	65000 (450)	155000 (1100)	172000 (1200)
Steel		2.5% Chrome	Nitrided	60 HRC	55000 (380)	65000 (450)	192000 (1300)	216000 (1500)
- Materia								
Heat tre	satment: Throug	gh-hardened and tempered	п					
Hardne	ss: 400 Bł	NH						
Grade:	-						Cance	×

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e Strength, psi(M	Grade2	60000 (410)	75000 (520)	85000 (590)	¥
AGMA Surface Fatigu	Grade1	50000 (340)	65000 (450)	75000 (520)	Cancel
trength,psi(MPa)	Grade2	5000 (35)	8500 (69)	13000 (90)	
AGMA Bending S	Grade1	5000 (35)	8500 (69)	13000 (90)	
	Min.Hardness at Surface		175 BHN	200 BHN	
	Heat Treatment	As cast	As cast	As cast	
	Designation				
	Commercial				vs Cast 00 BHN
	AGMA Class Commercial	20	8	40	al reatment: As Cast ess: 200 BHN 1

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	AGMA Bending Strength,psi(MPa) AGMA Surface Fatigue Strength,psi(MPa)	Surface Grade1 Grade2 Grade1 Grade2	10000 (69) 72000 (495)	13000 (30) 78000 (540)	16000 (110) 83000 (570)	21000 (145) 94000 (650)	Cancel
	A Bending Strength, psi(MPa) 📔 AGMA Surfa	årade1 Grade2 Grade	000 (69) 72000 (4	000 (30) 28000 (2	00 (110) 83000 (5	000 (145) 94000 (6	
	AGM	Min.Hardness at Surface 0	165 BHN 10	13 180 BHN 13	195 BHN 160	240 BHN 210	
		Heat Treatment					
CINTLA LINIS		Commercial Designation	45007	50005	53007	80002	t treatment N
		AGMA Class	A-8-c	A-8-e	A-8-f	A-8-i	nt: 195 BH
HLLEADLE IN		Material	Malleable Iron	Malleable Iron	Malleable Iron	Malleable Iron	Material Heat treatme Hardness: Grade:

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Heat Treatm Sand Cast Heat Treatm Heat Treatm	oy 9C 620 M		AGMA Bending Strength, psi(MPa) AGMA Surface Fatigue Strength, psi(MPa)	nation   Heat Treatment   Min.Hardness at Surface   Grade1   Grade2   Grade1   Grade2	Sand Cast Min. Tensile Strength 40000psi(275MPa) 5700 (40) 30000 (205)	oy 9C Heat Treated Min. Tensile Strength 90000psi(620MPa) 23600 (160) 65000 (450)			620 Mpa	Cancel
FICATIONS Commercial Designation AGMA 2C ASTM B-148-52 alloy 9C Heat Treated Hin. Tensile Strength 620 M		GEAR SPECI		AGMA Class	Bronze 2	AI/Br 3	stial	treatment:	ness:	ö
GEAR SPECIFICATIONS       AGMA Class     Commercial Designation       AGMA Class     Commercial Designation       Bronze 2     AGMA 2C       Al/Br 3     ASTM B-148-52 alloy 9C       arial     Astronated       strial     Astronated       it reatment:     Heat Treated       hess:     Min. Tensile Strength 620 M	GEAR SPECT AGMA Class Bronze 2 Al/Br 3 Al/Br 3 arial treatment: [ Iness: [			<u>e</u>	es.	Se	date	ea	lard	jrad

Figure A. 9 Input form, bronze gear specifications

MULTISTEP SPUR	GEARBOX DESIGN		×
File Input Output Hel	p		
Power Requirement and Tooth,Cutter Geometry	Material	Factors	
Application Factor (Ca,Ka Surface Condition Factor Reliability Factor (Cr,Kr): Size Factor(Cs,Ks): Temparature Factor (Ct,K Dynamic Factor (Cv,Kv):	a): [1,5 [Cf,Kf]: [1 [1 [1] [1] [1] [0,7]	Mounting Type Straddle Mounting Overhang Mounting Application Type Commercial Application Citical Service	
Pitting Stress Geometry F Radii of Curvature an Radii of Curvature and	Factor I e calculated at LPSTC e calculated both at HPSTI	Bending Stress Geometry Factor J C Tip Loading IC and LPSTC Cloading HELP	J
Lubricant Temperature Temperature up to 12 Temperature greater	20 degrees centigrade than 120 degrees centigrac	ide OK Output	

Figure A. 10 Input form, factors

	GEARBOX DESIGN				×				
File Input Output Hel	0								
Power Requirement and Tooth,Cutter Geometry	Material		Factors						
		- Mountin	д Туре						
STRADDLE MOUNTING	i				×				
Lead Correction Fact	or(Cmc)	Pinion Pro	portion Modifier(C	(pm)					
Unmodified Leads		Straddle n	nounted pinions w	idth					
- Modified Leads b	crowning or lead	@ (\$1/\$	12.0.175						
correction		- (c1/c)	N 0.17E	S S1					
- Mesh Alignment Fact	or(Cma)	0 (51/5	> 0.175						
C Open Gearing		- Mesh Alig	nment Correction	Factor(Ce)					
Commercial Enclo	sed Gear Units	🖲 Gearin	ig adjusted at ass	embly	or J –				
C Precision Enclose	d Gear Units	Gearing improved by lapping							
C Extra Precision Er	nclosed Gear Units	C All oth	er conditions						
			Help	Ok					
Lubricant Temperature					- 1				
Temperature up to 12	<ul> <li>Temperature up to 120 degrees centigrade</li> </ul>								
C Temperature greater than 120 degrees centigrade									
			_		utput				
55									

Figure A. 11 Input form, straddle mounting
File Input Output Help	
Power Requirement and Material Factors	
Evaluation of 5 and 51	
Applica	
Surface Centerline of Sear Face	
Reliabi	
Size Fa Centerline Centerline	
Tempa or Bearing	
Pitting y Fac	tor J –
© Ra	
C Ra	
s i	
S1 = the offset of the pinion, i.e., the distance from the bearing span centerline to the	
Lubrica pinion midface.(mm)	
Tet 5 = the the bearing span, i.e., the distance between the bearing centerlines,(mm)	
C Temperature greater than 120 degrees centigrade OK Outp	out



	GEARBOX DESIGN		$\mathbf{X}$
File Input Output He	lp		
Power Requirement and Tooth,Cutter Geometry	Material	Factors	
Application Factor (Ca,K Surface Condition Factor Overhung M	a): 1,5 Journal Designment of the second second second second second second second second second second second second s	Mounting Type	X
Reliability     Operating (       Size Fac     Contact A       Tempara     Contact A       Dynamic     Tooth Stiffr       Pitting St     Total Lead       Image: Total Lead     Total Lead	Condition .cross the Entire Face Width ss than the Entire Face Widt ness Constant.G. (Mpa) Mismatch between Mating T	th C reeth,et.(mm)	ess Geometry Factor J –
C Radi		HELP	Loading
Lubricant Temperature     Temperature up to 1     Temperature greater	20 degrees centigrade than 120 degrees centigrad	e	OK Output
×			

Figure A. 13 Input form, overhung mounting

Overal Gear Ratio:         222.222           Number of gear train::         116         Np1         Ng2         Ng2         Ng3         Ng3         Ng4         Ng4         Ng4         Ng5         Ng5         Overall Ratio           Allow gear ratio to be integer         116         50         3.333         15         48         3.2         16         50         2.778         18         44         2.667         222.2222         222.2222           Allow gear ratio to be integer         4         15         51         3.4         15         48         3.2         17         50         2.941         18         50         2.778         18         45         2.5         222.2222           Allow each ratio to be equal         5         15         13         4.1         15         50         3.333         17         46         2.667         18         45         2.5         222.22222           Solver rate:         DONE         5         13         3.4         15         50         3.333         17         40         2.667         18         45         2.5         222.22222         222         222         222         222         222         222         222 <td< th=""><th>F</th><th>ive-Stage (</th><th>Gear Traii</th><th>n</th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th></th><th>×</th></td<>	F	ive-Stage (	Gear Traii	n																			×
Overal Gaer Ratio:       222.222       Initial wpi mgi mgi mgi mgi mgi mgi mgi mgi mgi mg							1 11-1	In a	- 1	lu al	N-2		<b>N</b> 2	11-2		11-4			DI-E	DI-E	E STATE	0	
1       1       15       50       3,33       15       48       3.2       16       50       2,178       18       48       2,222222       2         Number of gear trains:       116       15       50       3,33       15       48       3.2       16       50       2,155       18       48       2,2667       18       45       2,5       222,22222         Allow gear ratio to be integer       Allow gear ratio to be equal       50       3,33       15       48       3.2       17       50       2,941       18       50       2,778       18       45       2.5       222,22222       22       222       222       222       22		Overal Gea	r Batio	000	2.000	1 1 10	а порт	INGI	mgi	NP2	ING2	mg2	ирз	Ng3	mg3	19p4	ING4	mg4	сди	CDN	mgo	Uverall Hatio	김 🎒
Number of gear trains:       116         Allow gear ratio to be integer       3       15       9       3,15       18       32       17       50       2,941       18       15       222,22222         Allow gear ratio to be integer       4       15       50       3,33       15       48       3,21       17       50       2,941       18       45       2,55       222,22222         5       15       51       3,4       15       50       3,333       17       50       2,941       18       45       2,5       222,22222         5       15       51       3,4       15       50       3,333       17       48       2,667       18       45       2,5       222,22222         7       15       50       3,333       15       48       3,233       18       50       2,778       18       45       2,5       222,22222         9       15       54       3,6       15       50       3,333       18       50       2,778       18       48       2,667       18       45       2,5       222,22222       2       2         5       15       51       3,41       15.0		010101000	a ritado.	220	2,222		15	50	3,333	15	48	3,2	16	40	2,812	18	50	2,118	18	48	2,667	222,22222	
Allow gear ratio to be integer       Image: Solver state:       Image: Solver state:       DONE       Since Solver state:       Since Solver state:       DONE       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since Solver state:       Since		Number of	gear trains:	1	16	2	15	50	3,333	15	48	3,2	16	50	3,125	18	48	2,667	18	45	2,5	222,22222	4
Allow each ratio to be equal       5       15       51       3.4       15       50       3.33       17       48       50       2.778       18       45       2.5       22.22222         Solver state:       DONE       5       15       51       3.4       15       50       3.33       17       48       2.667       18       45       2.5       22.22222         Solver state:       DONE       5       15       51       3.4       15       50       3.333       18       50       2.778       18       45       2.5       22.22222       <		Allow goor	ratio to be i	integer	E		15	50	3,333	15	48	3,2	17	50	2,941	18	51	2,833	18	45	2,5	222,22222	
Allow each ratio to be equal       b       15       51       3.4       15       90       3.33       17       48       2.624       18       50       2.7.78       18       45       2.5       222.22222         Solver state:       DDNE       8       15       51       3.4       15       50       3.333       17       50       2.778       17       45       2.667       17       45       2.647       222.22222       22       22       222.2222       22       22       222.2222       22       22       222.2222       22       22       222.22222       2       3       15       50       3.333       15       50       2.778       18       48       2.667       17       45       2.647       222.22222       22       22       222.22222       2       3       15       50       3.333       18       50       2.778       18       48       2.667       18       45       2.5       2.22.22222       2       22		Allow gear	Tallo to be	integer	1	4	15	51	3,4	15	48	3,2	17	50	2,941	18	50	2,778	18	45	2,5	222,22222	
Solver state:         DUNE         b         15         51         3.4         15         50         3.33         17         50         2.941         18         48         2.667         18         45         2.5         2.22222           Stat         Interrupt         Select AI         Select AI         Select AI         15         50         3.333         15         48         3.633         15         50         2.778         18         48         2.667         17         45         2.647         222.22222         3         15         54         3.6         15         50         3.333         18         50         2.778         18         48         2.667         17         45         2.647         222.22222         3         15         54         3.6         15         50         3.333         18         50         2.778         18         48         2.667         17         45         2.647         222.222222         3         16         3.333         18         50         2.778         18         48         2.667         17         45         2.647         222.222222         3         3         333         18         2.55         2.778         18		Allow each	n ratio to be	equal	Г	5	15	51	3,4	15	50	3,333	17	48	2,824	18	50	2,778	18	45	2,5	222,22222	4
Vorter Nature         DURCe         V         15         50         3.33         15         48         3.2         18         50         2.78         17         47         45         2.647         2222222           Statt         Interrupt         Select All         8         15         51         3.4         15         50         3.333         18         50         2.778         18         48         2.667         18         45         2.5         222.222222         9         15         54         3.6         15         50         3.333         18         50         2.778         18         48         2.667         18         45         2.5         222.222222         9         15         54         3.6         15         0.333         18         50         2.778         18         48         2.667         18         45         2.5         222.2222222         9         10         0.81         10.83         0.393         0.314         0.68         15.94         0.37         0.352         0.111         0.55         0.783         14/2/38         3609393         300014         6.83         17         Feasible         98.40174         98.40174         98.40174         <		Solver state			ONE		15	51	3,4	15	10	3,333	17	50	2,941	18	48	2,667	18	45	2,5	222,22222	-
Stat         Interrupt         Select All         B         15         51         3.4         15         50         3.333         18         50         2.778         18         48         2.667         17         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.647         12         45         2.52         22222222         ×           Selected         Kv6.Cv5         Za5/pb5         Zb5/pb5         µp32         µp11         µp33         µp33         µp33         µp33         µp33         µp33         µp33         µp33         µp33         µp33         µp34         µp34         µp34         µp34         µp34         µp34         µp34         µp34         µp34         µp34 <td< td=""><td></td><td>Solver statt</td><td></td><td></td><td>UNE</td><td></td><td>15</td><td>50</td><td>3,333</td><td>15</td><td>48</td><td>3,2</td><td>18</td><td>51</td><td>2,833</td><td>18</td><td>50</td><td>2,778</td><td>17</td><td>45</td><td>2,647</td><td>222,22222</td><td></td></td<>		Solver statt			UNE		15	50	3,333	15	48	3,2	18	51	2,833	18	50	2,778	17	45	2,647	222,22222	
9       15       54       3.6       15       50       3.3.33       18       50       2.7.78       18       48       2.66       18       45       2.5       22.22222 v         Selected       Kv5.v5       Zz5.pb5       Zb5.pb5		Start	Interru	upt Sei	lect All	8	15	51	3,4	15	50	3,333	18	50	2,118	18	48	2,667	17	45	2,647	222,22222	
Selected         Kv5.Cv5         Za5/pb5         Zb5/pb5         mp5         Jp5         Jg5         I5         Snap5         Viotalgear         Vprismaticgeabox         Venclosedgeabox         Feasible         98/2016           18         0.99         0.914         0.68         1.984         0.937         0.352         0.111         0.55         0.783         14/24738         3667371         2986933         Feasible         99/2016           77         0.99         0.914         0.68         1.594         0.337         0.352         0.111         0.55         0.783         14/24738         3667371         2986933         Feasible         99.77802           91         0.93         0.914         0.68         1.594         0.337         0.352         0.111         0.55         0.783         14/2366         3736593         3160444         Feasible         99.77802           91         0.93         0.914         0.68         1.594         0.337         0.352         0.111         0.55         0.783         14/26372         366603         3098066         Feasible         99.77546           1         0.99         0.914         0.68         1.593         0.342         0.110         0.55							15	54	3,6	15	50	3,333	18	50	2,118	18	48	2,667	18	45	2,5	222,2222	~
18       0.93       0.914       0.63       1.534       0.607       0.332       0.111       0.55       0.783       1416.557       2.50533       2.00520       Feasible       5020031         77       0.98       0.914       0.68       1.594       0.397       0.382       0.111       0.55       0.783       1416.557       2.505333       2.00520       Feasible       504.1223         15       0.99       0.914       0.68       1.594       0.397       0.382       0.111       0.55       0.783       1422010       378579       3149107       Feasible       59.48174         44       0.39       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1422010       3786716       3149107       Feasible       59.48174         44       0.39       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       142516       3863107       2986317       Feasible       59.1778       1432433       3078922       3778798       Feasible       59.17778       1432433       3078922       3778798       Feasible       59.10977       1432433       3078922       3778798       Feasible       59.1097		Selected	Kv5 Cv5	Za5/pb5	Zh5/nh5	mp5	Jo5	Jo5	15	Snan5	Snac	15 VI	otalne	ar N	/nrismatic	nearhr	nx Ve	enclosedo	1earbo	× F	easihility	Grade	
77       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1424739       3687371       2986893       Feasible       99.41223         15       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1424739       3687371       2986893       Feasible       99.41223         91       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1422616       3963107       2986317       Feasible       99.477602         1       0.93       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       142272       3868603       3098806       Feasible       99.47768         1       0.93       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1426372       3868603       3098806       Feasible       99.17548         87       0.93       0.904       0.683       1.593       0.322       0.510       108       0.557       0.779       1432439       382537       3262300       Infeesible       97.95914		18	0,99	0,914	0,68	1,594	0,397	0,352	0,111	0,55	0,78	3 1	41983	5	3609	398		30203	40	F	easible	98,2066	
15       0.99       0.914       0.68       1.594       0.397       0.382       0.111       0.55       0.782       1422366       3736599       3160444       Feasible       97.7802         91       0.99       0.914       0.68       1.594       0.397       0.382       0.111       0.55       0.783       1422306       3736599       3160444       Feasible       97.7802         91       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       14223010       3786716       3149107       Feasible       98.46174         44       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       142872       3968100       2996806       Feasible       98.16788         87       0.39       0.904       0.689       1.593       0.322       0.351       0.108       0.557       0.779       1433439       3967862       3179786       Feasible       98.2168         105       0.99       0.904       0.689       1.593       0.342       0.108       0.557       0.779       1432107       3985782       2987826       Feasible       99.2168		77	0,99	0,914	0,68	1,594	0.397	0,352	0,111	0,55	0,78	3 1	42473	8	3687	371		29868	93	F	easible	98,41223	
91       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       14.23010       3789716       31.49107       Feasible       98.49174         44       0.98       0.314       0.68       1.594       0.397       0.352       0.111       0.55       0.783       14.22616       3863107       2986803       79.682317       Feasible       99.46788         1       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1422616       3863107       2986803       3098906       Feasible       99.1674       16.81       16.057       0.779       143349       3078922       3178798       Feasible       99.10007         58       0.99       0.904       0.689       1.583       0.320       0.351       0.108       0.557       0.779       1432703       3898943       3262300       Infeasible       99.2169         105       0.99       0.904       0.689       1.593       0.320       0.351       0.108       0.557       0.779       1432703       389543       3262300       Infeasible       98.2713         105       0.99       0.904       0.689 <t< td=""><td></td><td>15</td><td>0,99</td><td>0.914</td><td>0,68</td><td>1.594</td><td>0.397</td><td>0.352</td><td>0,111</td><td>0,55</td><td>0.78</td><td>3 1</td><td>42326</td><td>6</td><td>3736</td><td>599</td><td></td><td>31604</td><td>44</td><td>F</td><td>easible</td><td>97,77802</td><td></td></t<>		15	0,99	0.914	0,68	1.594	0.397	0.352	0,111	0,55	0.78	3 1	42326	6	3736	599		31604	44	F	easible	97,77802	
44       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1422516       3963107       2996317       Feasible       99.45788         1       0.99       0.914       0.68       1.594       0.397       0.352       0.111       0.55       0.783       1422516       3963107       2996317       Feasible       99.45788         87       0.99       0.904       0.689       1.593       0.322       0.517       14324372       3966603       3098806       Feasible       99.17548         58       0.99       0.907       0.674       1.593       0.321       0.108       0.557       0.779       1432473       3989578       2987826       Feasible       99.29914         33       0.39       0.904       0.689       1.593       0.322       0.351       0.108       0.557       0.779       1432470       3896578       2987826       Feasible       99.2169         105       0.99       0.904       0.689       1.593       0.345       0.109       0.556       0.779       1433058       3922030       2907198       Infeasible       97.49322         24       0.39       0.905       0.6757 <td></td> <td>91</td> <td>0,99</td> <td>0,914</td> <td>0,68</td> <td>1,594</td> <td>0,397</td> <td>0,352</td> <td>0,111</td> <td>0,55</td> <td>0,78</td> <td>3 1</td> <td>42301</td> <td>0</td> <td>3788</td> <td>716</td> <td></td> <td>31491</td> <td>07</td> <td>F</td> <td>easible</td> <td>98,48174</td> <td>1  </td>		91	0,99	0,914	0,68	1,594	0,397	0,352	0,111	0,55	0,78	3 1	42301	0	3788	716		31491	07	F	easible	98,48174	1
1       0.99       0.914       0.68       1.594       0.397       0.382       0.111       0.55       0.783       1426372       3965803       3099806       Feasible       97.17546         87       0.39       0.304       0.683       1.593       0.382       0.361       0.108       0.557       0.773       143343       3875622       3178798       Feasible       97.17546         38       0.99       0.904       0.683       1.593       0.342       0.116       0.557       0.773       1433439       3875622       3178798       Feasible       97.95914         33       0.99       0.904       0.689       1.593       0.342       0.110       0.557       0.779       143200       398578       2987826       Feasible       99.2163         105       0.99       0.904       0.689       1.593       0.332       0.351       0.108       0.557       0.779       1432007       3916233       3322129       Infeasible       99.2163         105       0.99       0.905       0.675       1.58       0.33       0.345       0.109       0.536       0.781       14314707       3916233       3322129       Infeasible       99.2713         24		44	0,99	0,914	0,68	1,594	0,397	0,352	0,111	0,55	0,78	3 1	42261	6	3863	107		29863	17	F	easible	98,46788	1
87       0.99       0.904       0.689       1.593       0.332       0.361       0.108       0.557       0.772       1433439       3975822       3178798       Feasible       99.10907         58       0.99       0.907       0.674       1.561       0.345       0.11       0.555       0.772       1432439       3985578       22300       Infreasible       99.79914         33       0.99       0.904       0.689       1.593       0.322       0.351       0.108       0.557       0.779       1432700       3998578       Feasible       99.79914         105       0.99       0.904       0.689       1.593       0.322       0.351       0.108       0.557       0.779       1432700       3998578       Feasible       99.7169         5       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.556       0.781       1431007       3916233       3322129       Infreasible       99.74332         24       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.536       0.781       1431143       33222537       Infreasible       99.22713         53       0.99       0.905		1	0,99	0,914	0,68	1,594	0,397	0,352	0,111	0,55	0,78	3 1	42637	2	3868	503		30988	06	F	easible	97,17548	1
58       0.99       0.907       0.674       1.581       0.381       0.346       0.11       0.535       0.782       1432703       3889578       2387826       Feasible       99.9914         33       0.99       0.904       0.689       1.593       0.382       0.351       0.108       0.557       0.773       1432700       3896578       2387826       Feasible       99.2914         105       0.99       0.905       0.675       1.58       0.38       0.345       0.109       0.556       0.771       1432007       3322129       Infeasible       99.4332         5       0.99       0.904       0.689       1.593       0.332       0.351       0.108       0.556       0.779       1433058       3922033       2907198       Feasible       97.43322         24       0.39       0.905       0.675       1.58       0.39       0.345       0.109       0.556       0.781       1431143       39223650       315       1       1       1       97.9822713       1       1       1       143362       3945667       315       1       1       1       97.9822       1       1       97.9822       1       1       1       1       1		87	0,99	0,904	0,689	1,593	0,392	0,351	0,108	0,557	0,77	9 1	43943	9	3875	322		31787	98	F	easible	98,10807	1
33       0.99       0.904       0.689       1.593       0.392       0.315       0.108       0.576       0.775       1432007       3936578       2987826       Feadble       98.2169         105       0.39       0.904       0.689       1.593       0.345       0.103       0.536       0.781       1431007       3916239       3323123       infreasible       98.67434         5       0.99       0.904       0.689       1.593       0.325       0.516       0.108       0.557       0.779       1433058       3320203       2907198       Feasble       98.67434         24       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.536       0.781       143143       39225670       39       1.68       98.67434         53       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.536       0.781       1434970       39223650       35       1.68       97.97189         101       0.99       0.907       0.674       1.581       0.391       0.345       0.782       1.434970       3923650       35       1.68       39.2713       1.68       1.68       1.68       1.68		58	0,99	0,907	0,674	1,581	0,391	0,346	0,11	0,535	0,78	2 1	43278	3	3889	943		32623	00	lr	nfeasible	97,95914	1
105         0.99         0.905         0.675         1.58         0.39         0.345         0.109         0.536         0.781         1431007         3916233         3322129         Infeasible         96,7434           5         0.99         0.904         0.689         1.593         0.345         0.108         0.556         0.771         1433068         3920203         2907198         Feasible         97,44332           24         0.99         0.905         0.675         1.58         0.38         0.108         0.556         0.781         1431143         3922537         310000         Infeasible         98,27134           53         0.99         0.905         0.675         1.58         0.38         0.109         0.536         0.781         143143         3922537         310000         Infeasible         99,2713           53         0.99         0.905         0.675         1.58         0.38         0.199         0.536         0.781         1434970         3922850         35         Infeasible         97,99622         1         1433652         3445667         10         Infeasible         97,9718         20           5         0.80         0.81         0.755         0.780		33	0,99	0,904	0,689	1,593	0,392	0,351	0,108	0,557	0,77	9 1	43270	0	3896	578		29878	26	F	easible	98,2169	1
5       0.99       0.904       0.689       1.593       0.392       0.351       0.108       0.576       7.77       143058       3920203       2907198       Infeasible       97.49332         24       0.39       0.305       0.675       1.58       0.39       0.345       0.109       0.536       0.781       1431143       3922537       31       Infeasible       97.49332         53       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.556       0.781       1431143       3922537       31       Infeasible       97.982273         101       0.99       0.905       0.675       1.58       0.39       0.345       0.109       0.556       0.781       1434370       3922567       31       Infeasible       97.9822       1       1       197.9822       31       1       1434370       3923650       31       5       1       1       1       9.7872       1433362       3945667       31       1       1       1       97.9372       1433362       3945667       31       1       1       1       1       1       1       30       35       1       1       1       1       1       1		105	0,99	0,905	0,675	1,58	0,39	0,345	0,109	0,536	0,78	1 1	43100	7	3916	239		33231	29	lr	nfeasible	98,67434	1
24         0.99         0.905         0.675         1.58         0.39         0.345         0.109         0.536         0.781         1431143         3922537         31000000         Infeestble         9822713           53         0.39         0.905         0.675         1.58         0.39         0.345         0.109         0.536         0.781         1431143         3922537         3100000         Infeestble         9822713           101         0.99         0.905         0.675         1.58         0.39         0.345         0.109         0.536         0.781         1434370         3923650         3         5         Infeestble         97.99622         3         5         101         0.923         3105         1016         97.99623         3         5         101         10.836         0.781         1434970         3923650         3         5         10         97.9718		5	0,99	0,904	0,689	1,593	0,392	0,351	0,108	0,557	0,77	9 1	43305	8	3920	203		29071	98	F	easible	97,48392	1
53         0.99         0.905         0.675         1.58         0.39         0.345         0.109         0.536         0.781         1434970         3923650         31g         Infeesible         97.99622           1011         0.99         0.905         0.675         1.58         0.345         0.109         0.526         0.781         1434970         3923650         31g         Infeesible         97.99622           1011         0.99         0.907         0.674         1.581         0.341         0.346         0.11         0.755         1.433862         3945667         310         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         188         Infeesible         97.99622         18		24	0,99	0,905	0,675	1,58	0,39	0,345	0,109	0,536	0,78	1 1	43114	3	3922	537		31200		lr	nfeasible	98,22713	1
Infi         0.99         0.907         0.872         1.581         0.391         0.346         0.11         0.782         1.433362         3945667         310         10		53	0,99	0,905	0,675	1,58	0,39	0,345	0,109	0,536	0,78	1 1	43497	0	3923	650		315	-	lr	nfeasible	97,99622	
Solve Find Ascending Solve		101	0.66	N 907	0.674	1 581	0.391	0.346	011	0.535	0.78	2 1	43336	2	3945	667		3 10		l Ir	nfeasihle	97 97189	1 🗠
Solve     Grading     Grading     Find     Ascending     Ascending     Filtering       Solve     Descending     25 • 25 • 10 • 0 • 0 • 0 • 0 • 0 • 0 • 0 • 0 •		<																15				فارك	2
Find     Ascending       Save     Descending       Solver state:     DONE		Solve			Grad	ing —												20				Filtering	a –
Save     Descending       Solver state:     DONE		Find	1 400	anding	- II	-	7h)			le.		1.0	v		Gentle	- 6		(1 30					
Save Descending Solver state: DONE			A	entailing	4	.a / (2.a+	20) III	, 			, 	Jy		.v.,UV	dealv	UL 1	-usur, v	<sup>rui</sup> 35	~				
Solver state: DONE Next		Sava	Des	onding		25 🔻	25	• 2	25 👻	10	-	0 🗖	· 0	-	0	-	ו	- 0	-	D	ONE	SELE	CT 📋
Solver state: DONE Next				Jonding					_											_			_
Next		Solver sta	ate: D	ONE																	-		1
																						Nex	t

Figure A. 64 Input form, selection of gear pairs

Fire Charles Constraint	***********************										
Five-Stage Gear Train											×
Overal Gear Ratio:	222,222 Trial	Np1 Ng1 m	g1 Np2 Ng2 643 14 50	mg2 Np3	Ng3 mg3	Np4 Ng4 14 31	mg4	Np5 Ng5 15 31	mg5 2.067	Overall Ratio	
Number of gear trains:	Filtering Options for Five-	14 55 3 Gear Tr	ain	35 14	39 2 786	14 34	2 429	18 43	2,389	222,22151	-
Allow gear ratio to be inte									1,889	222,22151	-
Allow each ratio to be ed									1,889	222,22151	1
Calum states		First Stage	Second Sta	ge <u>Third S</u>	Stage Fourt	h Stage	Fifth Sta	age	1,833	222,22151	1
Solver state:	mp greater than:	1.25 💌	1.25	-	-	-		•	2,167	222,22208	
Start Interrupt								_	1,722	222,22208	
	Za / (Za+Zb) greater than:	0.56 💌	0.54	-	-	-		•	1,122	222,22200	
Selected xp3 xg	K 0 1 1	0.005	0.05					_	M4	b4 Kv	4. 🔨
3 0,281 -0,2	NV,UV greater than:	0.625	10.60		<b>_</b>	-	1	-	3,5	32,508 0	,9
39 0,226 -0,2	I greater than:	-		-	-	-		•	3,25	34,125 0	.9
4 0,281 -0,2							-		3,5	30,957 0	3
5 0.281 -0.2	Jp greater than:	-		-	-	-		•	3,25	33,262 0	9
6 0,281 -0,2								_	3,75	32,787 0	,e
19 0,243 -0,2	Jg greater than:	-			<u> </u>	•	1	-	3,25	30,453 0	,9
25 0,263 -0,2	- Outside Diameter (mm)-								3	31,627 0	,9
23 0,281 -0,2	deat a lange	4-24			d== 4 + [		e . 🕅		3,5	34,694 0	9
23 0,263 0,2				-	dop4 <		0 <	_	3,25	33,225 0	2
41 0,281 -0,2	dog1 < 🗨	dog2 < 50	dog3 <	•	dog4 <	▼ doc	5 <	•	3,5	36,429 0	9
31 0,311 -0,3		100		_	- ,			_	3,5	35,745 0	ē
9 0.266 -0.2		200							1	34 278 0	٩Ň
		250	_			Filter	Remo	ve Filter			
Solve		300	~							Filtering	,
Find Ascendi	ng Za/(Za+Zb)	mp T	Jp	Jg	Kv,Cv GearV	ol. Prism.\	/ol. Enc.	Vol.			
Save Descend	ling 25 💌	25 💌 25	▼ 10 ▼	0 - 0	• 0	• 0	• 15	▼ D	DNE	SELE	CT
Solver state: DONE											
										Nex	:

Figure A. 75 Input form, filtering options

Five	Stage Output									
Grad	hics									
	STAGE-1 STAGE-2	STAG	E-3	STAGE-4	STAGE-5	GEAR TRAIN				
	Pinion Tooth Profile	<u> </u>		Gear Tooth Profile		Strength and Durability Rating				
	Gear Drive Properties-1			Gear Drive Properties-2	<u> </u>	Factors				
	Number of Teeth: Normal Pressure Angle(degrees): Pitch Diameter(mm): Operating Pitch Diameter(mm): Base Diameter(mm): Dutside Diameter(mm): Root diameter(mm): Face Width(mm): Circular Tooth Thickness at Tooth Tip.(1/m): Addendum Modification Coefficient:	PINION 15 20 15 15 14,095389 17,394 12,894 11,638 0,5315617 0,197	ק ק ק ק ק ק ק	<b>GEAR</b> 50 20 50 50 46,984631 51,606 47,106 11,638 0,7789768 0,197						
							H			

Figure A. 86 Output form, gear drive properties 1

hics					
STAGE-1 STAGE-2	STAGE-3	STAGE-4	STAGE-5	GEAR TRAIN	
Pinion Tooth Profile	Y	Gear Tooth Profile		Strength and Du	rability Rating
Gear Drive Properties-1		Gear Drive Properties-2		Factors	
Module(mm): Operating Pressure Angle(degrees): Total Addendum Modification:	1  20  0	Total Length of Action(m Bottom Clearence(mm): Contact Ratio:	im): 4,6528 0,25 1,5761	0833	
Standard Center Distance(mm):	32,5	Total Gear Volume(mm^3	3): 24907,	7639	
Operating Center Distance(mm):	32,5				
Pinion Addendum Length of Action(mm):	2,53071087				
Pinion Dedendum Length of Action(mm):	2,12216808				

Figure A. 97 Output Form, Gear Drive Properties 2

e-Stage Output				
STAGE-1 STAGE-2	STAGE-3	STAGE-4	STAGE-5	GEAR TRAIN CONFIGURATIO
Pinion Tooth Profile		Gear Tooth Profile	Y	Strength and Durability Rating
Gear Drive Properties-1	Ţ.	Gear Drive Properties-2	Y.	Factors
Application Factor(Ca,Ka): Surface Factor(Cf,Kf): Reliability Factor(Cr,Kr): Size Factor(Cs,Ks): Temparature Factor(Ct,Kt): Dynamic Factor(Cv,Kv): Load Distribution Factor(Cm,Km): Life Factor for Pitting Resistance(CL): Life Factor for Pitting Resistance(KL): Hardness Ratio Factor (Ch): Geometry Factor for Pitting Resistance( Geometry Factor for Bending Resistance	PINION           1.5           1           1,0000001           1           0,9524021           1           1,0000194           1,0176434           Not valid           1):         0,364	<b>GEAR</b> 1.5         1         1.0000001         1         0.9524021         1         1.0697302         1.0396834         1.0016         0.356		

Figure A. 108 Output form, factors

Five-Stage Output				X
STAGE-1 STAGE-2 S	TAGE-3 STAGE-4	STAGE-5	GEAR TRAIN	
Gear Drive Properties-1	Gear Drive Properties-2		Factors	
Pinion Tooth Profile	Gear Tooth Profile	Y'	Strength and Durabi	lity Rating
Draw Eras	•			

Figure A. 119 Output form, pinion tooth profile

e-Stage Output phics								
STAGE-1 STAGE-2	STAGE-3	STAGE-4	STAGE-5	GEAR TRAIN	CONFIGURATIC			
Gear Drive Properties-1		Gear Drive Properties-2		Factors				
Pinion Tooth Profile	Ţ.	Gear Tooth Profile		Strength and Durability	Rating			
Bending Stress(MPa): Permissible Bending Stress(MPa): Contact Stress(MPa): Permissible Contact Stress(MPa): Tangential Load(N): Torque(Nmm): Pitch Line Velocity(m/s): Load Cycles:	PINION           189.34824                     386.70446                     1243.5447                     1250.0241                     509.29581                     3819.7186                     1.9634954                     10000000	GEAR 193.60326 395.07965 1243.5447 1339.316 509.29581 3819.7186 1.9634954 3000000						

Figure A. 20 Output form, strength and durability rating

Five-Stage Output										×
Graphics										
STAGE-1 STAGE-2	2	STAGE-3	) ľ	STAGE-4	Ϋ́	STAGE-5	GI	EAR TRAIN	CONF	FIGURATION
Stage		1		2		3		1	Ę	5
Gear	Pinion1	Gear1	Pinion2	Gear2	Pinion3	Gear3	Pinion4	Gear4	Pinion5	Gear5
Number of teeth	15	50	17	54	18	51	18	50	18	48
Gear ratio	3,3	333	3,1	765	2,8	333	2,7	778	2,6	567
Module,(mm)		1	1,	25	1,	75	2	.5	3,2	25
Addendum modification coefficient	0,197	-0,197	0,284	-0,284	0,256	-0,256	0,248	-0,248	0,242	-0,242
Pitch diameter,(mm)	15	50	21,25	67,5	31,5	89,25	45	125	58,5	156
Outside diameter,(mm)	17,394	51,606	24,46	69,29	35,896	91,854	51,24	128,76	66,573	160,927
Face width,(mm)	11,	638	16,	062	20,783		25,0	588	38,	045
Contact stress,(MPa)	1243	,5448	1308	2199	1395.	,0814	1479	.7573	1564	.5249
Permissible contact stress,(MPa)	1250,0242	1339,316	1337,1627	1428,713	1426,5702	1514,157	1512,2439	1603,25	1601,2859	1693,645
Bending stress,[MPa]	189,3482	193,6033	229,5748	264,339	270,2218	306,9658	303,0907	342,587	336,3481	379,3472
Permissible bending stress,[MPa]	386,7045	395,0797	395,0797	453,4365	453,4365	513,3699	513,3699	579,8546	5/9,8546	651,7703
Bending resistance geometry factor,[J]	0,364	0,356	0,403	0,35	0,401	0,353	0,399	0,353	0,397	0,352
Contact resistance geometry factor,[1]	0,1	084	0,1	16	0,1	128	0,1	119	0,1	
Addendum portion of the line of action	0,8572	0,7189	0,9293	0,6508	0,9226	0,6713	0,9178	0,6769	0,9142	0,6795
Circular tratto	1,5	/61	1,0	501	1,5	939	1,3	347	1,00	337
Chaft diamates (was)	0,0316	0,779	0,515	0,7913	0,0426	0,7867	0,0460	0,8001	0,5494	0,7832
Shart diamater,(mm)	5	· · · ·	0		2		6	2	3	32
									5	Save to Excel

Figure A. 21 Output form, gear train

Figure A. 22 Output form, configuration



Figure A. 23 Output form, graphics, and performance variables



Figure A. 24 Output form, graphics, ratio split and module



Figure A. 25 Output form, graphics, pitch diameter and face width



Figure A. 26 Output form, graphics, and dynamic factors



Figure A. 27 Output form, graphics, and volume

Application Factors, Ka and Ca

### **APPLICATION FACTORS, Ka and Ca**

The loading model assumed that the transmitted load Wt was uniform with time. The fluctuating moments on the teeth described in that section are due to the teeth coming into and out of mesh under a uniform or average load. If either the driving or driven machine has time-varying torques or forces, then these will increase the loading felt by the gear teeth above the average values. In the absence of definitive information about the dynamic loads in the driving and driven machines, an application factor Ka can be applied to increase the tooth stress based on the "shockiness" of the machinery connected to the gear train. For example, if the gear train connects an electric motor to a centrifugal water pump (both of which are smooth-running devices) there is no need to increase the average loads and Ka=1. But, if a single-cylinder, internal-combustion engine drives a rock crusher through a gear train, both the power source and the driven device deliver shock loads to the gear teeth and Ka > 1. Table below shows some AGMA-suggested values for Ka based on the assumed level of shock loading in driving and driven devices.

×

Recomme	ended AGMA app	lication factors, Ka	- v 1993)						
	DRIVEN MACHINE								
POWER SOURCE	Uniform	Moderate Shock	Heavy Shock						
Uniform (Electric motor, turbine)	1.00	1.25	1.75 or higher						
Light shock (Multicylinder engine)	1.25	1.50	2.00 or higher						
Medium shock (Single-cylinder engine)	1.50	1.75	2.25 or higher						

Figure A. 28 Help form, application factor

Surface Condition Factor, Cf
SURFACE CONDITION FACTOR, CF
The surface condition factor, Cf , used only in the pitting resistance formula, depends on:
<ul> <li>(1) Surface finish as affected by, but not limited to, cutting, shaving, lapping, grinding, shot peening.</li> <li>(2) Residual stress</li> </ul>
<ul> <li>(3) Plasticity effects (work hardening) Standard surface condition factors for gear teeth have not yet been</li> </ul>
established for cases where there is a detrimental surface finish effect.
In such cases, some surface finish factor greater than unity should be used.

Figure A. 29 Help form, surface condition factor



Figure A. 3012 Help form, size factor

## Temperature Factors, Kt and Ct 👘

## **TEMPERATURE FACTORS, Kt and Ct**

Temperature factor is taken as unity when gears operate at temperatures of oil or gear blank not exceeding 120 degrees centigrade (250 degrees fahrenite). If temperatures are above this value than ,it should be a value greater than unity to allow for the effect of temperature on oil film and material properties.

×

Consideration must be given to the loss of hardness and strength of some materials due to tempering effect of temperatures over 175 degrees centigrade( 350 degrees fahrenite)

Figure A. 31 Help form, temperature factor

Reliability Factors, Kr and Cr	X	
RELIABILITY FACTORS, Kr and Cr		
The reliability factors account for the effect of the normal statistical distribution of failures found in materials testing. The allowable stress numbers given in AGMA tables are based upon a statistical probability of one failure in 100 at 10^7 cycles. Table below contains reliability factors which may be used to modify these allowable stresses to change that probability. These numbers are based upon general statistical studies and other values may be used if specific data is used by		
REQUIREMENTS OF APPLICATION	Kr,Cr *	
Fewer than one failure in 10000	1.50	
Fewer than one failure in 1000	1.25	
Fewer than one failure in 100	1.00	
Fewer than one failure in 10	0.85 #	
# At this value plastic flow might occur rather than pitting		
<ul> <li>Tooth breakage is sometimes considered a greater hazard than pitting. In such cases a value of Kr greater than Cr is selected</li> </ul>		





Figure A. 33 Help form, life factor,  $K_L$ 

#### Life Factors, KL and

The life factors, CL and KL, adjust the allowable stress numbers for the required number of cycles of operation. For the purpose of this Standard, N, the number of load cycles is defined as the number of mesh contacts, under load, of the gear tooth being analyzed. AGMA allowable stress numbers are established for 10°7 tooth load cycles at 39 percent reliability. The life factor adjusts the allowable stress numbers for design lives other than 10 cycles.

The life factor accounts for the S/N characteristics of the gear material as well as for the gradual increased tooth stess which may occur from tooth wear, resulting in increased dynamic effects and from shifting load distribution which may occur during the design life of the gearing.

A CL or KL value of unity (1) may be used, where justified by experience

At the present time there is insufficient data to provide accurate life curves for all types of gears and gear applications. Experience, however, suggests life curves f pitting and strength of steel gears as shown in below figures. These figure do not include data for initided gears

he upper potion of the shaded zones on the figures may used for general commercial applications. The lower tion of the shaded zones are uplically used for critical sivice applications where little pitting and tooth wear is smissible and where smoothness of operation and low tration levels are required.







Figure A. 35 Help form, straddle mounting





Overhung Mounting Help		
Load Distribution factors,Cm and Km		
The Load distribution factor modifies the rating equations to reflect the non-uniform distribution of the load along the lines of contact. The amount of non uniformity of the load distribution is caused by,and is dependent upon,the folowing influences:		
<ol> <li>Gear tooth manufacturing accuracy i.e. lead profile and spacing.</li> <li>Alignment of the axes of rotation of the pitch cylinders of the mating gear elements.</li> <li>Elastic deflections of gear elements, shafts, bearings, housings, and foundations which support the gear elements.</li> <li>Bearing clearances.</li> <li>Hertzian contact and bending deformations at the tooth surface.</li> <li>Thermal expansion and distortion due to operating termparature(especially important for wide face gearing.</li> <li>Cetrifugal deflections due to operating speed.</li> <li>Tooth crowning and end relief.</li> </ol>		
Cm = Km = Cmf * Cmt		
where		
Cmf = face load distribution factor Cmt = transverse loa	ad distribution factor(mostly unity value used)	
Contact across the entire facewidth		
Cmf = 1.0 + (Get. F / 2.Wt)		
Contact across the entire facewidth		
Cmf = square root ( (2, G .et, F) / Wt )		
where		
G = tooth stiffness constant, (Mpa)	et =total lead mismatch between mating	
The average mesh stiffness of a single gear pair of teeth in the normal direction. The usual range of this value that is compatible with this analysis, for steel gears is 10000 to 14000 Mpa. The most conservative value is the highest.	teetn.(mm)	

Figure A. 37 Help form, overhung mounting



## Figure A. 38 Help form, AGMA quality numbers

# Gear Materials × Only a limited number of metals and alloys are suitable for gears that transmit significant power. Steels, cast irons, and malleable and nodular irons are the most common choices for gears. Surface- or through-hardening is recommended (on those alloys that allow it) to obtain sufficient strength and wear resistance. Where high corrosion resistance is needed, such as in marine environments, bronzes are often CAST IRONS are commonly used for gears. The gray cast irons (CI) have advantages of low cost, ease of machining, high wear resistance, and internal damping (due to the graphite inclusions), which makes them acoustically quieter than steel gears. However, they have low tensile strength, which requires larger teeth than steel gears to obtain sufficient bending strength. Nodular irons have higher tensile strength than gray CI and retain the other advantages of machinability, wear resistance, and internal damping, but are more costly. The combination of a steel pinion (for strength in the higher-stressed member) and a cast-iron gear is often used. STEELS are also commonly used for gears. They have superior tensile strength to cast iron and are cost-competitive in their low-alloy forms. They need heat treatment to get a surface hardness that will resist wear, but soft steel gears are sometimes used in low-load, low-speed applications or where long life may not be a prime concern. For heat treatment, either a medium-to-high carbon (0.35 to 0.60% C) plain or alloy steel is needed. Small gears are typically through-hardened and larger gears flame or induction hardened to minimize distortion. Lower-carbon steels can be case hardened by carburizing or nitriding. A case-hardened gear has the advantage of a tough core and a hard surface, but if the case is not deep enough, the teeth may fail in bending fatigue beneath the case in the soft, weaker core material. It is often necessary to use econdary finishing methods such as grinding, lapping, and honing to remove the heat-treatment distortion from hardened gears if high accuracy is needed. BRONZES are the most common nonferrous metals used for gears. The lower modulus of elasticity of these copper alloys provides greater tooth deflection and improves load sharing between the teeth. Since bronze and steel run well together, the combination of a steel pinion and a bronze gear is often used. NONMETALLIC GEARS are often made of injection-molded thermoplastics such as nylon and acetal, sometimes filled with inorganics such as glass or talc. Teflon is sometimes added to nylon or acetal to lower the coefficient of friction. Dry lubricants such as graphite and molybdenum disulphide (MoS2) can be added to the plastic to allow dry running. Composite gears of cloth-reinforced thermosetting phenolic have long been used for applications such as the camshaft-drive (timing) gear driven by a steel pinion in some gasoline engines. Nonmetallic gears have very low noise but are limited in torque capacity by their low material strengths.

Figure A. 39 help form, gear materials



Figure A. 40 Help form, gear manufacturing, forming gear teeth

### **Roughing Processes**

FORM MILLING requires a shaped milling cutter. The cutter must be made to the shape of the gear tooth space for the tooth geometry and number of teeth of each particular gear. The rotating cutter is plunged into the blank to cut one tooth at a time. The gear blank is then rotated through one circular pitch and the next tooth cut. Because a different shape cutter is needed for each gear size to be made, the tooling cost becomes high. To reduce costs, the same cutter is often used for multiple-sized gears resulting in profile errors for all but one number of teeth. This method is the least accurate of the roughing methods.

RACK GENERATION A rack cutter for any involute pitch can be easily made since its tooth shape is a trapezoid. The hardened and sharpened rack is then reciprocated along the axis of the gear blank and fed into it while being rotated around the gear blank so as to generate the involute tooth on the gear. The rack and gear blank must be periodically repositioned to complete the circumference. This repositioning can introduce errors in the tooth geometry, making this method less accurate than others still to be discussed.

GEAR SHAPING uses a cutting tool in the shape of a gear, which is reciprocated axially across the gear blank to cut the teeth while the blank ro tates around the shaper tool. It is a true shape-generation process in that the gear-shaped tool cuts itself into mesh with the gear blank. The accuracy is good, but any errors in even one tooth of the shaper cutter will be directly trans ferred to the gear. Internal gears can be cut with this method as well.

HOBBING A hob, is analogous to a thread tap. Its teeth are shaped to match the tooth space and are interrupted with grooves to provide cutting surfaces. It rotates about an axis perpendicular to that of the gear blank, cutting into the rotating blank to generate the teeth. It is the most accurate of the roughing processes since no repositioning of tool or blank is required and each tooth is cut by multiple hob-teeth, averaging out any tool errors. Excellent surface finish can be achieved by this method and it is one of the most widely used for production gears.

Figure A. 41 Help form, gear manufacturing, machining, roughing process



Figure A. 42 Help form, gear manufacturing, machining, finishing process



Figure A. 43 Solid modeling of a five-stage spur gear drive in Autodesk Inventor

## **APPENDIX B**

## CONFIGURATION DESIGN CONSTRAINTS FOR SIX STAGE SPUR GEAR DRIVE

MODEL: [OBJECTIVE] MIN=(((x1-x4)^2+(y1-y4)^2)^0.5)+(((x1-x6)^2+(y1- $y10)^2)^0.5)+(((x1-x12)^2+(y1-y12)^2)^0.5);$ Pinion and gear center distance contraints(c1-c12) [c1]  $(dp1+dp2)*0.5-(((x1-x2)^2+(y1-y2)^2)^0.5)=0;$ [c2]  $(dp3+dp4)*0.5-(((x3-x4)^2+(y3-y4)^2)^0.5)=0;$  $[c3] (dp5+dp6)*0.5-(((x5-x6)^2+(y5-y6)^2)^0.5)=0;$ [c4] (dp7+dp8)\*0.5-(((x7-x8)^2+(y7-y8)^2)^0.5)=0; [c5] (dp9+dp10)\*0.5-(((x9-x10)^2+(y9-y10)^2)^0.5)=0; [c6] (dp11+dp12)\*0.5-(((x11-x12)^2+(y11-y12)^2)^0.5)=0; [c7] z1-z2=0; [c8] z3-z4=0; [c9] z5-z6=0; [c10] z7-z8=0; [c11] z9-z10=0; [c12] z11-z12=0; Gears having same shaft distance constraints(c13-c27) [c13] z3=z2+((b2+b3)\*0.5);[c14] x2-x3=0; [c15] y2-y3=0; [c16] z5=@IF( dp3+dp4 #GT# do2+do5, z4-((b4+b5)\*0.5),z4+((b4+b5)\*0.5)); [c17] x4-x5=0; [c18] y4-y5=0; [c19] z7=@IF( ((dp5+dp6 #LT# do4+do7) #AND# (Z5#LT# Z4)) #OR# ((dp5+dp6 #GT# do4+do7) #AND# (Z5#GT# Z4)) , z6-((b6+b7)\*0.5),z6+((b6+b7)\*0.5)); [c20] x6-x7=0; [c21] y6-y7=0; [c22] z9=@IF((dp7+dp8 #GT# do6+do9) , z8-((b8+b9)\*0.5), z8+((b8+b9)\*0.5));[c23] x8-x9=0; [c24] y8-y9=0; [c25] z11=@IF((dp9+dp10 #GT# do8+do11) , z10-((b10+b11)\*0.5),z10+((b10+b11)\*0.5)); [c26] x10-x11=0; [c27] y10-y11=0;

Gear interference constraints(c28-c57)
[c28] (((x1-x5)^2+(y1-y5)^2)^0.5)-(0.5*(do1+do5))>0;
[c29] (((x1-x6)^2+(y1-y6)^2)^0.5)-(0.5*(do1+do6))>0;
[c30] (((x1-x7)^2+(y1-y7)^2)^0.5)-(0.5*(do1+do7))>0;
[c31] (((x1-x8)^2+(y1-y8)^2)^0.5)-(0.5*(do1+do8))>0;
$[c32]$ (((x1-x9)^2+(y1-y9)^2)^0.5)-(0.5*(do1+do9))>0;
[c33] (((x1-x10)^2+(y1-y10)^2)^0.5)-(0.5*(do1+do10))>0;
[c34] (((x1-x11)^2+(y1-y11)^2)^0.5)-(0.5*(do1+do11))>0;
[c35] (((x1-x12)^2+(y1-y12)^2)^0.5)-(0.5*(do1+do12))>0;
$[c_{36}] (((x_{2}-x_{5})^{2}+(v_{2}-v_{5})^{2})^{0},5)-(0.5*(d_{0}2+d_{0}5))>0;$
$[c_{37}] (((x_{2}-x_{6})^{2}+(y_{2}-y_{6})^{2})^{0},5)-(0,5*(d_{02}+d_{06}))>0;$
$[c_{38}]$ (((x2-x7)^2+(y2-y7)^2)^0.5)-(0.5*(do2+do7))>0;
[c39] (((x2-x8) <sup>2</sup> +(y2-y8) <sup>2</sup> ) <sup>0</sup> ,5)-(0,5*(do2+do8))>0;
[c40] (((x2-x9) <sup>2</sup> +(y2-y9) <sup>2</sup> ) <sup>0</sup> ,5)-(0,5*(do2+do9))>0;
$[c41] (((x^2-x10)^2+(y^2-y10)^2)^0 5) - (0.5*(do^2+do10)) > 0;$
$[c42] (((x^2-x^{11})^2+(y^2-y^{11})^2)^{0} 5) - (0.5*(do^2+do^{11})) > 0;$
$[c43] (((x2-x12)^{2}+(y2-y12)^{2})^{0},5) - (0,5*(do2+do12)) > 0;$
$[c44] (((x3-x7)^2+(v3-v7)^2)^0, 5)-(0, 5*(do3+do7))>0;$
$\begin{bmatrix} c45 \end{bmatrix} (((x3-x8)^{2}+(x3-x8)^{2})^{0} 5) - (0.5^{*}(do3+do8)) > 0;$
$\begin{bmatrix} c46 \end{bmatrix} (((x3-x9)^{2}+(y3-y9)^{2})^{0}, 5) - (0, 5*(do3+do9)) > 0;$
$\begin{bmatrix} c47 \end{bmatrix} (((x3-x10)^{2}+(x3-x10)^{2})^{0} = 5) - (0 = 5 \times (do3+do10)) > 0;$
$[c48] (((x3-x11)^2+(x3-v11)^2)^0 5) - (0.5*(do3+do11))>0;$
$[c49]$ (((x3-x12)^2+(x3-x12)^2)^0 5)-(0.5*(do3+do12))>0:
$\begin{bmatrix} c50 \end{bmatrix} (((x3 x12) 2+(y3 y12) 2) 0.5) (0$
$\begin{bmatrix} c51 \end{bmatrix} (((x1 x7) 2+(y1 y7) 2) 0.5) (0.5 (d01+d07)) > 0; \\ \begin{bmatrix} c51 \end{bmatrix} (((x4-x8)^2+(x4-x8)^2)^0 5) = (0.5 + (d04+d08)) > 0; \\ \end{bmatrix}$
$\begin{bmatrix} c52 \end{bmatrix} (((x4-x9)^2+(x4-x9)^2)^0, 5) = (0.5^*(do4+do9)) > 0;$
$\begin{bmatrix} a52 \end{bmatrix} (((x+x)) 2 + (y+y) 2) 0 = 0 = 0 = 0 = 0 = 0 = 0 = 0 = 0 = 0 $
$\begin{bmatrix} c53 \end{bmatrix} (((x4 + x11))^2 + (y4 + y11))^2 + (0.5)^2 + (0$
$\begin{bmatrix} c54 \end{bmatrix} (((x4 - x12))^{2} + (y4 - y12)^{2})^{2} = (0.5^{\circ}(0.5^{\circ}(0.04 + 0.01))^{2})^{2}$
$\begin{bmatrix} a56 \end{bmatrix} (((x7-x12))^2 + (y7-y12)^2) = (0.5)^2 (0.5)^2 (0.5)^2 (0.5)^2 + (0$
$\begin{bmatrix} (0.50) & (((x) - xI2) & 2 + (y) - yI2) & 2 & (0.5) - (0.5" (00) + 00I2) > 0; \\ \begin{bmatrix} 0.57 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0.57 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0.57 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0.57 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix} = \begin{bmatrix} 0.57 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 \end{bmatrix}$
$[(c_{3})] (((x_{0}-x_{12}))^{2}+(y_{0}-y_{12})^{2})^{-}(0,5)-(0,5)^{-}(0,$

## Constraints between gear and shaft(c58-c92)

```
[c82] (((x10-x1)^2+(y10-y1)^2)^0.5)-(0.5*(ds6+do1))>0;
[c83] (((x10-x2)^2+(y10-y2)^2)^0.5)-(0.5*(ds6+do2))>0;
[c84] (((x10-x4)^2+(y10-y4)^2)^0.5)-(0.5*(ds6+do4))>0;
[c85] (((x10-x6)^2+(y10-y6)^2)^0.5)-(0.5*(ds6+do6))>0;
[c86] (((x10-x8)^2+(y10-y8)^2)^0.5)-(0.5*(ds6+do8))>0;
[c87] (((x12-x1)^2+(y12-y1)^2)^0.5)-(0.5*(ds7+do1))>0;
[c88] (((x12-x2)^2+(y12-y2)^2)^0.5)-(0.5*(ds7+do2))>0;
[c89] (((x12-x4)^2+(y12-y4)^2)^0.5)-(0.5*(ds7+do4))>0;
\label{eq:c91} \end{tabular} (((x12-x8)^2+(y12-y8)^2)^0.5)-(0.5*(ds7+do8))>0;
[c92] (((x12-x10)^2+(y12-y10)^2)^0.5)-(0.5*(ds7+do10))>0;
DATA:
Input
x1=0;
y1=0;
Pitch diameters of pinion and gear
dp1=@pointer( 1);
dp2=@pointer( 2);
dp3=@pointer( 3);
dp4=@pointer( 4);
dp5=@pointer( 5);
dp6=@pointer( 6);
dp7=@pointer( 7);
dp8=@pointer( 8);
dp9=@pointer( 9);
dp10=@pointer( 10);
dpl1=@pointer( 11);
dp12=@pointer( 12);
Outer diameters of pinion and gear
dol=@pointer( 13);
do2=@pointer( 14);
do3=@pointer( 15);
do4=@pointer( 16);
do5=@pointer( 17);
do6=@pointer( 18);
do7=@pointer( 19);
do8=@pointer( 20);
do9=@pointer( 21);
dol0=@pointer( 22);
dol1=@pointer( 23);
dol2=@pointer( 24);
Face width
b1=@pointer( 25);
b2=@pointer( 26);
b3=@pointer( 27);
b4=@pointer( 28);
b5=@pointer( 29);
b6=@pointer( 30);
b7=@pointer( 31);
b8=@pointer( 32);
b9=@pointer( 33);
```

```
b10=@pointer( 34);
b11=@pointer( 35);
b12=@pointer( 36);
Shaft diameter
ds1=@pointer( 37);
ds2=@pointer( 38);
ds3=@pointer( 39);
ds4=@pointer( 40);
ds5=@pointer( 41);
ds6=@pointer( 42);
ds7=@pointer( 43);
@pointer(44) = x1;
@pointer( 45) = y1;
@pointer( 46) = z1;
@pointer( 47) = x2;
@pointer( 48) = y2;
@pointer( 49) = z2;
@pointer( 50) = x3;
@pointer( 51) = y3;
@pointer(52) = z3;
@pointer(53) = x4;
@pointer(54) = y4;
@pointer( 55) = z4;
@pointer( 56) = x5;
@pointer( 57) = y5;
@pointer( 58) = z5;
@pointer( 59) = x6;
@pointer( 60) = y6;
@pointer( 61) = z6;
@pointer(62) = x7;
@pointer( 63) = y7;
@pointer( 64) = z7;
@pointer( 65) = x8;
@pointer( 66) = y8;
@pointer( 67) = z8;
@pointer( 68) = x9;
@pointer( 69) = y9;
@pointer( 70) = z9;
@pointer( 71) = x10;
@pointer( 72) = y10;
@pointer( 73) = z10;
@pointer( 74) = x11;
@pointer( 75) = y11;
@pointer( 76) = z11;
@pointer( 77) = x12;
@pointer( 78) = y12;
@pointer( 79) = z12;
@pointer( 80) = @STATUS();
ENDDATA
END
```