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# Flow Characterization of Viscoelastic Fluids around Square Obstacle

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#### Abstract

This study focuses on the computational implementation of structured non-uniform finite volume method for the 2-D laminar flow of viscoelastic fluid past a square section of cylinder in a confined channel with a blockage ratio 1/4 for  $Re = 10^{-4}$ , 5, 10 and 20. Oldroyd-B model (constant viscosity with elasticity) and the PTT model (shear-thinning with elasticity) are the constitutive models considered. In this study effects of the elasticity and inertia on the drag coefficients and stress fields around the square cylinder are obtained and discussed in detail. With an increase elasticity, drag coefficients get smaller due to stronger shear thinning effects for PTT fluid, however, the drag coefficients show slightly enhancement for the Oldroyd-B fluid.

#### Keywords

square cylinder, viscoelastic flow, PTT fluid, Oldroyd-B

# **1** Introduction

Numerical analysis of flow past the obstacles has received considerable attention for a long time [1-5]. Fluids around obstacles is encountered in many industrial applications such as shell and tube heat exchangers, coating processes, cooling towers, extruders and membrane processes. The bulk of the literature of fluid flow studies relate to flow over circular cylinders, followed by square, elliptic and rectangular cylinders or obstacles [6-8]. Indeed, even for the simplest shape of a circular cylinder which is free from geometrical singularities [7], the flow exhibits a rich variety of phenomena depending upon the nature of the mainstream flow (type of fluid Newtonian or non-Newtonian), blockage ratio of the cylinder (length to diameter ratio) and the characteristic Reynolds number of the flow. The behavior of a fluid flow past a square cylinder often has many complex phenomena such as flow separation, vortex shedding, recirculation length of wake flow (the flow path behind the square cylinder), distribution of the shear and normal profile around the solid surfaces of the square cylinder and, drag and lift force coefficients [9-12]. Majority of these studies in the literature deal with flow of a Newtonian fluid around a circular

obstacle and to a smaller extend, around a square obstacle. For example Breuer et al. [9] examined laminar Newtonian flow around a square cylinder in a 2D channel using two different computational techniques, finite volume and lattice-Boltzmann automata. Their blockage ratio B, defined as the ratio between the obstacle dimension and channel height, was 1/8. They compared the results of the techniques in terms of velocity field, drag coefficient, Cd, recirculation length and Strouhal number. They observed an excellent agreement between the results of the techniques. A numerical study to investigate Newtonian flow past a square cylinder for Reynolds numbers  $Re \leq 40$  was conducted by Sen et al. [10] using a stabilized finite-element formulation with a non-uniform structured mesh. In order to mimic an unconfined flow, a small blockage ratio, B = 1/100, was employed. For comparison purposes they also used cylinders of elliptical and circular cross-sections. They investigated impact of Re on the flow separation angle and Cd. They found that, flow separation over the square cylinder occurs at a smaller Re compared to the other cylinders giving rise to the highest Cd among the investigated obstacles. The flow structures

and wake flow characteristics, vortex shedding behavior of behind square cylinder at high Reynolds numbers were experimentally studied using particle image velocimetry (PIV) with different blockage ratios by Saha et al. [11]. They employed the two-dimensions flow past a stationary square cylinder at zero incidence for Reynolds number,  $Re \leq 150$  using a stabilized finite-element formulations. They also compared with a circular cylinder that the flow separated at a much lower Re from a square cylinder leading to the formation of a bigger wake. Consequently, at a given Re, the drag on a square cylinder was higher more than that on drag of a circular cylinder.

In spite of their important industrial implications, number of studies involving non-Newtonian fluids around enclosed obstacles is much smaller than those on Newtonian fluids in the literature. For example in their computational study Dhiman et al. [12] employed finite volume technique to investigate 2D flow of power-law fluids around a confined square cylinder. The fluids had index values between  $0.5 \le n \le 2.0$ . Their results revealed that the effects of *Re* and *B* on the size of the recirculation zone and on *Cd* were stronger than that of the power-law index.

Rao et al. [13] extended the results on momentum and heat transfer characteristics to highly shear thinning fluids, especially  $n \le 0.5$  at low Reynolds number. Fluid elements followed the contour of the square cylinder and flow remained attached to the surface. These works reveal that shear-thinning behavior increases both *Cd* and the rate of convective heat transfer from the square cylinder surface. Shear thinning behavior not only delays the formation of a visible wake but the resulting wake is also somewhat shorter than that of Newtonian fluid case. The shear thickening, on the other hand, has exactly the opposite influence on wake formation.

In the case of non-Newtonian fluid flow around confined obstacles, the literature is dominated by the studies with generalized Newtonian model to capture shear thinning or thickening effects. On the other hand, investigating the effects of viscoelasticity on the hydrodynamics of the flow around the obstacles has potentially crucial implications on many industrial applications. Therefore the authors believe that, the absence of viscoelastic flows around various obstacles in the literature merits a study on the hydrodynamics of such industrially important flows to reveal both microscopic and macroscopic flow characteristics. Hence, the objective of the present study is to investigate flow of a viscoelastic fluid, a linear PTT fluid and Oldroyd-B fluid, around a confined square obstacle computationally. Finite volume method is employed to solve coupled equations of continuity, motion and constitutive model along with appropriate boundary conditions. Effects of inertia in terms of *Re*, elasticity in terms of Weissenberg number, *We*, and constitutive equation parameters on the recirculation length, drag and on the flow field in terms of stress and velocity fields are examined and presented in detail.

#### **2** Numerical Method

In this study isothermal flow of a viscoelastic fluid over a 2D confined square cylinder is considered. The flow system is schematically depicted in Fig. 1. The ratio between heights of the square and the channel, referred to as the blockage ratio, is 1/4 (b/H = 1/4). In the computations the upstream region length is set as 1/6 of the total channel length, *L* to ensure fully developed flow region. The ratio between channel length and height was set as L/H = 30.

The formulation of the flow begins by considering basic equations of fluid flow, i.e. continuity and momentum equations given below.

$$\nabla . u = 0 \tag{1}$$

$$\rho \frac{\partial u}{\partial t} + \rho u \cdot \nabla u = -\nabla p + \nabla \cdot \tau + \beta \eta_0 \nabla^2 u \tag{2}$$

where *u* is the velocity, *p* is the pressure,  $\eta_0$  is the total viscosity and  $\tau$  represents the polymer or non-Newtonian contribution to the deviatoric stress tensor. The constant  $\beta$  is the ratio between the solvent viscosity and the total viscosity ( $\beta = \eta_s/\eta_0$ ). A viscoelastic constitutive model provides the additional relation needed to solve the conservation equations. In this study, Linear PTT model [14, 16, 17] and Oldroyd-B model [14, 15, 17] given by Eqs. (3)-(5) are employed. Linear PTT model captures both shear thinning and normal stress effects in the flow. The PTT fluid model generally refers to a nonlinear viscoelastic equation derived by Phan-Thien and Tanner using the network theory. Unlike other non-Newtonian fluids, a distinctive advantage of PTT model is the inclusion of an

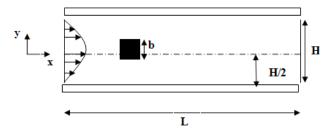


Fig. 1 Schematic of 2D flow around a square cylinder

extensional parameter, ( $\varepsilon$ ). The extensional parameter is considered  $\varepsilon \ge 0$ , as a constant. The extensional parameter imposes an upper bound on the extensional viscosity which is inversely proportional to  $\varepsilon$  [16, 17]. It is worth mentioning that  $\varepsilon = 0.25$  in the PTT model correspond to the flow behavior of extremities for concentrated polymer melts polymer solutions [17].

The linear PTT model [14, 16]:

$$\lambda \left[ \frac{\partial u}{\partial t} + \nabla . u\tau \right] + f \left( Tr\tau \right) \tau = \eta_s \left( \nabla u + \nabla u^T \right)$$
  
+  $\lambda \left( \tau . \nabla u + \nabla u^T . \tau \right)$ (3)

$$f(tr(\tau)) = 1 + \frac{\varepsilon\lambda}{\eta_s} tr(\tau) .$$
(4)

The Oldroyd-B model [15, 16]:

$$\lambda \left[ \frac{\partial u}{\partial t} + \nabla . u \tau \right] + \tau = \eta_p (\nabla u + \nabla u^T) + \lambda (\tau . \nabla u + \nabla u^T . \tau)$$
 (5)

where  $\eta_p$  is the zero shear rate viscosity for viscoelastic flow contribution and  $\tau$  is the extra stress tensor,  $\lambda_1$ , is the relaxation time of flow. Total viscosity,  $\eta_0$ ,  $\eta_0 = \eta_P + \eta_S$ , is the sum of the viscoelastic flow contribution viscosity and solvent or Newtonian flow contribution viscosity parts. The Reynolds number (*Re*) and Weissenberg number (*We*) which is defined as the ratio of characteristic fluid relaxation time to characteristic time scale in the flow are given through:

$$\operatorname{Re} = \frac{\rho U H}{\eta_0}$$
$$We = \frac{\lambda_1 U}{H} .$$

. . . .

Where H and U are the characteristic length and velocity in the flow, respectively.

A finite volume method [18-20] with non-uniform staggered grid is used to obtain discrete form of the flow equations. Second order central difference scheme is used for the approximation of the diffusion terms in the momentum equations. Convective terms in the constitutive equations are approximated by at least second-order accurate, bounded and non-uniform version of CUBISTA [21] scheme. This scheme is preferred due to its documented advantages on the higher order schemes when viscoelastic fluids are considered [22].

The implementation of the CUBISTA scheme is carried out via deferred correction method that was proposed by Rubin and Khosla [23], Hayase et al. [24]. The method is widely used to ensure stability of the higher-order schemes for the evaluation of the variables at the faces of the control volumes. Mathematically this technique may be stated as;

$$\boldsymbol{\phi}_{f} = \boldsymbol{\phi}_{LO} + \left(\boldsymbol{\phi}_{HO} - \boldsymbol{\phi}_{LO}\right)^{0} \tag{6}$$

where  $\phi_f$  is flow property  $(u, v, p, \tau_{xx}, \tau_{yy}, \tau_{xy})$ , the first term in Eq. (6) is the result from the low order (*LO*) scheme, and is used to evaluate the coefficient of the discretized equation. The other term is obtained at the previous iteration. Upwind Differencing Scheme (UDS) is used to handle the first term in Eq. (6). High order (*HO*) results are also obtained using Eq. (7). The purpose of the convection scheme use is then to specify the values of  $\phi_f$  at the face of the control volume, based on existing values at the neighbouring cell centres,  $\phi_c$  ( $\phi_p$ ). The detailed numerical computations were given in the study of Tezel (2016). [25]

$$\phi_{f} = \begin{cases} \frac{7}{4}\phi_{c} & 0 < \phi_{c} < \frac{3}{8} \\ \frac{3}{4}\phi_{c} + \frac{3}{8} & \frac{3}{8} \le \phi_{c} < \frac{3}{4} \\ \frac{1}{4}\phi_{c} + \frac{3}{4} & \frac{3}{4} < \phi_{c} < 1 \\ \phi_{c} & \text{elsewhere} & (\phi_{c} > 1 \text{ or } \phi_{c} < 0) \end{cases}$$

$$(7)$$

The final form of the two dimensional discretized governing equations over a control volume can be expressed symbolically as follows using finite volume method:

$$A_{P}\phi_{i,j} = A_{E}\phi_{i+1,j} + A_{W}\phi_{i-1,j} + A_{N}\phi_{i,j+1} + A_{S}\phi_{i,j-1} + b \quad . \tag{8}$$

Where  $\phi_{i,j}$  represents i<sup>th</sup> and j<sup>th</sup> component of  $\phi$  in control volume. The coefficients are also approximated by the upwind differencing scheme. Gradients of velocities in the source terms of the constitutive equations are computed by central differences at interior points of the domain. However, at the flow boundaries they are calculated by approximating velocity profiles in the form of polynomial functions [24].

The SIMPLE [26] method is employed to handle the coupled system of the continuity, momentum and constitutive equations. The set of linearized algebraic equations are then solved by using the Thomas algorithm or the tridiagonal matrix algorithm (TDMA). The solution process is reiterated until the maximum relative change of flow variables,  $\mathcal{O}_{i,j}(u, v, p, \tau_{xx}, \tau_{yy}, \tau_{xy})$  is less than a prescribed tolerance or residual as:

$$res = MAX \left\{ \frac{\left| \phi^{n+1} - \phi^{n} \right|}{\left| \phi^{n+1} \right|} \right\} \le 1x10^{-8} .$$
(9)

Use of correct boundary conditions especially for stress components is crucial to capture the physics of the flow. Boundary values of stress components are obtained by solving constitutive equations with the known velocity boundary conditions. The following inlet conditions are imposed for x and y-components of the velocity, u and v respectively. Corresponding values of stresses are given in Eq. (11).

$$u = 1 - \left| (1 - 0.5y)^2 \right|$$
  

$$v = 0$$
(10)  

$$0 \le v \le 4$$

$$\tau_{xx} = 2We(1-\beta)\left(\frac{\partial u}{\partial y}\right)^{2}$$
  

$$\tau_{yy} = 0$$
(11)  

$$\tau_{xy} = (1-\beta)\left(\frac{\partial u}{\partial y}\right).$$

No slip boundary conditions are assumed at the solid-fluid interfaces:

$$u = 0 (12)$$
$$v = 0.$$

Stress boundary conditions for solid channel walls can then be expressed by the following equations:

$$a = 1 + \varepsilon We(\tau_{xx} + \tau_{yy}) / (1 - \beta)$$
  

$$\tau_{xy} = (1 - \beta) \frac{\partial u}{\partial y} / a$$
  

$$\tau_{xx} = 2We(1 - \beta) \left(\frac{\partial u}{\partial y}\right)^2 / a .$$
(13)

At the channel exit Neumann boundary conditions are imposed for the flow variables.

$$\frac{\partial u}{\partial x} = 0, \ \frac{\partial v}{\partial x} = 0, \ \frac{\partial \tau_{xx}}{\partial x} = 0, \ \frac{\partial \tau_{yy}}{\partial x} = 0, \ \frac{\partial \tau_{xy}}{\partial x} = 0, \ \frac{\partial \tau_{xy}}{\partial x}, \ \frac{\partial P}{\partial x} = 0.$$
(14)

### **3** Numerical Results and Discussion

A Reynolds number range  $10^{-4} \le Re \le 20$  was investigated numerically, where *Re* is based on the cylinder diameter *b* and the maximum flow velocity  $u_{max}$  of the parabolic inflow profile as seen in Fig. 1. The following section starts with a description of the different flow patterns observed with respect to *Re* and *We*. The subsequent sections present a detailed comparison of the computed results based on velocity, pressure and stress profiles at several positions in the flow field for both of the viscoelastic model. Furthermore, the computations are analyzed and compared in terms of drag coefficient.

#### 3.1 Streamlines around the obstacle

We begin presenting the computational results of the flow system by the inertial and elasticity effects on the recirculation patterns behind the obstacle depicted in Figs. 2(a)-(d). The values of the constitutive model parameters used to obtain these results are  $\varepsilon = 0.25$  and  $\beta = 0.2$ . Dimensionless recirculation length that is also known as the wake region is defined by Breuer et al. [9] as the distance between the obstacle surface and reattachment point of streamlines to form the encapsulated region behind the obstacle. It should be noted that at a given Re upper limit of We is determined by the stability of the computations. The higher the value of Re, the lower We that can be attained for stable computations. For example when Re is set as 20, the maximum value of the attainable We is 3 and 6 for PTT and Oldroyd-B fluid respectively. Fig. 2 illustrate that increasing fluid elasticity or inertia leads to larger recirculation lengths and eventually formation of symmetric vortexes as depicted in Fig. 2(c) similar to the results reported by Breuer et al. [9]. Larger recirculation lengths and observed vortexes can be attributed to Hoop stresses getting stronger at increased fluid inertia and

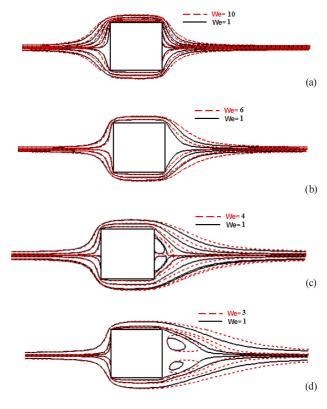


Fig. 2 Effect of *We* on the recirculation length for PTT fluid at (a) Re = 0 (b) Re = 5 (c) Re = 10 (d) Re = 20

elasticity. Hu and Joseph [27] used UCM model, which has the same behavior as Oldroyd-B, and reported similar trend in the flow around circular cylinder. They also observed larger vortices with increasing elasticity.

The flow quantities such as stress and velocity fields are determined through the complex interactions between inertia, elasticity and shear thinning that gets stronger at elevated *Re* and *We*. In Figs. 3(a)-(c), viscoelastic fluid effects on the recirculation length (*Lr*) are compared at different *Re*. Oldroyd-B fluid with constant viscosity flow has larger wake region size or recirculation length depicted in Table 1. Huang and Feng [28] obtained the same wake lengths for Oldroyd-B flow around circular cylinder at *Re* = 10 when *We* is increased zero to one. Also, *Lr* is increased with *Re* numbers as in Newtonian flow.

Viscoelastic wake behind square obstacle is longer than the Newtonian wake (We = 0). Because, the wake formation has strong dependence on the structure of the flow and on the presence of vortices in the region of highest stress that resides downstream of the rear stagnation point of obstacle surface. Therefore, lacking shear thinning property, Oldroyd-B fluid leads to higher stresses and larger wake field around the obstacle compared to PTT and Newtonian flows. Oldroyd-B flow vortex centers also shift upward and downward direction with respect to PTT fluid due to

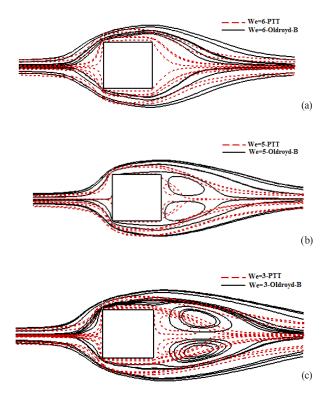


Fig. 3 Viscoelastic fluid effect on the recirculation length for (a) Re = 5 (b) Re = 10 (c) Re = 20

Re	We	Lr of PTT at ( $\psi = 0$ )	<i>Lr</i> of Oldroyd-B at $(\psi = 0)$
	0	20.2132	20.2132
	1	21.6133	22.8912
5	2	21.9957	23.0755
	5	23.4713	25.1997
	6	23.8912	25.3664
	0	21.0341	21.0341
10	1	24.7001	25.1285
	2	25.1213	26.4356
	3	25.3634	27.3259
	4	26.1784	27.8712
	5	26.2467	28.0125
	0	22.3452	22.3452
20	1	25.7823	26.5648
20	2	26.1239	27.1547
	3	26.8745	28.7645

 Table 1 Variation of recirculation length (Lr) with Reynolds number (Re)

 for different We numbers

expanding wake region in Figs. 3(b) and 3(c). This can be clearly seen in Table 2. The vortex pair size increases in wake region as Re number increases as in Fig. 3(c), when vortex intensities increase for both model listed in Table 2.

However, at Re = 10, the magnitude of vorticity becomes lower. For low Re number, inertial force gets lower and impact of the elasticity on the flow is comparatively smaller than at high Re number. So, at Re = 20, the elasticity of polymer molecule is higher for Oldroyd-B than PTT flow in Fig. 3(c). So, vortex intensity values in wake region  $(1.3079 < \psi < 1.3179)$  gets higher at *x* locations away from the obstacle at a constant y = 2.2006 for Oldroyd-B flow at Re = 20 as depicted in Table 2. Also, this vortex enhancement may be attributed to large and constant elongational viscosity of Oldroyd-B flow. Because, large elongational viscosity delays the acceleration of the fluid which results in the increase of vortex size.

### 3.2 Velocity and Pressure profiles around the obstacle

Fig. 4 presents velocity distribution of component, u the flow field for Re = 20 with different We numbers along the center line. Axial velocity, u, dominates the flow field as compared to vertical velocity, v. At x = 19 that is the position prior to the cylinder region, the velocity profiles differ from Poiseuille flow. Streamwise velocity, u, has a local minimum points for each We numbers. As the We number increases, the local minimum velocity decreases as in Fig. 4(b) at x = 19for both model. On the other hand, at x = 22, local maximum

their locations in the wake region						
PTT						
We	Re	ω	Ψ	x	У	
5	10	-0.1275	1.3314	21.2222	2.1487	
5	10	-0.1462	1.3315	21.2502	2.1487	
5	10	-0.1855	1.3317	21.3040	2.1487	
5	10	-0.2073	1.3320	21.3405	2.1487	
5	10	-0.2284	1.3324	21.3501	2.1487	
5	10	-0.2504	1.3329	21.4063	2.1487	
3	20	-0.2354	1.3263	21.5527	2.1487	
3	20	-0.2620	1.3267	21.6340	2.1487	
3	20	-0.2891	1.3276	21.7213	2.1487	
3	20	-0.3024	1.3282	21.7673	2.1487	
3	20	-0.3154	1.3291	21.8149	2.1487	
3	20	-0.3495	1.3322	21.9683	2.1487	
Oldroyd						
We	Re	ω	Ψ	x	У	
5	10	-0.1281	1.3051	21.5527	2.2006	
5	10	-0.1591	1.3068	21.6340	2.2006	
5	10	-0.1963	1.3076	21.7212	2.2006	
5	10	-0.2172	1.3085	21.7673	2.2006	
5	10	-0.2402	1.3104	21.8149	2.2006	
5	10	-0.3305	1.3123	21.9683	2.2006	
3	20	-0.3884	1.3073	22.2625	2.2006	
3	20	-0.3918	1.3086	22.3278	2.2006	
3	20	-0.4036	1.3112	22.3954	2.2006	
3	20	-0.4095	1.3117	22.4654	2.2006	
3	20	-0.4145	1.3157	22.6128	2.2006	

 
 Table 2 Intensities of the primary eddies and vorticity and their locations in the wake region

velocity increases at elevated *We*. In the wake region, x = 22, *u*-velocity profiles become highly modified due to the presence of the obstacle. For both model, flow is nearly independent of We number at Re = 20. These results associated with the velocity field can be attributed to shear thinning property of the fluid that becomes more pronounced at elevated We values as shown in Fig. 4(a) for PTT fluid. On the other hand, u velocity for Oldroyd-B has more deformation due to more elastic behavior and the maximum velocity region shifts (see Fig. 4(b)). In other words, the elasticity changes dramatically the flow at high We numbers. The streamwise velocity at high We is slower to recover the undisturbed bulk velocity than low We flows especially in the wake region as seen in Figs. 2(a)-(d). Fig. 4(c) shows influence of We on profiles of the streamwise velocity component, u, along the centerline (y = 2) for PTT fluid. We also compare these results with the corresponding numerically computed velocity profiles for a Newtonian fluid under at Re = 20 (see Fig. 4(d)). Upstream of

the cylinder the flow is essentially independent of *We* along the cylinder, in agreement with Figs. 4(c)-(d). For PTT flow, required length to achieve the fully developed velocity,  $\partial u / \partial x = 0$ , is nearly the same for all *We* numbers plotted in Figs. 4(a) and 4(c) while fully developed velocity in magnitude is lower than Newtonian flow. Axial velocity is shifted downstream and thus decreases relative to Newtonian flow with increasing *We*. For Oldroyd-B flow elasticity of the fluid also leads to increase in the required length for velocity recovery as shown in Fig. 4(d).

Fig. 5 shows the pressure variation in the flow. For viscoelastic flow case, pressure gradients are generated by mainly elastic effects in near front and rear stagnation points of the cylinder. Viscoelastic pressure drop values are also higher than Newtonian flow case due to increasing effect of longitudinal flow as seen in Figs. 5(a) and 5(b). Fig. 5(a) depicts pressure profiles for PTT flow. As *We* increases, pressure value increases as shown in Fig. 5(a). This result can be attributed to the smaller extensional viscosities associated with higher *We* for a PTT fluid. Pressure drop also gets amplified as in Fig. 5(b) due to the breaking of the fore-aft symmetry of flow around obstacle (see Fig. 4(d)) at higher *We* for Oldroyd-B.

# 3.3 Normal and Shear stress profiles around the obstacle

In order to further examine the behavior of viscoelastic flows around the obstacle, we plot shear and normal stress profiles as in Fig. (6).  $\tau_{xx}$  and  $\tau_{xy}$  variation around the cylinder are given along x and y direction of the flow. Both shear and normal stresses are zero at the front and the rear stagnant points since the velocity gradient  $(\partial u / \partial x = 0)$  is zero at the centerline as in Figs. 6(a) and 6(b). In the case of PTT fluid, at Re = 0 and 10, absolute value of normal and shear stresses decreases as We increases due to stress decaying of flow field as shown in Figs. 6 and 8. Higher the We is, stronger the shear thinning effects that lead to the lower stress values. As the inertial effects get amplified,  $\tau_{xx}$  and  $\tau_{xy}$  along the x direction becomes smaller at y = 0. Also, shear stress peak is observed at the front stagnation point due to sudden change in the boundary condition (at singularity point) for all We numbers as shown in Fig. 8(a). For the Oldrody-B fluid, absolute magnitude of stresses increases compared to PTT in the flow field as in Figs. 7 and 9. In Fig. 7(a) overshoot in  $\tau_{rv}$  profile occurs at We = 15. It may be resulted from u velocity gradient with respect to y direction at x = 19. On the other hand, as We gets higher, there is an increase in all stress gradients as shown in Fig. 7(b). The saturation of normal stresses at

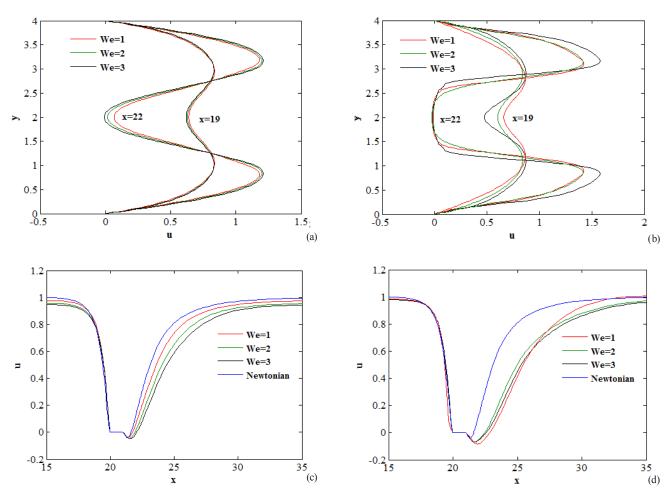


Fig. 4 Velocity component, u, profiles for different positions at x = 19 (before cylinder), x = 22 (wake region) and (a) for PTT fluid and (b) for Oldroyd-B fluid, at y = 2 (centerline) and (c) for PTT fluid and (d) for Oldroyd-B fluid.

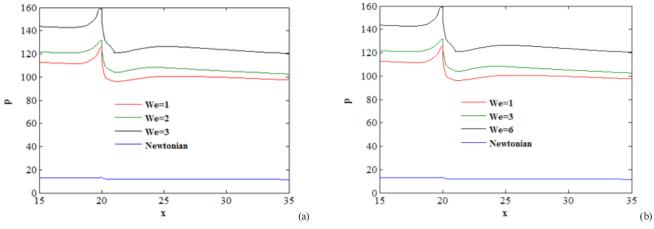


Fig. 5 Pressure profiles around the cylinder and (a) for PTT fluid and (b) for Oldroyd-B fluid along the centerline.

the obstacle surface was also observed in creeping flow of Oldroyd-B fluid around cylinder [29, 30].

At Re = 10, for PTT in the wake region, shear stress relaxation occurs in an oscillation manner as in Fig. 8(a). However, Oldroyd-B have more absolute shear stresses, stress relaxation is more quickly and suddenly owing to more flexible behavior of polymer chain, when the stress source removed (away from the cylinder), it is returned into undeformed case of the flow as in Fig. 9(a). Another important flow feature is the response of normal stress

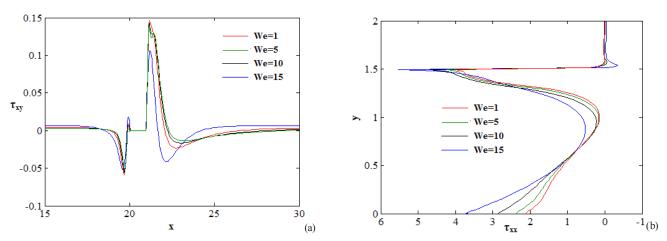


Fig. 6 Shear stress,  $\tau_{xy}$ , and normal stress,  $\tau_{xx}$ , profiles around the obstacle for PTT fluid at  $Re = 10^{-4}$ .

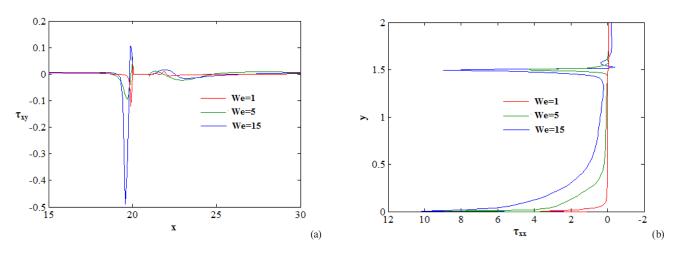


Fig. 7 Shear stress,  $\tau_{xy}$ , and normal stress,  $\tau_{xx}$ , profiles around the obstacle for Oldroyd-B fluid at  $Re = 10^{-4}$ .

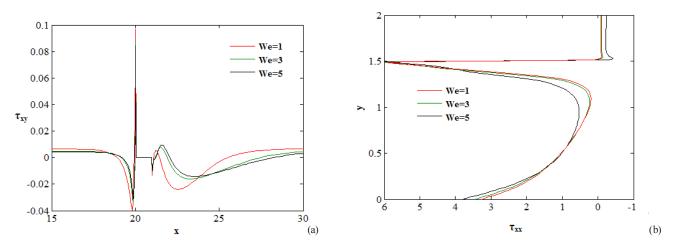


Fig. 8 Shear stress,  $\tau_{xy}$ , and normal stress,  $\tau_{xx}$ , profiles around the obstacle for PTT fluid at Re = 10.

behavior along the obstacle surface at x = 20. Normal stresses can be considered conceptually due to the tension of the streamlines of the flow field. Normal stress relaxation or distribution occurs between the obstacle

surface and the channel wall at y = 0. PTT fluid delays  $\tau_{xx}$  momentum transfer from the obstacle surface to the channel wall. Near the obstacle walls, the deformation rate of the fluid gets higher that in turn results in decrease in the

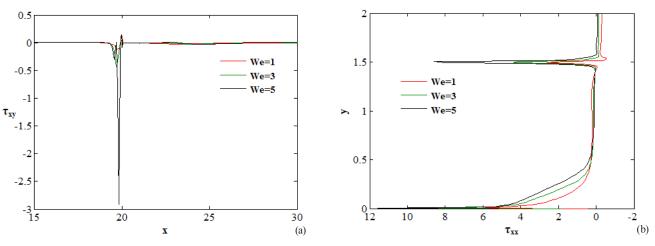


Fig. 9 Shear stress,  $\tau_{yy}$ , and normal stress,  $\tau_{yy}$ , profiles around the obstacle for Oldroyd-B fluid at Re = 10.

shear viscosity decreases along with the normal stress. For Oldroyd-B fluid, all elastic or normal stresses are nearly recovered as suggested by the plots in Fig. 7(b). However, as *Re* increases, elastic stresses reach a maximum value (We = 5) at the channel wall as in Fig. 9(b). Oldroyd-B flow relaxes quickly elastic stresses at the center of the obstacle compared as PTT flow.

# **3.4 Viscoelastic Drag Phenomena over the** square cylinder

The corresponding drag coefficient can be splitted into three parts. These are drag coefficient due to normal stress contribution, drag coefficient due to shear stress contribution, drag coefficient due to pressure drop contribution [31]. All these contributions are expected to be functions of *Re* and *We* numbers and also coupled each other.

In the case of Oldroyd-B fluid, investigation of drag coefficients becomes more involved due to occurrence of higher stress gradients than those of PTT fluid. The drag increases with *We* number as tabulated in Table 3.

This increase in drag is associated with a remarkably long recirculation region as shown in Table 1. Another reason is the formation of Hoop stresses at high *We* flows in the wake region. For PTT fluid, this region is more stabilized owing to decaying of stress field. When *We* increases, shear and normal stresses contributions to drag coefficient are small due to constant viscosity of Oldroyd-B fluid at a given *Re*. At Re = 5, up to We = 4, there is an effective balance between normal stress and shear stress drag contribution leading to nearly the same drag coefficients. For We = 5 and 6, pressure drop contribution to drag coefficient decreases because pressure magnitude suddenly drops as in Fig. 7(c), while the

$\begin{array}{cccccccccccccccccccccccccccccccccccc$	Cd
$5 \begin{array}{cccccccccccccccccccccccccccccccccccc$	33.484
5         4         17.494         120.037         129.797         13           5         12.182         121.342         150.182         13           6         11.176         122.959         160.955         13	4.402
4       17.494       120.037       129.797       13         5       12.182       121.342       150.182       13         6       11.176       122.959       160.955       13	34.310
6 11.176 122.959 160.955 13	4.922
	32.975
1 16.062 105.028 114.612 7	39.917
	6.051
2 16.092 107.543 115.453 7	6.783
<i>10</i> 3 16.149 109.912 118.582 7	7.821
4 17.023 110.123 118.591 7	9.788
5 17.699 114.723 119.400 8	2.222
1 18.595 94.852 83.381 5	5.013
20 2 18.909 95.695 85.561 5	5.943
3 22.642 96.088 108.727 6	5.767

normal stress contribution to drag coefficients gets amplified. Furthermore, Cd increases slightly with increased We at high Re. This behavior of drag can be explained by the secondary flow formation resulting from vortex pairs in the wake region.

At Re = 20, large deformation is observed in normal stress field especially at We = 3. Therefore, normal stress contribution to drag coefficient increases from 85 to 108, along with stronger pressure contribution as shown in Table 3. Also, the rapid change of pressure and the formation of normal stress boundary layer generates high extensions and shears around the obstacle. For We < 3, negative normal stress field around the obstacle has also decreasing effect in magnitude of normal stress contribution to drag

0.6         21.131         34.884         62.087           0.7         21.290         41.441         73.045           0.8         21.483         49.076         75.840           0.5         18.727         33.503         47.132	51.960 51.959
1         0.7         21.290         41.441         73.045           0.8         21.483         49.076         75.840           0.5         18.727         33.503         47.132           0.6         18.841         38.548         49.264	51.050
0.7       21.290       41.441       73.045         0.8       21.483       49.076       75.840         0.5       18.727       33.503       47.132         0.6       18.841       38.548       49.264	51.959
0.518.72733.50347.1320.618.84138.54849.264	54.030
0.6 18.841 38.548 49.264	55.456
	45.517
	46.463
	49.294
0.8 19.704 46.934 71.230	51.224
0.5 18.079 29.123 45.342	43.604
	45.695
<sup>3</sup> 0.7 18.920 37.342 63.651	47.939
0.8 18.951 39.357 70.370	11.757

 Table 4 Drag coefficients for PTT fluid at Re = 20 for different polymer viscosity ratio

coefficient. Table 4 shows drag coefficients variation with polymeric viscosity ratio for PTT fluid.

In our governing equations, polymer concentration effect is characterized by the polymer viscosity ratio, wr. Polymeric concentration affects both the viscosity and the relaxation time of polymer molecules in viscoelastic medium. We carry out a few tests for various wr for We = 1, 2 and 3 at fixed Re = 20 tabulated in Table 4. Although shear and normal stresses contribution to drag increase, the variation of Cd is nearly same for the range of We. However, pressure drop contribution has slightly increased with wr. The higher is the value of wr, the more drag enhances for We = 3 case due to increased elasticity effect. As demonstrated in Table 5, at higher wr, drag coefficient increases at a constant We. At high We flows, further increase of polymer concentration makes no significant contribution to drag force around the obstacle as shown in Table 5

# **4** Conclusions

In this study numerical studies are carried out for steady laminar flows of viscoelastic fluids around the confined square cylinder to reach physical mechanism of flow behavior around the square obstacle. Oldroyd-B model (constant viscosity with elasticity) and the PTT model (shear thinning with elasticity) are used to capture viscoelasticity. Flows are simulated at various Reynolds and Weissenberg numbers by utilizing the finite volume method. The results obtained in this study allow one to draw the following conclusions:

We	wr	Pressure Contribution to Cd	Shear Stress Contribution to Cd	Normal Stress Contribution to <i>Cd</i>	Cd
	0.5	16.453	42.234	44.321	41.561
1	0.6	17.262	43.564	49.124	43.792
	0.7	18.231	44.122	51.984	46.072
	0.8	18.595	44.852	53.381	47.013
2	0.5	17.231	54.546	54.235	45.340
	0.6	17.871	55.167	55.146	46.773
	0.7	18.453	55.875	55.456	48.039
	0.8	18.909	55.695	55.561	48.943
3	0.5	22.134	56.972	65.123	56.477
	0.6	22.345	57.012	66.341	57.025
	0.7	22.456	58.981	67.891	57.599
	0.8	22.643	59.0887	68.727	58.067

Table 5 Drag coefficients for Oldroyd-B fluid at Re = 20 for differentpolymer viscosity ratio

- Increasing fluid elasticity or inertia leads to larger recirculation lengths and eventually formation of symmetric vortexes as in Fig. 2(d) and Fig. 3(c).
- Strong impact of *Re* on the highest attainable *We* was observed for stable computations. At higher value of *Re*, upper limit of *We* should be reduced to get stable solutions for PTT and Oldroyd-B model.
- A detail examination of velocity profiles around the obstacle of PTT and Oldroyd-B flow reveal that streamwise velocity at high We flow delays recover undisturbed bulk velocity in the wake region for both flow at constant Re number. On the other hand, for PTT flow, required length to achieve the fully developed is nearly same for Newtonian flow at all We. But for Oldroyd-B flow with increased of We, elasticity of fluid leads to increase required length to achieve the fully developed region in the wake compared to Newtonian flow as in Figs. 4(c)-(d).
- Another important feature is the response of normal stress behavior along the obstacle surface. Normal stress relaxation occurs between obstacle surface and the channel wall. PTT fluid delays normal stress momentum transfer from obstacle surface to the channel wall. As the deformation rate is raised, shear viscosity decreases approaching to the channel wall and normal stress loss is observed. As shown in Figs. 7(b) and 9(b), for Oldroyd-B model, all elastic or normal stresses are nearly recovered and absorbed by the channel wall. It relaxes quickly elastic stresses compared as PTT model.

• Design parameter, drag coefficient, *Cd*, shows that viscoelastic wake behind the square obstacle is longer than the Newtonian wake. With an increase in *We*, *Cd* values get smaller due to stronger shear thinning effects for PTT fluid, whereas *Cd* continues slightly to enhance for Oldroyd-B fluid. The variation of *Cd* with increased *wr* is nearly same for the range of *We* numbers for both model. At high *We* flows, further increase of polymer concentration makes no significant contribution to *Cd* around the obstacle.

#### Nomenclature

- *Re* Reynolds number
- We Weissenberg number
- $\varepsilon$  Extensibility parameter

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- $\eta_s$  Solvent viscosity
- $\eta_0$  Total viscosity
- $\beta$  Ratio between solvent viscosity and total viscosity
- *B* Blockage ratio
- $\tau_{xx}$  Normal stress
- $\tau_{xy}$  Shear stress
- *P* Pressure
- *u* Horizontal velocity
- v Vertical velocity
- Cd Drag coefficient
- *n* Power law index
- $\omega$  Intensity of the primary vortex
- $\psi$  Stream function
- *wr* polymer viscosity ratio
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